

Curve Veering in Torsional Systems with Stepped Shafts

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Abstract

In this study, the influence of geometrical parameters on the curve veering phenomenon in a torsional system with stepped shaft is investigated. Three approximate solutions including finite element, Rayleigh-Ritz and discretization methods, along with an exact solution are employed to obtain the natural frequencies of the structure. The study reveals that, under specific circumstances, the results obtained by approximate methods are very close to the exact solution. The curve veering behavior is manifested irrespective of the method employed. It is concluded that for the structure studied the curve veering behavior is not because of the approximate techniques used to compute the natural frequencies, and is an inherent behavior of the structure.

Keywords

Curve Veering, Torsional System, Stepped Shaft

1. Introduction

Curve veering is defined as an abrupt veering of the natural frequency plots, when plotted against some system parameters [1]. This phenomenon was reported by Warburton [2], for the first time. Curve veering was observed, when variation of natural frequencies of the rectangular plate against side ratio was plotted. Leissa [3] observed curve veering in the vibration of square plates. When the variation of natural frequencies against aspect ratio was plotted, it was observed that the curves change smoothly everywhere except in some regions, where they show sudden changes. Leissa called these regions “transition zones” and showed that this behavior is attributed to approximate solution employed for finding the natural frequencies of the structure. The curve veering disappeared when an exact solution was employed. Schajer [4] reported an interesting feature of curve veering in vibration analysis of the vibrating string with a spring support. The study showed that curve veering is not limited

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to approximate solutions, and it may be an inherent behavior of some vibrating systems. Curve veering as an inherent property of the structure may be seen in rotating disks and plates [5], clamped beams on intermediate elastic supports [6] and vibration of disordered systems [7]. The significance of curve veering derives from the fact that, a small variation of frequency in the transition zone may yield a sudden change in the vibrational mode. If an external force excites the i th natural frequency of the structure in the transition zone, a small change of excitation frequency causes the $(i+1)$ th frequency of the structure to be excited. As a result, a small change in excitation frequency yields a sudden change in the normal mode of the structure, and the satisfactory performance of the structure may be severely affected. The effect of frequency curve veering in instability of mechanical structures has been widely addressed. For example, mode localization reported in shallow arch [8], engineering structures [9] and cantilever beam [10] may be regarded as a result of curve veering in these structures. Moreover, estimation and veering analysis of imperfect structures such as cracked plate [11], nonlinear beam with geometry imperfection [12] and system with gyroscopic coupling [13] have been reported in the literature. It is worth noting that, frequency curve veering may cause localized buckling [14] or wrinkling in specific structures [15].

In high speed rotating machinery, a considerable number of studies have been carried out on the natural frequencies and mode shapes. Most often, in view of the complex geometry of the rotor systems, they are treated as lumped rotors mounted on shafts. In many practical situations, the shafts may have different cross sections and may have stepped configuration. Accurate determination of the natural frequencies is imperative in order to ensure that the system does not operate near resonant frequencies and particularly in the vicinity of curve veering ranges. Exact solutions are possible only in the case of well-defined uniform shaft geometries, and for practical rotors with many cross sectional changes, approximate techniques such as the discretization method, the Rayleigh Ritz method, finite element method are used. The first general theory for free vibration analysis of torsional systems was reported by Beddoe [16]. A one dimensional wave equation was employed to derive equation of motion of the structure. Maltbeak [17] [18] and Rao [19] studied free torsional vibration of uniform shafts with discrete inertias. Maltbeak assumed a sinusoidal angular displacement along the shaft. Wilson [20] utilized effective inertia method to analyze torsional vibration of a complicated system. To this end, the main structure was divided into some simple sub models, and the sub models were analyzed individually. Finally, Wilson found frequency characteristics of the main structure using a combination of the results obtained from the sub models. Leissa and So [21] applied three dimensional solution for estimation of natural frequencies of the shaft structure.

In the present study, a stepped shaft supporting a rotating disk at the tip is analyzed for its curve veering behavior by computing the natural frequencies by different methods. The effect of geometric parameters of the stepped shaft disk system on the curve veering phenomenon is investigated. Although approximate solutions exhibit curve veering in the structure, an exact method is also employed to confirm this phenomenon as an inherent property of the structure.

2. Mathematical Formulation

An isotropic, homogeneous torsional system composed of a stepped shaft with a lumped disk at the tip, as depicted in **Figure 1**, is used in the study. The length and diameter of upper and lower shafts are L_1 , d_1 and L_2 , d_2 , respectively, and total length of the stepped shaft is L . Moreover, M and d denote the mass and diameter of the lumped disk.

2.1. Exact Solution

The equation of motion of the shaft is given by:

$$\frac{\partial^2 \theta}{\partial t^2} = \left(\frac{G}{\rho} \right) \frac{\partial^2 \theta}{\partial x^2} \quad (1)$$

where ρ , G and θ are density, shear modulus and twist angle of the shaft, respectively. The solution of Equation (1) may be found as follows [22]:

$$\theta(x, t) = \left[A \sin(\omega \sqrt{\rho/G} x) + B \cos(\omega \sqrt{\rho/G} x) \right] [C \sin \omega t + D \cos \omega t] = \Theta(x) f(t) \quad (2)$$

where ω is the frequency of vibration and A , B , C and D are unknown coefficients. The characteristic equation

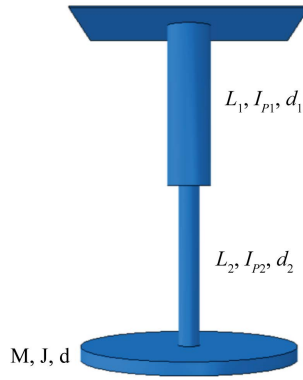


Figure 1. Stepped shaft connected to disk.

is obtained by substituting the boundary conditions as:

$$\begin{bmatrix} \sin(\lambda L_1) & -\cos(\lambda L_1) & -\sin(\lambda L_1) \\ I_{p1}\cos(\lambda L_1) & I_{p2}\sin(\lambda L_1) & -I_{p2}\cos(\lambda L_1) \\ 0 & J\omega^2\cos(\lambda L) + GI_{p2}\lambda\sin(\lambda L) & J\omega^2\sin(\lambda L) - GI_{p2}\lambda\cos(\lambda L) \end{bmatrix} \begin{Bmatrix} B_1 \\ A_2 \\ B_2 \end{Bmatrix} = \begin{Bmatrix} 0 \\ 0 \\ 0 \end{Bmatrix} \quad (3)$$

where $\lambda = \omega\sqrt{\rho/G}$ and I_{p1} , I_{p2} and J are polar moment of inertia of upper shaft, polar moment of inertia of lower shaft and polar mass moment of inertia of the disk, respectively, and A_2 , B_1 and B_2 are unknown coefficients. For non-trivial solutions, the determinant of the matrix should be set to zero which will yield the natural frequencies.

2.2. The Rayleigh-Ritz Method

In the Rayleigh-Ritz solution, displacement field of a structure is defined as linear combination of admissible functions. In this study, the deflection shape is considered as:

$$\theta(x) = c_1\theta_1(x) + c_2\theta_2(x) + \dots \quad (4)$$

where $\theta_i(x)$ are admissible functions satisfying at least the geometrical boundary conditions and c_i are arbitrary coefficients. For a shaft with distributed mass and elasticity, the kinetic and potential energy expressions are given by:

$$U_{\max} = \frac{1}{2}GI_p \int_0^L \left(\frac{d\theta}{dx}\right)^2 dx \quad (5)$$

$$T_{\max} = \frac{1}{2}\rho I_p \omega^2 \int_0^L \theta^2(x) dx \quad (6)$$

where ω is the frequency of vibration. For harmonic vibrations, we have:

$$U_{\max} = T_{\max} \quad (7)$$

The conditions for the stationary of the natural frequencies with respect to the arbitrary coefficients in the assumed deflection expression formulate the eigenvalue problem of the structure. It is well-known that the natural frequencies obtained by the Rayleigh-Ritz method are the upper bound. In this study, the following formulation is employed to obtain orthogonal admissible functions [23].

$$\theta_1(x) = x; \theta_2(x) = (x - B_2)\theta_1(x); \theta_i(x) = (x - B_i)\theta_{i-1}(x) - C_i\theta_{i-2}(x) \quad (8)$$

where

$$B_i = \frac{\int_0^L x(\theta_{i-1}(x))^2 dx}{\int_0^L (\theta_{i-1}(x))^2 dx} \quad (9)$$

$$C_i = \frac{\int_0^L x \theta_{i-1}(x) \theta_{i-2}(x) dx}{\int_0^L (\theta_{i-2}(x))^2 dx} \quad (10)$$

It should be noted that, increasing the number of admissible functions improves the convergence of the results.

2.3. Discretization Method

Discretization technique may be regarded as the simplest, and the least accurate method that is used to find the fundamental frequency of the structure quickly. In this solution, the stiffness constants of the upper and lower shafts are found, individually. Total stiffness of the stepped shaft is obtained as a series combination of these two shaft segments. It should be noted that, solving the problem using this approach necessitates assuming linear torsional deflection through the stepped shaft, while the exact solution reveals trigonometric functions for the shaft deflection. The stiffness constants of the upper and lower shafts are

$$K_i = \frac{GI_{pi}}{L_i}, \quad i = 1, 2 \quad (11)$$

where G , I_{pi} and L_i are shear modulus, polar moment of inertia and length of each shaft, respectively. Total stiffness of the stepped shaft is given by:

$$K_{eq} = K_1 K_2 / (K_1 + K_2) \quad (12)$$

Finally, fundamental frequency of the structure is obtained as:

$$\omega = \sqrt{K_{eq} / J} \quad (13)$$

where J is polar mass moment of inertia of the disk.

3. Results and Discussion

In this study, the following base line values are assumed in the analysis: shear modulus of the structure is 79.3 Gpa, density is 7800 kg/m³, L_1 , L_2 , d_1 , d_2 , d and M are chosen to be 1 m, 1 m, 0.1 m, 0.05 m, 0.5 m and 100 kg, respectively. It should be noted that, all natural frequencies are given in rad/s.

3.1. Comparison of the Results

The exact solution for the fundamental frequency of the structure is obtained as 121.022 rad/s. The results obtained by the Rayleigh-Ritz solution reveal that, using one term of admissible function gives fundamental frequency of the structure as 256.88 rad/s. It should be noted that, the admissible functions satisfy the geometric boundary condition, along with continuity of the angular displacement and torque at the point of step change in the shaft cross section. **Table 1** shows convergence of the fundamental frequency when the admissible functions satisfy only the geometric boundary conditions. The results given in **Table 1** indicate that, they converge to the exact value, although the rate of convergence is poor.

Table 1. Variation of fundamental frequency versus the number of admissible functions in Rayleigh-Ritz method, when $d_1 = 0.1$ m, $d_2 = 0.05$ m, $d = 0.5$ m and $M = 100$ kg.

Number of admissible functions	First frequency
1	256.868
2	165.885
3	137.485
4	136.992
5	130.562
6	130.562
7	127.832
8	127.790

Just using two elements in the finite element model yields the same result as obtained by the exact solution. It is interesting to note that the discretization technique gives fundamental frequency of the structure equal to 121.056 rad/s. The discretization technique considered the displacement field of the structure as a linear function, while the exact solution uses trigonometric functions to describe the displacement field. This is attributed to the magnitude of λ in the exact solution, which has a very small value. In fact, in this order of λ , $\sin(\lambda)$ may be assumed to be equal to λ ; as a result, the frequency obtained by linear deflection assumption is the same as that of the exact solution. In order to clarify this behavior, another example is presented. In this example, all parameter values of the structure remain unchanged except the density, which is assumed to be $\rho = 78000000 \text{ kg/m}^3$. In this case, λ does not have a small value and $\sin(\lambda) \neq \lambda$. The fundamental frequency obtained by the exact solution is 40.8408 rad/s, while the frequency obtained using discretization technique is equal to 121.056 rad/s. Moreover, the Rayleigh-Ritz method—using one admissible functions satisfying geometric boundary condition and continuity of torque and displacement—yields $\omega = 46.7483 \text{ rad/s}$. The difference between these results derives from the fact that, for higher values of λ , the assumption of linear deflection through the shaft is incorrect. In this case, more terms in the Rayleigh-Ritz method is required to obtain accurate results.

3.2. Curve Veering

When the variation of natural frequencies against length ratio (L_1/L) is plotted, the curves change smoothly everywhere except in some regions, where they show sudden changes. As mentioned earlier, Leissa [2] called these regions “transition zones”. This behavior was reported as a result of approximate solution method used to solve the problem, but in the present study curve veering is present whether the results are obtained by the exact method or the approximate methods. Hence, the curve veering may be regarded as an inherent behavior of the system. **Figure 2** shows the variation of the first five natural frequencies against length ratio, when $d_1 = 0.1 \text{ m}$, $d_2 = 0.01 \text{ m}$, $d = 0.5 \text{ m}$ and $M = 100 \text{ kg}$. It is observed that, the variation of the fundamental frequency of the structure against length ratio is not significant. In fact, the curve veering does not occur, when the structure vibrates in the first vibrational mode. The curve veering may be observed in other modes of vibration. The graph

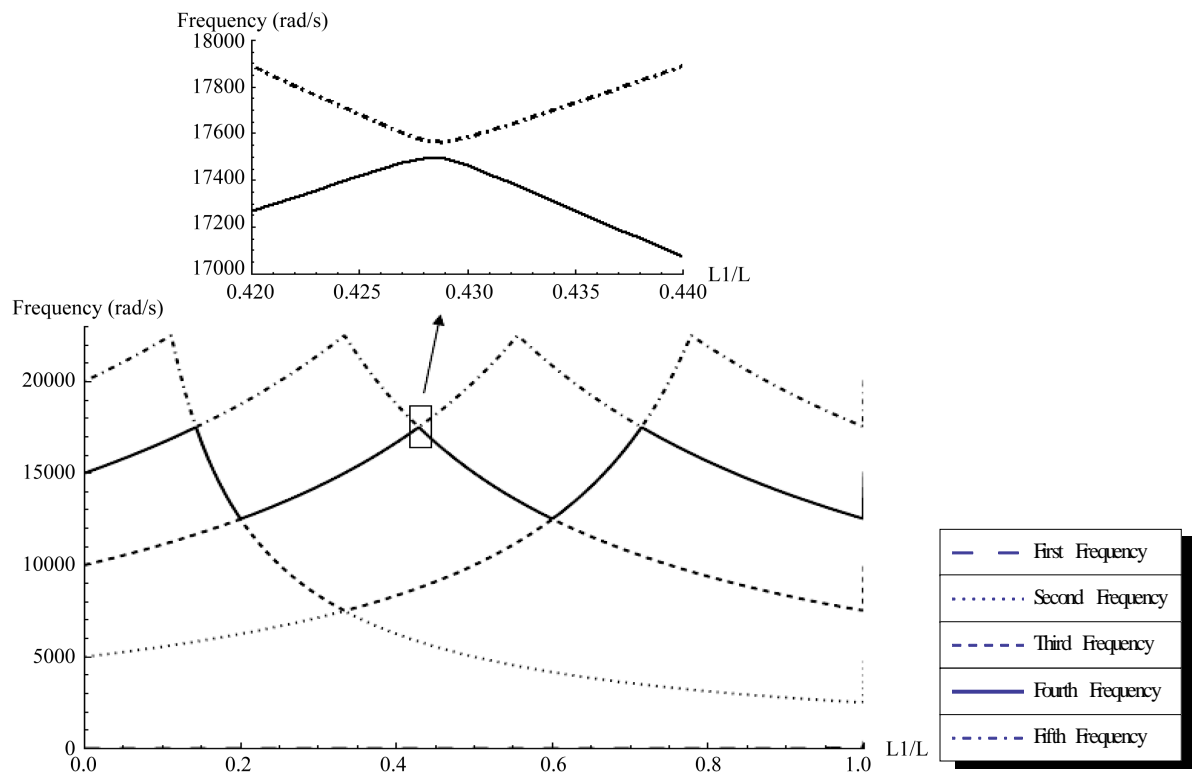


Figure 2. Variation of first five frequencies of the structure versus length ratio when $d_1 = 0.1 \text{ m}$, $d_2 = 0.01 \text{ m}$, $d = 0.5 \text{ m}$ and $M = 100 \text{ kg}$.

shows that the number of transition zones increases when mode number increases. For instance, there is just one transition zone in the second mode of vibration, while three, five and seven transition zones may be observed in the third, fourth and fifth vibrational modes, respectively. In order to understand the curve veering phenomenon, transition zone corresponding to the fourth and fifth modes has been magnified. It can be seen that, natural frequencies approach each other and veer away in this region. This behavior is of great significance for the designers.

Table 2 shows the variation of the first five resonant frequencies of the structure, when $d_1 = 0.1$ m, $d_2 = 0.01$ m, $d = 0.5$ m and $M = 100$ kg and $0.05 < L_1/L < 0.25$. For the range of L_1/L considered, the shaft behaves, approximately, as a slender bar of $d_2 = 0.01$ m; as a result, the fundamental natural frequency of the structure becomes very low, in the range of 3 - 4 rad/s. The bold frequencies indicate the curve veering point in the transition zones. In **Figure 3** the variation of the first five natural frequencies versus length ratio of the structure is shown when $d_1 = 0.1$ m, $d_2 = 0.05$ m, $d = 0.5$ m and $M = 100$ kg. In this case, the curve veering is observed. **Figure 4** shows an analogous study on the structure, when $d_2 = 0.095$ m. It may be seen that variation of length ratio does not show a drastic change in the frequencies. It is attributed to the diameter of upper and lower shafts, which have almost the same magnitudes.

Table 2. Variation of first five frequency of the structure versus length ratio, when $d_1 = 0.1$ m, $d_2 = 0.01$ m, $d = 0.5$ m and $M = 100$ kg.

L_1/L	First	Second	Third	Fourth	Fifth
0.06	3.640	5329	10657	15985	21313
0.07	3.660	5386	10771	16157	21542
0.08	3.679	5444	10888	16332	21776
0.09	3.700	5504	11008	16512	22015
0.10	3.720	5565	11130	16695	22260
0.11	3.741	5628	11255	16883	22501
0.12	3.762	5692	11383	17074	20869
0.13	3.783	5757	11514	17270	19265
0.14	3.806	5824	11648	17467	17680
0.15	3.828	5893	11785	16694	17889
0.16	3.851	5963	11925	15652	17893
0.17	3.874	6035	12069	14732	18104
0.18	3.897	6108	12215	13914	18325
0.19	3.897	6184	12365	13183	18551
0.20	3.945	6261	12482	12562	18873
0.21	3.970	6340	11924	12682	19020
0.22	3.996	6421	11383	12844	19264
0.23	4.022	6505	10888	13010	19514
0.24	4.048	6590	10435	13181	19771

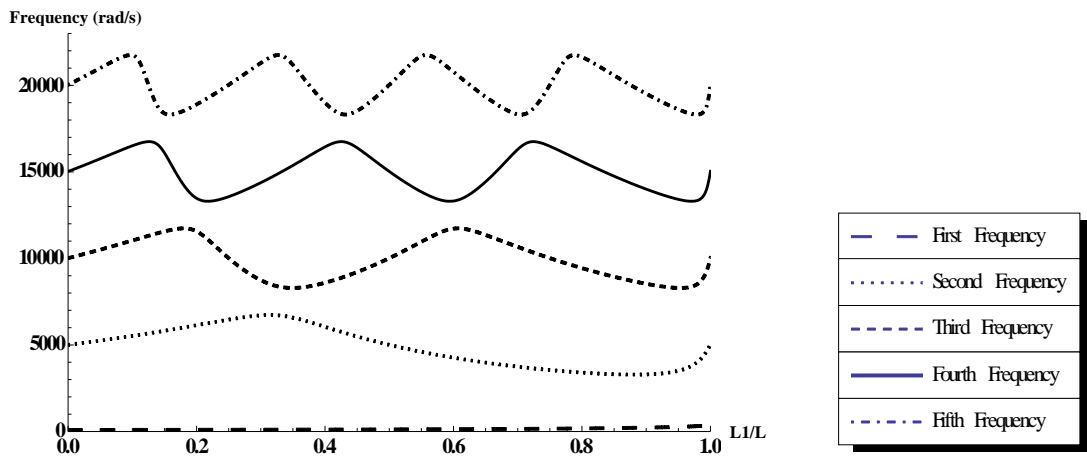


Figure 3. Variation of first five frequencies of the structure versus length ratio when $d_1 = 0.1$ m, $d_2 = 0.05$ m, $d = 0.5$ m and $M = 100$ kg.

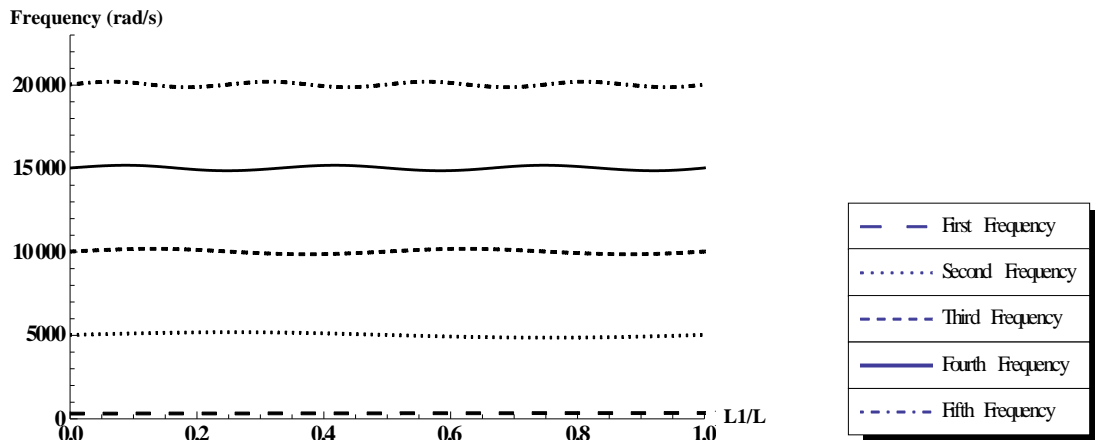


Figure 4. Variation of first five frequencies of the structure versus length ratio when $d_1 = 0.1$ m, $d_2 = 0.095$ m, $d = 0.5$ m and $M = 100$ kg.

4. Conclusion

This study deals with an analysis on the curve veering phenomenon in a torsional structure, which consists of a stepped shaft and a rotating disk. Different approximate techniques including the Rayleigh-Ritz, finite element and discretization methods, along with the exact solution were employed to extract natural frequencies of the structure. The results reveal that curve veering in this structure is not due to application of approximate solution, and it appears even if an exact solution is employed. As a result, the curve veering may be regarded as an inherent behavior of the structure. The geometric parameters affect the curve veering, noticeably. Moreover, a comparison of the results obtained by approximate solutions and those of the exact one was carried out. It was realized that, under some specific geometries and material properties, the frequencies obtained from approximate solutions are as accurate as the exact solution. Under such conditions, the trigonometric functions which describe angular displacement field can be replaced by a linear function.

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