A Discrete-Time Traffic and Topology Adaptive Routing Algorithm for LEO Satellite Networks

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Abstract

"Minimizing path delay" is one of the challenges in low Earth orbit (LEO) satellite network routing algorithms. Many authors focus on propagation delays with the distance vector but ignore the status information and processing delays of inter-satellite links. For this purpose, a new discrete-time traffic and topology adaptive routing (DT-TTAR) algorithm is proposed in this paper. This routing algorithm incorporates both inherent dynamics of network topology and variations of traffic load in inter-satellite links. The next hop decision is made by the adaptive link cost metric, depending on arrival rates, time slots and locations of source-destination pairs. Through comprehensive analysis, we derive computation formulas of the main performance indexes. Meanwhile, the performances are evaluated through a set of simulations, and compared with other static and adaptive routing mechanisms as a reference. The results show that the proposed DT-TTAR algorithm has better performance of end-to-end delay than other algorithms, especially in high traffic areas.

Keywords: LEO Satellite Network, Discrete-Time Traffic and Topology Adaptive Routing (DT-TTAR) Algorithm, End-to-End Delay, Traffic Distribution, Snapshot

1. Introduction

To meet the rapidly growing real-time multimedia services, satellite networks should offer quality of service (QoS) guarantees. Geostationary Earth orbit (GEO) satellites suffer from high propagation delay, which is not suitable for most applications. Therefore, focus has been directed towards development of non-geostationary (NGEO) systems, especially low Earth orbit (LEO) satellite systems, due to their superior performances such as extensive geographic coverage, small end-to-end delay, and inherent multicast capabilities and so on [1-4].

Inter-satellite links (ISLs) in new LEO satellite systems introduce both challenges and opportunities to space segment routings [5]. Some authors capitalize on varies of ISLs and periodically evaluate the propagation delays in ISLs, considering this as the metric to be minimized in the route calculation process. A centralized routing scheme geared toward ATM-based LEO/MEO networks, called dynamic virtual topology routing (DVTR) [6], was proposed, which relied on the Dijkstra shortest path algorithm to compute the optimal path for any pair of satellites. After that, Werner [7] extended that scheme by introducing a heuristic metric for calculating the cost of each ISL. Ekici *et al.* [8] introduced the datagram routing algorithm (DRA) for an idealized polar constellation. It selected the one that most reduced the remaining distance to the destination as the candidate next hops. Sun *et al.* [9] dealt with static routing in a regular LEO satellite network that was modeled as a two-dimensional Nary hypercube. The minimum-hop path was found by the Dijkstra algorithm, and contention resolution schemes were investigated for maximizing the throughput. However, such schemes are not designed or not supposed to take into account traffic related metrics such as the queuing delay.

Non-uniform population and unbalanced distribution of traffic requirements on the Earth impact traffic load in LEO satellite networks directly. This may lead to congestion in some resources, while others are un-utilized. Therefore, improved "minimum-delay path" routing not only involves the propagation delay, but also should consider the ISLs' queuing delay. E. Papapetrou [10] proposed a distributed routing algorithm called location assisted on-demand routing (LAOR), considering these two delays together. This protocol introduced on-demand



routing for traffic load balance in satellite systems first time. Unfortunately, it only used the local traffic information, which might not reflect the global traffic load distribution. The similar link cost was adopted by an explicit load-balancing (ELB) scheme [11], which achieved load balances by aware of traffic conditions at the next-hop satellite. This method reduced the packet dropping probability of each hop [12].

In this paper, a discrete-time traffic and topology adaptive routing (DT-TTAR) algorithm is derived without considering fractional handover but integer handover for a LEO satellite system. For the integer handover, the ground users are switched from one satellite to another entirely. On the hand, fractional handover allows the users shift to the other satellite individually depend on the coverage changing. Supposed the routing program is changing along with fractional handover, it has a name of continuous-time traffic and topology adaptive routing (CT-TTAR). In DT-TTAR algorithm, the routing programs are kept "fixed" in each discrete interval. Such discrete-time topology still exhibits dynamic features in terms of considerable ISLs' distance variations, and the traffic patterns on both up-down and inter-satellite links show variations due to the rapid relative movements between serving satellites and served (global and inhomogeneous) users' distributions. The traffic model [13] incorporated into the DT-TTAR algorithm, takes into account both geographic and demographic factors. Another important aspect of the study is that Jackson queuing network model [14] is adopted to obtain the ISLs' queuing delay.

The remainder of this paper is organized as follows: In Section 2, the topology architecture of the LEO satellite network is introduced. In Section 3, a reasonable method for calculating discrete-time intervals is derived. In Section 4, global traffic distribution model is presented. A queuing model for inter-satellite links is computed in Section 5. Section 7 builds up simulations and analyzes the results. Finally, conclusions are given in Section 8.

2. LEO Constellation Architecture

The satellite network architecture looks like a Twisted Manhattan network [15]. In fact, a real network has small differences which affect the optimal routing in this mesh architecture. To facilitate following discussions, some terminologies are first introduced in **Table 1**. The size of a LEO constellation is $N_L \times M_L$, where N_L is the number of the orbits and M_L is the number of satellites per orbit. Each satellite has four inter-satellite links (ISLs): two to neighbors fore and aft in the same orbital plane, called intra-satellite links, and two to satellites in neighboring orbits to either side, called inter-satellite

links. A paradigm of a π -Constellation in two-dimensional plane is shown in **Figure 1**.

The intra-satellite links are maintained permanently whereas the inter-satellite links between adjacent orbits must be switched-off due to the relative position exchanges when two satellites in consideration cross the poles. In fact, the periodically topological changes of a constellation are resulted from inter-satellite links' switches. For example, when satellite $S_{3,2}$ (the second satellite in the third orbit in **Figure 1**) approaches the polar area, the ISLs $\overline{S_{3,2}S_{2,2}}$ and $\overline{S_{3,2}S_{4,2}}$ should be switched-off.

3. Discrete-Time Intervals Determination

In the earth-fixed cell system [16], the cell on the Earth is fixed and the satellites' antenna beams are steered so as to point toward this fixed cell during some intervals. The status of satellites and related links should be frequently collected in order to accurately depict the current network status. If we consider an instant and think the network as static, the routing can be done easier. According to our assumption, we associate the routing information with the virtual network model [17]. In the virtual network model, the Earth is uniformly divided into $N_L \times M_L$ terrestrial zones which are corresponding to $N_L \times M_L$ virtual nodes. The terrestrial zone corresponding to a footprint area of a satellite is rectangle, contiguous and non-overlapping, and each of them has its logical address Z_i , $i = 1, 2, \dots, N_L \times M_L$. When the satellites move on their orbits, they switch logical addresses in a discrete manner. Figure 2 is a paradigm of virtual network model with 66 virtual nodes, and the logical addresses of terrestrial zones from Z_1 to Z_{66} are allocated longitudinally.

Definition 1. Let τ_z be a time unit for a satellite to travel a distance that equals to the width of a zone in **Figure 2**.

Table 1. Terms and their meanings for satellite orbits [13].

Terms	Descriptions		
N_L	the number of orbits in a LEO satellite constellation		
M_L	the number of satellites per-orbit		
$S_{i,j}$	the <i>j</i> th satellite in the <i>i</i> th orbit		
v_s	the angular velocity of a LEO satellite		
Т	the period of a satellite network		
β	the latitude of the polar area border		
h	the radius of the satellite orbit		

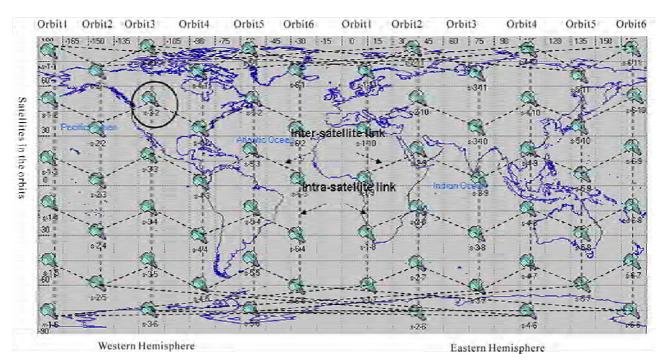


Figure 1. Topology of a π -Constellation with 6×11 satellites.

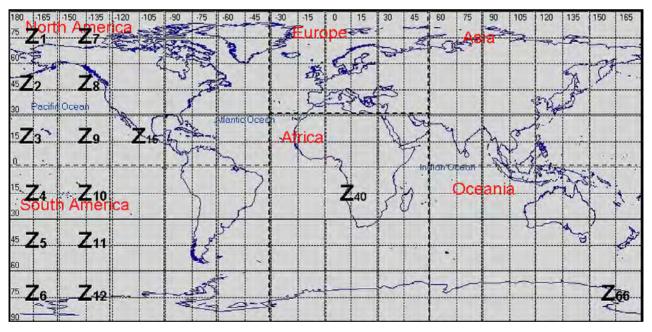


Figure 2. Virtual network model.

$$\tau_z = \frac{d_w}{v_s} \tag{1}$$

 d_w is the width of a zone and v_s is the speed of a satellite. Traffic in a zone currently is served by a given satellite $S_{i,j}$ and then will be served by satellite $S_{i,j+1}$ on the same orbit after τ_z time units. Definition 2. Let τ_c be an interval between two con-

secutive occurrences of a satellite going into or departing

a specific area one after other [18]. If M_L is even, $\tau_c = T/M_L$. It is indicated that satellite $S_{i+1,j}$ and $S_{i+1, \text{mod}(j+M_I/2, M_I)}$ are geometrically symmetric with respect to the center of the earth. On the other hand, M_L is odd, $\tau_c = T/2M_L$.

Definition 3. Let τ_d be a switch-off duration of an ISL $\overline{S_{i+1,j}S_{i,j}}$ traversing one polar area and can be cal-

culated by Equation (2) [18].

$$\tau_d = \frac{\pi - 2\beta + 4\pi/M_L}{v_s} \tag{2}$$

Definition 4. Let τ_s be the length of a snapshot in our network model. It is determined by au_c and au_d as shown in **Figure 3**. t_i and t_j are the entrance occurrence and the departure occurrence respectively.

- Case 1 $\tau_d = k\tau_c, k \in \mathbb{Z}$: a interval of topological dynamics caused by entrance and departure activities is $\tau_s = \tau_c$.
- Case 2 $\tau_d \neq k\tau_c, k \in \mathbb{Z}$: a interval consists of two distinct values τ_{smax} and τ_{smin} which can be calculated by Equation (3).

$$\begin{cases} \tau_{s\max} = \max\left(\operatorname{mod}\left(\tau_{d}, \tau_{c}\right), \tau_{c} - \operatorname{mod}\left(\tau_{d}, \tau_{c}\right) \right) \\ \tau_{s\min} = \min\left(\operatorname{mod}\left(\tau_{d}, \tau_{c}\right), \tau_{c} - \operatorname{mod}\left(\tau_{d}, \tau_{c}\right) \right) \end{cases}$$
(3)

The link state information for each satellite is collected at each snapshot. Figure 4 shows the discrete-time intervals in a LEO satellite system. The period of the link state information S is determined by the least common multiple of (τ_s, τ_z) .

4. Traffic Generation

Because of the asymmetry of the IP traffic, the user behavior and host behavior are different for each zone. We build two databases for the user density level and the host density level for each zone [19]. A distribution of traffic sources on the Earth surface is shown in Figure 5. Numbers in circles denote the user level $u_{i,i}$ (million minutes/year) corresponding to zones Z_{ij} in Figure 2. The larger the number is, the more calls are collected. It

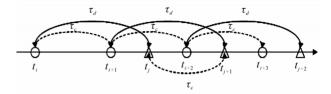


Figure 3. Determination of snapshots with τ_{\perp} and τ_{\perp} .

can be seen that satellites covering medium latitude or land mass regions will probably collect more calls from ground than satellites crossing polar or oceans regions.

According to these data, the host level $h_{i,i}$ of zone $Z_{i,i}$ can be obtained by Equation (4) [13]:

$$h_{i,j} = \frac{u_{i,j}}{\sum_{k} u(k)} \times N_h(k) \tag{4}$$

where $N_h(k)$ is the number of hosts in continent k, and $\sum u(k)$ is the sum of users' density levels of zones in

continent k.

The traffic requirement from zone $Z_{i,i}$ to zone $Z_{m,n}$ is proportional with the users' density $u_{i,j}$, host density $h_{m,n}$, and distances between centers of two zones $d(Z_{i,j}, Z_{m,n})$.

$$Traffic(Z_{i,j}, Z_{m,n}, k\tau_z) = \frac{(u_{i,j}, h_{m,n})^{\alpha}}{(d(Z_{i,j}, Z_{m,n}))^{\beta}} \quad k = 1, 2, \cdots, 11$$
(5)

 $k = 1, 2, \dots, n_c$, n_c is the number of topology changes during an orbit period. $\alpha = 0.5, \beta = 1.5$ [13]. We assume that the arrival of packets from $Z_{i,j}$ to $Z_{m,n}$ is a Poisson process with rate $\lambda(Z_{i,j}, Z_{m,n}, k\tau_z), (k = 1, 2, \dots, n_c)$ (packets/second).

 τ_s = a snapshot of a satellite network

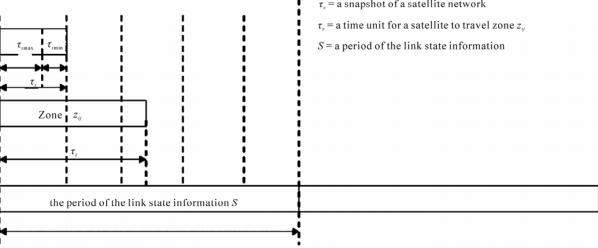


Figure 4. Discrete-time intervals.

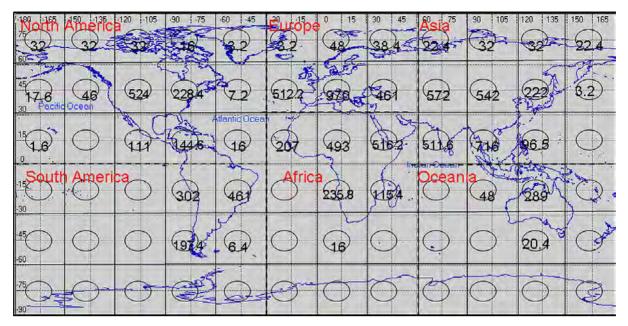


Figure 5. Discrete distributions of traffic sources.

$$\lambda(Z_{i,j}, Z_{m,n}, k\tau_z) = \frac{Traffic(Z_{i,j}, Z_{m,n}, k\tau_z)}{\sum_{\forall Z_{i,j}} \sum_{\forall Z_{m,n}} Traffic(Z_{i,j}, Z_{m,n})}$$
(6)
$$\times \frac{a}{100} \times \frac{A}{3600} \quad k = 1, 2, \dots, 11$$

a is the activity percentage in the corresponding hour [13] and *A* is the aggregate traffic that represents total traffic generated worldwide a day (packets per day).Note that arrival rate λ is not only affected by locations of the source-destination pair, but also a function of time slot.

5. Queuing Model and Queuing Delay

In a LEO satellite network, an inter-satellite link can be modeled as a finite capacity queue (see **Table 2**) [20]. **Figure 6** shows a stationary open Jackson queuing network for analysis of the queuing delay [14]. The arrival is assumed to follow Poisson process with rate λ . Service time of satellite *i* may be exponential-distributed with service rate μ_i . q_{ij} in Equation (7) is a routing probability from satellite *i* to *j*.

$$q_{id} + \sum_{i=1}^{M} q_{ij} = 1 \quad 1 \le i \le M$$
(7)

where *d* is the destination, $0 \le q_{ij} \le 1$. In real satellite networks, some of q_{ij} 's values may be zero, which represents that no packet enters into the queue. λ_i is the sum of the arrival rates into inter-satellite link *i* from all sources and can be computed by Equation (8):

$$\lambda_i = q_{si}\lambda + \sum_{k=1}^M q_{ki}\lambda_k \quad 1 \le i \le M$$
(8)

where *s* is the source and $q_{si}\lambda$ is the arrive from source. ρ_i is the severing utilization of the inter-satellite link *i*, which reflects directly traffic load on the link.

$$\rho_i = \frac{\lambda_i}{\mu_i} < 1 \tag{9}$$

Equation (8) and (9) give the relationships among in-

Table 2. Parameters of a queuing system [20].

Terms	Descriptions		
λ	arrival rate		
μ	service rate		
t _s	average time spending packets in server per packet		
t_p	average time spending packets in queue per packet		
ρ	server utilization		
L	average packet size		

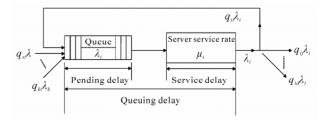


Figure 6. A finite-population queuing model.

ter-satellite links in a network.

Based on the queuing theory and Little's equation [21], the queuing delay and some parameters of ISLs' state information can be deduced by following equations. The average service rate of a satellite i can be written as:

$$\mu_i = \frac{C_0}{L} \tag{10}$$

 C_0 is an inter-satellite link's capacity and L is the average packet length. The service delay is calculated by

$$t_s = \frac{1}{\mu_i} \tag{11}$$

The pending delay in a queue is given by

$$t_p = \frac{\lambda_i}{\mu_i \left(\mu_i - \lambda_i\right)} \tag{12}$$

Finally, the queuing delay of an inter-satellite link can be obtained as:

$$E_q = t_s + t_p = \frac{1}{\mu_i - \lambda_i} = \frac{L}{C_0 - L \times \lambda_i}$$
(13)

6. Routing Algorithms

When designing the routing algorithm, a suitable linkcost metric must be introduced firstly, which can represent the information of network status. In this algorithm, there are two dynamically changing parameters, which have significant effects on the routing performance. The first one is the propagation delay, which is determined by ISLs' distances and can be calculated in advance. The second one is the queuing delay. It is affected by the traffic load on a particular satellite and its outgoing links, as the satellite's coverage traverses varying traffic zones. In this paper, in each snapshot, the link-cost metric can be calculated as:

$$L_{\cos t} = E_d + E_q \tag{15}$$

where E_d denotes the propagation delay updating with discrete interval τ_s , which takes into account the specific dynamics of the LEO satellite system. It has discussed in our previous works, considering the motion of satellites [22]. E_q denotes a predicted value of the queuing delay updating with discrete interval τ_z . Suppose a path passing *M* satellites and the queuing delay of this path can be given as:

$$E_{q}(k\tau_{z}) = \sum_{i=1}^{M} E_{qi}(t) = \sum_{i=1}^{M} \frac{1}{\mu_{i} - \lambda_{i}(k\tau_{z})} = \sum_{i=1}^{M} \frac{1/\mu_{i}}{1 - \rho_{i}(k\tau_{z})}$$
(16)

 $k = 1, 2, \dots, n_c$. Combined Equation (9) with Equation

(16), Equation (17) can be obtained:

$$E_{q}\left(k\tau_{z}\right) = \sum_{i=1}^{M} E_{q}\left(t\right) = \sum_{i=1}^{M} \frac{1}{\mu_{i} - \lambda_{i}\left(k\tau_{z}\right)}$$
$$= \sum_{i=1}^{M} \frac{1}{\mu_{i} - \left(q_{si}\lambda\left(k\tau_{z}\right) + \sum_{k=1}^{M} q_{ki}\lambda_{K}\left(k\tau_{z}\right)\right)}$$
(17)

where $k = 1, 2, \dots, n_c$, λ and λ_i , $1 \le i \le M$ is calculated by Equation (7).

This paper will modify Dijkstra's algorithm to give a result that is relevant to our case. In this algorithm, the shortest path is defined as the minimum-delay path. Here, we assumed that topology model and traffic model are changed with a discrete-time manner, which is mentioned in Section 3. Thus, a discrete-time traffic and topology adaptive routing (DT-TTAR) algorithm is proposed. It can be described as follows:

Step1. Initialize routing tables and link-cost of all satellites and ISLs.

Step 2. For $t = (k-1)\tau_z, \dots, k\tau_z$ //initiate k = 1For $t = (p-1)\tau_s, \dots, p\tau_s$ //initiate p = 1{for all the ISLs} 1) Calculate link-cost $L_{\text{costi}}(k\tau_z)$ of all the ISLs using Equation (15) with current arrival rate

 $\lambda_i(k\tau_z)$. The propagation delay $E_{di}(p\tau_s)$ can be obtained in advance and the queuing delay $E_{qi}(k\tau_z)$ is the average queuing delay in an interval $k\tau_z$.

2) An iterative calculation of a modified Dijkstra's algorithm is used to discover feasible, minimum-delay paths between each satellite and every other satellite.

3) The path-cost $P_{\cos ti}(k\tau_z)$ of a minimumdelay path *i* is the sum of link-cost of all links on the path.

- 4) If $t = p\tau_s$ p = p + 1; Update the propagation delay $E_{di}(p\tau_s)$ and the routing tables.
- 5) If $t = k\tau_Z$
 - k = k + 1;

Update arrival rate $\lambda_i(k\tau_z)$ and the routing tables.

6) Else, repeat from Step 2.

Step 3. Obtain the minimum-path routing tables and delays in every snapshot.

To compare the end-to-end delays of different schemes, one static and two adaptive shortest path algorithms are considered in this study [23]:

1) Static Shortest Path Routing (STA): When a satellite receives a packet, it sends it in *y* direction (South or North) if the satellite is not in the same latitude with the destination. Otherwise, it sends in x direction (East or West). The established route consists of two straight paths that first goes in y direction, and then in x direction (if s and d are in different longitudes and latitudes).

2) Fixed Adaptive Routing (FAR): A satellite that receives a packet, always selects dir y as the initial direction. If dir y is empty or ISL on that direction is busy, it tries dir x.

3) Random Adaptive Routing (RAR): Satellite randomly selects one of the dir y or dir x. If it is empty or ISL on that direction is busy, it tries the other direction.

7. Simulations and Results

7.1. Setting up Simulations

Iridium constellation is chosen for simulations. **Table 3** gives relative parameters of Iridium constellation. For simplicity, all ISLs are assumed to be identical (in terms of length and capacity): Capacity of each ISL is assumed to be 0.16 Gbps and a buffer is 40 Mbytes. A packet size is assumed to be 1 Kbyte.

Figure 7 gives the simulation diagram of proposed DT-TTAR algorithm, which includes all the modules and

their functions. In this paper, simulations can be completed mainly with four steps:

1) Topology model of Iridium constellation, which considers network connectivity and topology dynamics in a discrete-time manner;

2) A global traffic flow model based on different geographic user distributions;

3) A link-cost ISLs' dynamic characteristics and the traffic load on inter-satellite links;

4) A simulation model covering packets generation and forwarding, queue management and implementation of the DT-TTAR algorithms considered.

7.2. Results Analysis

The first step of experiments researches a dynamic topology of Iridium constellation. **Table 4** presents the discrete interval distributions of Iridium system under different polar border threshold β . τ_z , τ_s , $\tau_{s\min}$ and $\tau_{s\max}$ are the theoretical results according to the analysis in Section 3. In an Iridium system, the number of satellites in each orbit is odd, a snapshot τ_s separates into two parts $\tau_{s\min}$ and $\tau_{s\max}$. Different β values only determine the partition of τ_s into $\tau_{s\min}$ and $\tau_{s\max}$.

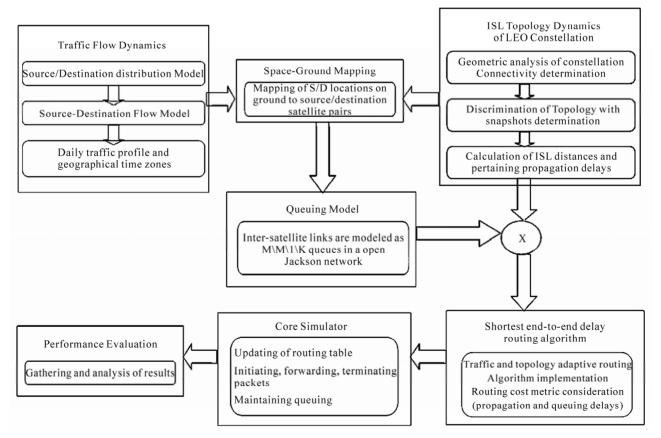


Figure 7. Simulation diagram.

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Table 3. Parameters of iridium satellite constellation [24]

LEO	Iridium		
Altitude	h = 780 km		
Number of plane	$N_L = 6$		
Number of satellite per plane	$\mathbf{M}_{\mathrm{L}}=11$		
Orbital velocity of the satellites	$\omega = 27,000 \text{ km/h}$		
Orbital Period (min)	100.4		
Inclination <i>i</i>	82.2970 (deg)		
Number of intra-plane ISLs	2		
Number of inter-plane ISLs	2		

Table 4. Discrete intervals distributions of iridium system.

System	eta(°)	$\tau_z(s)$	$\tau_s(s)$	$ au_{s\min}(s)$	$ au_{s\max}(s)$
Iridium	60	547.63	273.95	91.32	182.63
	65	547.63	273.95	7.05	266.90
	70	547.63	273.95	111.22	162.73

In this simulations, $\beta = 60^{\circ}$ is chosen. The length of snapshot τ_s is 273.95 s and it is separated into $\tau_{smax} = 182.63$ s and $\tau_{smin} = 91.32$ s. It is can be seen that an Iridium satellite travels through a traffic zone using 547.63 s, which is about twice a discrete interval 273.95 s. It means that 4 topology changes of Iridium system will occur during a satellite travelling through a traffic zone, successively at $(t + \tau_{smin})$, $(t + \tau_{smin} + \tau_{smax})$, $(t + 2\tau_{smin} + \tau_{smax})$ and $(t + 2\tau_{smin} + 2\tau_{smax})$. This model of discrete-time intervals in an Iridium system can be named as an alternating dual-topology model.

The second step of experiments analyzes arrival rates based on global traffic flow model. **Figure 8** shows the arrival rates of different traffic zones at a given time. The higher the bar is, the larger the arrival rate is. There are 6×11 zones in X-Y plane corresponding to 66 satellites in a constellation. It can be seen that distributions of arrival rates are assigned according to traffic requirements in Equation (6). Satellites covering medium latitude or land mass regions have larger arrival rates than satellites crossing oceans.

In our experiments, three inter-satellite links in different areas are chosen to observe the assignments of global traffic. The first pair is located at $(115^{\circ}E, 40^{\circ}N)$ and $(78^{\circ}E, 32^{\circ}N)$ in Asia. The second pair is located at $(45^{\circ}W, 32^{\circ}N)$ in Oceania and $(50^{\circ}W, 65^{\circ}N)$ over the Atlantic. The last link connects two satellites at $(137^{\circ}E, 20^{\circ}S)$ and $(145^{\circ}E, 32^{\circ}S)$ in Africa.

The dynamic arrival rates of these three inter-satellite

links are depicted in **Figures 9(a)**, (b) and (c), respectively. The time unit of an Iridium satellite travelling a traffic zone is about 547.63 s, so there are four parts in **Figure 9** corresponding to four consecutive traffic zones during simulation time (2200 s). Non-uniform traffic distribution is the main reason for variations of arrival rates in different parts. For example, in **Figure 9(a)**, the intersatellite link starts in Asia with high traffic requirements and the arrival rate in part 1 are about 3500 packet/sec. After 18 minutes, it goes into Oceania area in part 3 and arrival rate decreases to 250 packet/sec. In addition, we can find that there are small varies of arrival rates in each part, that is because the distances between to satellite change with time.

At last, we compare the performance of the end-to-end delay which are computed by proposed DT-TTAR algorithm and three other routing algorithms: static shortest path routing (STA), fixed adaptive routing (FAR) and random adaptive routing (RAR).

Simulations are based on the observation of the end-to-end delay between three terrestrial source-destination pairs. The source nodes of the first two pairs are both located in North America: Washington D.C. and San Jose. The destination nodes are at Sydney in Oceania and Buenos Aires in South African. The third pair is from San Paulo in South America to Cape Town in African. The path associated with the third pair does not travel through high traffic concentration areas.

To compare the performances of the end-to-end delay in different schemes, we increase the ISLs' capacity gradually. The end-to-end delay of DT-TTAR algorithm and three other routing algorithms are depicted in Fig. 10-12. It can be seen that all of the end-to-end delays are decreasing with ISL capacity increasing.

When the ISL's capacity is more than 200, the endto-end delays of four algorithms are similar, even for

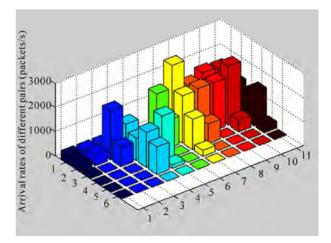
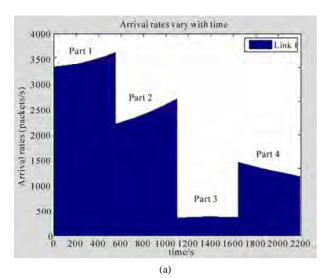
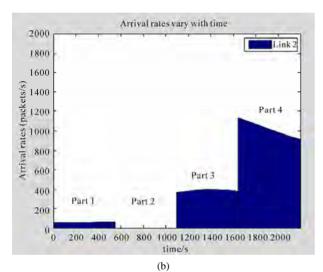


Figure 8. Global distributions of arrival rates at a given time.





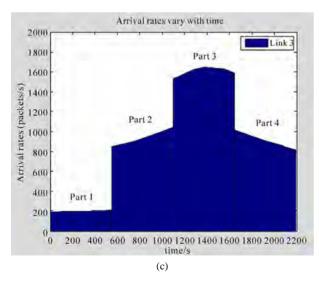


Figure 9. Arrival rates of three satellite pairs vary with time, (a) Inter-satellite link 1; (b) Inter-satellite link 2; (c) Intersatellite link 3.

the path crossing high traffic concentration area, e.g., source-destination pair 1. This is reasonable because when the ISL capacity is large enough, the queue is short even empty. However, in **Figure 10**, as the ISL capacity decreases, the delays' curves of STA, FRA and RPR deviate from the curve returned by DT-TTAR algorithm obviously. The end-to-end delays of the paths calculated by RPR and FAR increase dramatically when the ISL capacity smaller than 125. That is because when the server rates become slow, the pending queue of inter-satellite link 1 with higher traffic load tends to be very long even full more easily. Whereas the routing scheme based on DT-TTAR can choose the shorter delay paths and reduce the traffic entering into the congested areas.

The second pair has the similar curves as the first pair. DT-TTAR performs the best and STA performs the worst. The end-to-end delay of STA and RPR are very close to each other. However, in **Figure 11**, the maximum value about 0.081s is only half the value of 0.16s in **Figure 10**. Delays' curves of second pair decrease more slowly with ISL capacity increasing. That is because source-destination pair 2 goes through paths with lighter traffic than source-destination pair 1 does.

For the path only through areas with lower traffic concentration, e.g., source-destination pair 3, the delays' curves of STA, FAR and RPR are very close to DT-TTAR. Even when the ISL capacity is smaller than 125 (compare with source-destination pair 1), the end-to- end delays deviation from the DT-TTAR for STA, FAR and RPR are within 0.05 s, 0.05 s and 0.25 s, respectively. That is because the links in areas with very light traffic are seldom congested. Comparing **Figures 10**, **11** and **12**, we can find that DT-TTAR algorithm has better performance of the end-to-end delay, especially for the path that travels through high traffic areas.

8. Conclusions

In this paper, a new discrete-time traffic and topology adaptive routing (DT-TTAR) algorithm in LEO satellite networks is addressed. It is able to adapting to inherent dynamics of network topology and variations of traffic load of satellites' coverage in a discrete manner. The decision criterion of the routing algorithm depends on the link-cost metric mostly concerned by the propagation delay and the queuing delay. The propagation delay is determined by motions of the satellites. The queuing delay is evidently affected by the mechanisms of the queuing network and traffic load on ISLs. The performance evaluation has been carried out with a set of simulations.

In our simulations, DT-TTAR algorithm has better

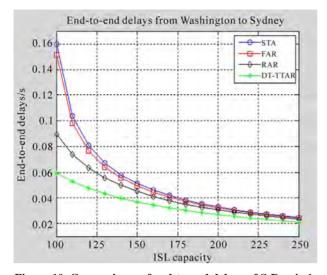


Figure 10. Comparisons of end-to-end delays of S-D pair 1.

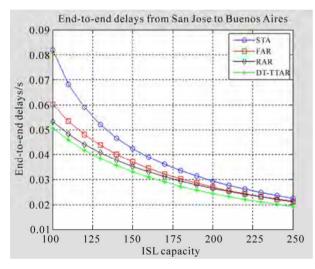


Figure 11. Comparisons of end-to-end delays of S-D pair 2.

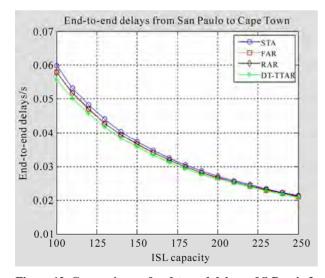


Figure 12. Comparisons of end-to-end delays of S-D pair 3.

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performance of end-to-end delay than three other shortest path algorithms: STA, FAR and RAR. Especially in some high traffic concentration areas, DT-TTAR algorithm has obvious advantages in end-to-end delay characteristics. However, our algorithm will cause the large sum of routing changing as all the satellites shift their entire traffic to the others. The more sophisticated routing algorithms are required to meet the dynamic traffic accessing against the topological coverage changing. Meanwhile, our algorithm only focuses on the performance of end-to-end delay and a more comprehensive performance evaluation of a routing algorithm should be considered. Thus, we might enhance DT-TTAR algorithm as continuous-time traffic and topology routing algorithm (CT-TTRA) in order to attain traffic load balance and the maximum throughput accurately, as well as minimum end-to-end delay.

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