

Analytical Third Order Solution for Coupling Effects of Earth Oblateness and Direct Solar Radiation Pressure on the Motion of Artificial Satellites

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Received 21 June 2014; revised 18 July 2014; accepted 14 August 2014

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Abstract

Coupling effects of Earth oblateness and direct solar radiation pressure on the motion of an artificial satellite are evaluated. Secular and periodic terms are retained up to order three and two respectively, where the coefficient J_2 of the second zonal harmonic of the geopotential is considered of first order. The solution revealed the existence of secular terms at order three that arises from the couplings between terms, of lower orders, resulting from the solar radiation pressure.

Keywords

Earth Oblateness-Solar Radiation Pressure-Artificial Satellites

1. Introduction

Analytical theories of celestial mechanics are usually more tractable when tackled within the domain of Hamiltonian mechanics, and fortunately most non-Hamiltonian systems of differential equations can be Hamiltonized by a simple technique [1] [2]. As for as canonical perturbation methods are concerned, the basic demand is a canonical transformation such that the new Hamiltonian has fewer degrees of freedom, which results in integrals of motion equal in number to the number of ingrate coordinated, and thus successive transformations reduce the system to quadratures. If the Hamiltonian is a periodic function of time, a further requirement would be the averaging of the Hamiltonian to eliminate the time.

If the Hamiltonian admits a Taylor series expansion in powers of a small parameter ε at $\varepsilon = 0$, a generator of the transformation can obtain a series of ε up to any desired power.

Many works deal with the effects of solar radiation pressure, Musen (1960) [3] derived first order expansion for the rate of change in the osculating elements using the method of variation of vector elements. Kosai (1961) and Brouwer (1962) [4] [5] used Lagrange’s planetary equations to find the first order solutions with the integration performed between the times of exit from, and entry into, the shadow. The resonance effects produced by the commensurabilities between the different mean motions provided good field for detailed theoretical studies (e.g. Musen, 1960 [3]; Brouwer, 1962 [5] and Hori, 1966 [6]). De Moraes (1981) [7] developed semi-analytical theory including the joint effects of direct solar radiation pressure and atmospheric drag for satellites of perigee heights between 500 and 900 Kms. The difficulties arising from solar radiation pressure are analyzed in three very useful and interesting expositions given by Kampos (1968) and Sehnal (1970 and 1975) [8]-[10]. An interesting application is to use solar radiation pressure as means of spacecraft propulsion (solar sailing). The idea is described and its dynamics is studied by Mc Innes and Brouwn (1990) [11].

It has been repeatedly stated by many authors (e.g. Kampos, 1968; Ferraz-Mello, 1972 and Anselono *et al.*, 1983) [8] [12] [13] that there are no secular or long periodic changes of the semimajor axis and no second order terms (or even, by some author, no secular terms whatever the order). But Geyling and Westerman (1971), Lala (1972) and Sehnal (1975) [10] [14] [15] drew attention to secular terms which appear at higher orders. McMahon, Jay W., (2011) [16] modeled the solar radiation pressure acceleration as a Fourier series which depends on the Sun’s location in a body-fixed frame; a new set of Fourier coefficients are derived for every latitude of the Sun in this frame, and the series is expanded in terms of the longitude of the Sun. Lücking *et al.* (2012) [17] further explores a passive strategy based on the joint effects of solar radiation pressure and the Earth’s oblateness acting on a high area-to-mass-ratio object. In 2001, Cook found that the most significant effect relating to solar radiation pressure is the changing cross-sectional area of the satellite projected to the Sun [18].

The present work is concerned with the effects of solar radiation pressure at higher orders to emphasize the effects of the couplings between them and with those resulting from the oblate gravity field of the Earth. The canonical equations of motion are formed in terms of the Delaunay elements augmented with the pair (k, K) where k is the mean longitude of the sun. The equations include the radiation pressure force and the geopotential up to J_2 . Two canonical transformations are made to eliminate the short and long period terms in succession respectively. The Hamiltonians and generator are assumed to be expandable as

$$\begin{aligned}
 H(\ell, g, h, k; L, G, H, K; J_2) &= \sum_{n=0}^3 \frac{J_2^n}{n!} H_n \\
 H^*(\dots, \acute{g}, \acute{h}, \acute{k}; \acute{L}, \acute{G}, \acute{H}, \acute{K}; J_2) &= \sum_{n=0}^3 \frac{J_2^n}{n!} H_n^* \\
 H^{**}(\dots; \grave{L}, \grave{G}, \grave{H}, \grave{K}; J_2) &= \sum_{n=0}^3 \frac{J_2^n}{n!} H_n^{**} \\
 W(\acute{\ell}, \acute{g}, \acute{h}, \acute{k}; \acute{L}, \acute{G}, \acute{H}, \acute{K}; J_2) &= \sum_{n=0}^3 \frac{J_2^n}{n!} W_{n+1}
 \end{aligned}$$

As expected secular terms arise at order 3 from the brackets (H_{2r}^*, W_{1r}^*) and $((H_1^*; W_{1r}^*); w_{1r}^*)$ where the suffix r refers to terms arising from solar radiation pressure.

2. Equation of Motion

In terms of the Delaunay variables $(\ell, g, h, k; L, G, H, K)$ the equations of motion of an artificial satellite about an oblate Earth and perturbed by direct solar radiation

$$\frac{d(\ell, g, h, k)}{dt} = \frac{\partial H}{\partial(L, G, H, K)}; \quad \frac{d(L, G, H, K)}{dt} = \frac{-\partial H}{\partial(\ell, g, h, k)} \tag{1}$$

With the Hamiltonian [6] [19]

$$H = \sum_{n=0}^2 \frac{J_2^n}{n!} H_n + \frac{J_2^2}{2!} \Delta H_2 \tag{2}$$

$$\begin{aligned}
 H_0 &= -\frac{\mu^2}{2L^2}, \\
 H_1 &= \frac{A_2}{L^6} \varphi^3 Z_2 + \sigma K_2, \\
 H_2 &= \frac{A_3}{L^8} \varphi^4 Z_2 + \frac{A_4}{L^{10}} \varphi^5 Z_4, \\
 \Delta H_2 &= A_1 \frac{L^2}{\varphi} \left[(1+c)(1+c_\odot) \cos F_{11}^{1-1} + (1+c)(1-c_\odot) F_{11}^{11} + (1-c)(1+c_\odot) F_{11}^{-1,1} \right. \\
 &\quad \left. + (1-c)(1-c_\odot) F_{11}^{-1,-1} + 2ss_\odot (\cos F_{11}^{-0,-1} - \cos F_{11}^{01}) \right].
 \end{aligned} \tag{3}$$

$$\begin{aligned}
 Z_2 &= (3s^2 - 2)(-3s^2 \cos F_{22}), \\
 Z_3 &= (15s^3 - 12s) \sin F_{11} - 5s^3 \sin F_{33}, \\
 Z_4 &= (24 - 120s^2 + 105s^4) + (120s^2 - 140s^4) \cos F_{22} + 35s^4 \cos F_{44}, \\
 \varphi &= \frac{a}{r}, \quad F_{ij}^{st} = if + jg + sh + tk.
 \end{aligned} \tag{4}$$

And

$$\begin{aligned}
 A_1 &= \frac{\beta_1}{\mu J_2^2 a_\odot^2}, \quad A_2 = \frac{1}{4} \mu^4 R^2, \quad A_3 = \frac{\mu^5 R^3 J_2}{32 J_2^2}, \quad A_4 = \frac{\mu^6 R^4 J_4}{32 J_2^2}, \\
 \beta_1 &= \frac{AE_\odot}{mc_\ell} (1 + \alpha) a_\odot^2 \cos^2 \chi.
 \end{aligned}$$

E_\odot is the solar constant, c_ℓ is the speed of light, a_\odot is the mean distance Earth-Sun.

$$\sigma = \frac{v}{J_2}, \tag{5}$$

$$c = \cos I, \quad c_\odot = \cos \varepsilon, \quad s_\odot = \sin \varepsilon, \quad s = \sin I.$$

v being is the mean motion of the sun, A_1 is a constant, ε is the obliquity of the ecliptic and R is the equatorial radius of the Earth.

When deriving the above equations the Sun is assumed moving in a circular orbit so that its mean longitude is $t + \text{const}$, and the direction and distance of the satellite from the Sun are considered similar to those of the Earth. Clearly ΔH_2 represents the contribution of solar radiation pressure.

3. Short-Period Perturbations

Since H_0 depends only on L , the angle ℓ will be a fast variable while the other angles are slow variables. Therefore, we shall perform two transformations to eliminate, in succession, the short and long period terms. Adopting the transformation techniques developed by Deprit (1969) and Kamel (1969) [1] [20], the identities for the short period transformation will be:

$$H_0^* = H_0 \tag{6.1}$$

$$H_n^* = \tilde{H}_n + (H_n; W_n) \tag{6.2}$$

$$\tilde{H}_n = H_n + \sum_{j=1}^{n-1} \left[C_{j-1}^{n-1} (H_{n-j}; W_j) + C_j^{n-1} G_j H_{n-j}^* \right] \tag{6.3}$$

$$G_j = L_j - \sum_{m=0}^{j-2} C_m^{j-1} L_{n+1} G_{j-m-1}; \quad 1 \leq j \leq n. \tag{6.4}$$

where $L_i = L_{W_i}$ is the Lie derivative generated by W_i and the Hamiltonian and generating function at order n are obtained from (6.2) by choosing

$$H_n^* = \langle \tilde{H}_n \rangle_{\tilde{\nu}} \tag{6.5}$$

Then

$$(W_n; H_n) = P_n = \tilde{H}_n - H_n^* \tag{6.6}$$

The elements of the transformation are obtained from

$$(\ell, g, h, k) = (\tilde{\ell}, \tilde{g}, \tilde{h}, \tilde{k}) + \sum_{n=1}^2 \frac{J_2^n}{n!} (\ell^{(n)}, \dot{g}^{(n)}, \dot{h}^{(n)}, k^{(n)}) \tag{7.1}$$

$$(L, G, H, K) = (\tilde{L}, \tilde{G}, \tilde{H}, \tilde{K}) + \sum_{n=1}^2 \frac{J_2^n}{n!} (L^{(n)}, \dot{G}^{(n)}, \dot{H}^{(n)}, \dot{K}^{(n)}) \tag{7.2}$$

$$(\tilde{\ell}, \tilde{g}, \tilde{h}, \tilde{k}) = (\ell, g, h, k) + \sum_{n=1}^2 \frac{J_2^n}{n!} (\ell^{(n)}, g^{(n)}, h^{(n)}, k^{(n)}) \tag{7.3}$$

$$(\tilde{L}, \tilde{G}, \tilde{H}, \tilde{K}) = (L, G, H, K) + \sum_{n=1}^2 \frac{J_2^n}{n!} (L^{(n)}, G^{(n)}, H^{(n)}, K^{(n)}) \tag{7.4}$$

$$(\ell^{(n)}, \dot{g}^{(n)}, \dot{h}^{(n)}, k^{(n)}) = \frac{\partial W_n}{\partial (\tilde{\ell}, \tilde{G}, \tilde{H}, \tilde{K})} + \sum_{j=1}^{n-1} \binom{n-1}{j} G_j (\ell^{(n-j)}, \dot{g}^{(n-j)}, \dot{h}^{(n-j)}, k^{(n-j)}) \tag{8.1}$$

$$(\tilde{L}^{(n)}, \tilde{G}^{(n)}, \tilde{H}^{(n)}, \tilde{K}^{(n)}) = -\frac{\partial W_n}{\partial (\tilde{\ell}, \tilde{g}, \tilde{h}, \tilde{k})} + \sum_{j=1}^{n-1} \binom{n-1}{j} G_j (L^{(n-j)}, G^{(n-j)}, H^{(n-j)}, K^{(n-j)}), \quad n \geq 1 \tag{8.2}$$

$$(\ell^{(n)}, g^{(n)}, h^{(n)}, k^{(n)}) = -(\dot{\ell}^{(n)}, \dot{g}^{(n)}, \dot{h}^{(n)}, k^{(n)}) + \sum_{j=1}^{n-1} \binom{n-1}{j} (\ell^{(n-j)}, \dot{g}^{(n-j)}, \dot{h}^{(n-j)}, k^{(n-j)}), \quad n \geq 1 \tag{8.3}$$

$$(L^{(n)}, G^{(n)}, H^{(n)}, K^{(n)}) = -(L^{(n)}, G^{(n)}, H^{(n)}, K^{(n)}) + \sum_{j=1}^{n-1} \binom{n-1}{j} (L^{(n-j)}, \dot{G}^{(n-j)}, \dot{H}^{(n-j)}, \dot{K}^{(n-j)}), \quad n \geq 1 \tag{8.4}$$

Performing the operations described in Equations (6) with the secular and periodic terms retained up to the third and second order respectively, we arrive, after some lengthy manipulations, at the following results. All variables are understood to be single primed but the primes will be dropped out for the sake of simplicity of writing.

$$H_0^* = -\frac{\mu^2}{2L^2} \tag{9}$$

$$H_1^* = \frac{A_2}{L^3 G^3} (3s^2 - 2) + \sigma k \tag{10.1}$$

$$W_1 = \frac{A_2}{\mu^3 G^3} \left\{ (3s^2 - 2)(f - \ell + e \sin F) - \frac{3}{2} s^2 \left(\sin F_{22} + e \sin F_{12} + \frac{2}{3} \sin F_{32} \right) \right\} \tag{10.2}$$

At the subsequent orders (order 2 and 3) we will be concerned only with terms arising from the radiation pressure and those due to the coupling between oblateness and radiation pressure effects. We thus separate W_2 as

$$W_2 = W_{2g} + W_{2r} \tag{11}$$

where the suffixes g and r refer to gravity and radiation pressure, respectively. At the second order we have

$$H_{2r}^* = -\frac{3}{2} A_1 A e L^2 \tag{12.1}$$

$$W_{2r} = \frac{A_1}{\mu^2} L^5 \left\{ A \left[\left(1 - \frac{1}{2} e^2 \right) \sin E - \frac{e}{4} \sin 2E \right] - B \frac{G}{L} \left[\frac{e}{4} \cos 2E - \cos E \right] \right\} \quad (12.2)$$

where:

$$A = (1+c)(1+c_\odot) \cos F_{1-1} + (1+c)(1-c_\odot) \cos F_{111} + (1-c)(1+c_\odot) \cos F_{1-11} + (1-c)(1-c_\odot) \cos F_{1-1-1} + 2ss_\odot (\cos F_{10-1} - \cos F_{101}). \quad (13.1)$$

$$B = (1+c)(1+c_\odot) \sin F_{1-1} + (1+c)(1-c_\odot) \sin F_{111} \sin F_{111} + (1-c)(1+c_\odot) \sin F_{1-11} \sin F_{1-11} + (1-c)(1-c_\odot) \sin F_{1-1-1} \sin F_{1-1-1} + 2ss_\odot (\sin F_{10-1} \sin F_{10-1} - \sin F_{101} \sin F_{101}). \quad (13.2)$$

Regarding (6.3) we find that the manipulations at order 3 require evaluating

$$\tilde{H}_3 = \sum_{i=1}^4 \tilde{H}_{3i} \quad (14)$$

where:

$$\tilde{H}_{31} = 2(H_1; W_{2r}), \quad \tilde{H}_{32} = (H_1^*; W_{2r}), \quad \tilde{H}_{33} = (H_{2r}; W_1), \quad \tilde{H}_{34} = 2(H_{2r}; W_1).$$

Evaluating these brackets and performing the averaging process in Equation (6.5) we obtain, after lengthy calculations

$$H_{31}^* = \frac{2A_1A_2}{\mu^2} \left\{ \bar{\Psi}_{11}A + \bar{\Psi}_{16}B_h + (\bar{\Psi}_{11}^cA + \bar{\Psi}_{13}^c\tilde{A} + \bar{\Psi}_{16}^cB_h) \cos 2g + (\bar{\Psi}_{12}^sB + \bar{\Psi}_{14}^s\bar{B} + \bar{\Psi}_{15}^sA_h) \sin 2g + \left(\frac{2A_1\sigma}{\mu^2} \right) \bar{\Psi}_{17}A_k \right\} \quad (15.1)$$

$$H_{32}^* = \frac{A_1A_2}{\mu^2} \left\{ \bar{\Psi}_{21}A + \bar{\Psi}_{26}B_h + \left(\frac{2A_1\sigma}{\mu^2} \right) \bar{\Psi}_{27}B_k \right\} \quad (15.2)$$

$$H_{33}^* = \frac{A_1A_2}{\mu^2} \left\{ \bar{\Psi}_{31}A + \bar{\Psi}_{36}B_h + (\bar{\Psi}_{31}^cA + \bar{\Psi}_{33}^c\tilde{A} + \bar{\Psi}_{36}^cB_h) \cos 2g + (\bar{\Psi}_{32}^sB + \bar{\Psi}_{34}^s\bar{B} + \bar{\Psi}_{35}^sA_h) \sin 2g \right\} \quad (15.3)$$

$$H_{34}^* = \frac{A_1A_2}{\mu^2} \left\{ (\bar{\Psi}_{41}^cA + \bar{\Psi}_{43}^c\tilde{A}) \cos 2g + (\bar{\Psi}_{42}^sB + \bar{\Psi}_{45}^sA_h) \sin 2g \right\} \quad (15.4)$$

So that

$$H_3^* = \frac{A_1A_2}{\mu^2} \left\{ \bar{\Psi}_1A + \bar{\Psi}_6B_h + (\bar{\Psi}_1^cA + \bar{\Psi}_3^c\tilde{A} + \bar{\Psi}_6^cB_h) \cos 2g + (\bar{\Psi}_2^sB + \bar{\Psi}_4^s\bar{B} + \bar{\Psi}_5^sA_h) \sin 2g + \left(\frac{2A_1\sigma}{\mu^2} \right) (2\bar{\Psi}_{17}A_k + \bar{\Psi}_{27}B_k) \right\}. \quad (16)$$

where the $\bar{\Psi}'_i$ s are functions of $(\hat{L}, \hat{G}, \hat{H})$, the subscripts h and k mean partial derivatives with respect to each, and

$$\tilde{A} = -c \left\{ (1+c_\odot) \cos F_{1-1} + (1-c_\odot) \cos F_{111} - (1+c_\odot) \cos F_{1-11} - (1-c_\odot) \cos F_{1-1-1} - 2 \frac{cs_\odot}{s} (\cos F_{10-1} - \cos F_{101}) \right\} \quad (17.1)$$

$$\tilde{B} = -c \left\{ (1+c_\odot) \sin F_{1-1} + (1-c_\odot) \sin F_{111} - (1+c_\odot) \sin F_{1-11} - (1-c_\odot) \sin F_{1-1-1} - 2 \frac{cs_\odot}{s} (\sin F_{10-1} - \sin F_{101}) \right\} \quad (17.2)$$

For the elements of the transformations, substituting Equations (10.2) and (12.2) into Equation (8) yields

$$\hat{\iota}^{(2)} = \frac{A_1}{\mu^2} \left\{ A \left[\left(-\frac{5}{4} G^2 L^2 \varphi + 2G^2 L^2 + 5L^4 (1+e^2) \right) \sin E + \left((1+e^2) \frac{G^2 L^2}{2e} \varphi + \frac{1}{4e} G^2 L^2 - \frac{5e}{4} L^4 \right) \sin 2E \right] \right\}$$

$$-\left(\frac{L^2 G^2 \varphi}{4}\right) \sin 3E + \frac{3G^2 L^2}{2e}(\ell - E) + \frac{15}{2} L^4 e(\ell - E) + B \left[\left(\frac{G^2 L}{4} \varphi + 4GL^3\right) \cos E \right. \tag{18.1}$$

$$\left. + \left(\frac{G^3 L \varphi}{2e} - \frac{GL^3}{4e} - \frac{3e}{4} L^3 G\right) \cos 2E - \left(\frac{1}{4} G^3 L \varphi\right) \cos 3E + \frac{G^3 L \varphi}{2e} \right]$$

$$\begin{aligned} \dot{g}^{(2)} = \frac{L^5}{\mu^2} & \left\{ \left[A_1 \left(\frac{G}{L^2} \left(\frac{5}{4} \varphi - 2 \right) \right) + A_G (1 + e) \right] \sin E + \left[A \left(\frac{G}{2eL^2} \left(\frac{1}{2} - (1 + e^2) \varphi \right) \right) - \frac{e}{4} A_G \right] \sin 2E + A \frac{G\varphi}{4L^2} \sin 3E \right. \\ & + \left[B \left(-\frac{G^2 \varphi}{4L^3} + \frac{1}{4} \right) + B_G \frac{G}{L} \right] \cos E + \left[B \left(-\frac{G^2 \varphi}{4eL^2} + \frac{G^2}{4eL^3} - \frac{e}{4L} \right) - B_G \frac{eG}{4L} \right] \cos 2E \\ & \left. + B \frac{G^2 \varphi}{4L^3} \cos 3E + B \frac{G^2 \varphi}{2eL} - A \frac{3G}{2eL^3} (\ell - E) \right\}. \end{aligned} \tag{18.2}$$

$$\hat{h}^{(2)} = \frac{L^5}{\mu^2} A_1 \left\{ A_H \left[(1 + e^2) \sin E - \frac{e}{4} \sin 2E + \frac{3}{2} e(\ell - E) \right] + B_H \frac{G}{L} \left(\frac{e}{4} \cos 2E - \cos E \right) \right\} \tag{18.3}$$

$$\hat{l}^{(2)} = \frac{L^5}{\mu^2} A_1 \left\{ A \left[(1 + e^2) \varphi \sin E - \frac{e}{4} \varphi \cos 2E + \frac{3}{2} e(1 - \varphi) \right] + B \frac{\varphi}{\eta} \left(\frac{e}{2} \sin 2E - \sin E \right) \right\} \tag{19.1}$$

$$\hat{G}^{(2)} = \frac{L^5}{\mu^2} A_1 \left\{ B \left[(1 + e^2) \sin E - \frac{e}{4} \sin 2E + \frac{3}{2} e(\ell - E) \right] + A \frac{G}{L} \left(\frac{e}{4} \cos 2E - \cos E \right) \right\} \tag{19.2}$$

$$\hat{H}^{(2)} = \frac{L^5}{\mu^2} A_1 \left\{ A_h \left[(1 + e^2) \sin E - \frac{e}{4} \sin 2E + \frac{3}{2} e(\ell - E) \right] - B_h \frac{G}{L} \left(\frac{e}{4} \cos 2E - \cos E \right) \right\} \tag{19.3}$$

$$\hat{K}^{(2)} = \frac{A_1}{\mu^2} \left\{ A_k L^5 \left[(1 + e^2) \sin E - \frac{e}{4} \sin 2E - \frac{3}{2} e(\ell - E) \right] - B_k L^4 G \left(\frac{e}{4} \cos 2E - \cos E \right) \right\} \tag{19.4}$$

where:

$$A_H = -\frac{\hat{A}}{c_G}, \quad B_H = -\frac{\hat{B}}{c_G}$$

where we note that $\hat{l}^{(1)}, \dots, \hat{k}^{(1)}$ consists only of terms arising from the oblateness effects and therefore need be taken into consideration.

Substitution of Equations (18) and (19) into (7) yields the transformation and inverse.

4. Perturbations of Long Period

After the short-period terms have been eliminated the problem is now reduced to the system of canonical equations with Hamiltonian $H^*(\dots, \dot{g}, \dot{h}, \dot{k}; \hat{L}, \hat{G}, \hat{H}, \hat{K})$.

We now proceed to eliminate the long-period terms, *i.e.* those periodic in $\dot{g}, \dot{h}, \dot{k}$.

The transformation follows the procedure outlined in Equations (6)-(8) but with the averages and integrations, performed over \dot{g}, \dot{h} and \dot{k} . The basic identities are now

$$H_0^{**} = H_0^* \tag{20.1}$$

$$H_0^{**} = H_0^* + \sum_{j=1}^{n-1} \left\{ \binom{n-1}{j-1} (H_{n-j}^*; w_j^*) + \binom{n-1}{j} G_j^* H_{n-1}^* \right\}; \quad (n \geq 1) \tag{20.2}$$

Performing the processes outlined in Equations (16.1), ..., (16.6) and (20) to find H^{**} and w^* assuming no resonant conditions the following results follow (all variables are double primed)

4.1. Zero, First and Second Orders

Remembering that at orders 2 and 3, we are concerned only with the terms including radiation pressure effects we have

$$H_0^{**} = -\frac{\mu^2}{2L^2} \quad (21)$$

$$H_1^{**} = \frac{A_2}{L^3 G^3} (3s^2 - 2) + \sigma K \quad (22)$$

$$H_2^{**} = 0 \quad (\text{apart from oblateness perturbations}) \quad (23)$$

$$w_{1r}^* = -\frac{3}{4} A_1 e L^2 \sum_{i=-1}^1 \sum_{j=-1,1} B_{1ij} \sin(g + ih + jk) \quad (24)$$

where:

$$B_{1-1-1} = \frac{(1-c)(1-c_\odot)}{n_2 - n_3 - n_4}$$

$$B_{1-11} = \frac{(1-c)(1-c_\odot)}{n_2 - n_3 - n_4}$$

$$B_{1-11} = \frac{(1-c)(1-c_\odot)}{n_2 - n_3 - n_4}$$

$$B_{10-1} = \frac{2ss_\odot}{n_2 - n_4}$$

$$B_{101} = \frac{2ss_\odot}{n_2 - n_4}$$

$$B_{11-1} = \frac{(1+c)(1+c_\odot)}{n_2 + n_3 - n_4}$$

$$B_{111} = \frac{(1+c)(1-c_\odot)}{n_2 + n_3 + n_4}$$

And

$$n_2 = \frac{\partial H_1^*}{\partial G} = \frac{A_2(15c^2 - 3)}{L^3 G^4}, \quad n_3 = \frac{\partial H_1^*}{\partial H} = -\frac{6A_2 c}{L^3 G^4}, \quad n_4 = \frac{\partial H_1^*}{\partial K} = \sigma$$

4.2. Third Order

From (20.2)

$$H_3^{**} = \tilde{H}_3^* + 3(H_1^*; W_2^*) \quad (25.1)$$

$$\tilde{H}_3^* = \sum_{i=0}^3 \tilde{H}_{3i}^* \quad (25.2)$$

where:

$$\tilde{H}_{30}^* = H_{30}^*, \quad \tilde{H}_{31}^* = -((H_1^*; W_1^*); W_1^*), \quad (26)$$

$$\tilde{H}_{32}^* = (H_2^*; W_1^*), \quad \tilde{H}_{33}^* = 2(H_2^{**}; W_1^*).$$

So, we can write

$$\tilde{H}_3^{**} = \sum_{i=0}^3 \tilde{H}_{3i}^{**}; \tilde{H}_3^{**} = \langle \tilde{H}_{3i}^* \rangle_{\hat{g}\hat{h}\hat{k}}; W_2^* = \sum_{i=0}^3 W_{2i}^* \quad (27)$$

By following the above scheme, the expressions for \tilde{H}_3^{**} and W_2^* can be deduce after a lengthily manipulation.

4.3. There Order

4.3.1. Expressions for H_{30}^{**} and w_{20}^*

From (16), (26) and (27)

$$H_{30}^{**} = 0 \quad (28.1)$$

Then

$$W_{20}^* = \sum_{n=-1}^3 \sum_{i=-1}^1 \sum_{j=-1,1} \left\{ \gamma_{nij}^s \sin F_{nij} + \gamma_{nij}^c \cos F_{lij}^c \right\} \quad (28.2)$$

Where:

$$\gamma_{nij}^s = \frac{1}{3} \frac{A_1 A_2}{\mu^2} \frac{\Psi_{nij}^*}{nn_2 + in_3 + jn_4}, \quad (n = -1, 3),$$

$$\gamma_{nij}^c = \frac{1}{3} \frac{A_1 \sigma}{\mu^2} \frac{X_{nij}^c}{nn_2 + in_3 + jn_4}, \quad (n = 1).$$

Clearly, Equation (28.2) holds as long as

$$nn_2 + in_3 + jn_4 > J_2^{1/2} \quad (i = \pm 1, 0; j = -1, 1)$$

4.3.2. Expressions for H_{31}^{**} and Writing

$$W_1^* = W_{1g}^* + W_{1r}^* \quad (29)$$

Equation (26) can be decomposed such that

$$\begin{aligned} \tilde{H}_{31}^* &= \tilde{H}_{31rr}^* + \tilde{H}_{31rg}^* + \tilde{H}_{31gr}^* \\ \tilde{H}_{31rr}^* &= -\left((H_1^*; W_{1r}^*); W_{1r}^* \right) \\ \tilde{H}_{31rg}^* &= -\left((H_1^*; W_{1r}^*); W_{rg}^* \right) \\ \tilde{H}_{31gr}^* &= -\left((H_1^*; W_{1g}^*); W_{1r}^* \right) \end{aligned} \quad (30)$$

Regarding H_{31rr}^{**} and W_{21rr}^* , Equations (25) and (27) yield

$$\tilde{H}_{31rr}^* = \frac{3}{8} A_1 \sum_{i=-1}^1 \sum_{j=-1,1} \sum_{s=-1}^1 \sum_{t=-1,1} \left\{ R_{1ijst} \cos[(i-s)h + (j-t)k] + \tilde{R}_{1ijst} \cos[2g + (i+s)h + (j+t)k] \right\} \quad (31)$$

$$H_{31rr}^{**} = R_{100} \quad \text{at } n, m = 0 \quad (32)$$

$$W_{21rr}^* = \sum_{n=-2}^2 \sum_{m=-2}^2 \left\{ \tilde{R}_{1nm} \sin F_{0nm} + \tilde{R}_{1nm} \sin F_{2nm} \right\} \quad (33)$$

where

$$\begin{aligned} R_{1ijst} &= B_{lij} (d - ib + jC) \left(-eL^2 B_{1st;G} + \frac{G}{e} B_{1st} \right) - ieL^2 B_{lij} B_{1st;H} (d - ib - jC) \\ &\quad - eL^2 s B_{1st} \left[B_{lij;H} (d - ib + iC) + B_{lij} (d_H - ib_H + jC_H) \right], \\ \tilde{R}_{1ijst} &= -B_{lij} (d - ib + jC) \left(-eL^2 B_{1st;G} + \frac{G}{e} B_{1st} \right) + ieL^2 B_{lij} B_{1st;H} (d - ib - jC) \\ &\quad - eL^2 s B_{1st} \left[B_{lij;G} (d - ib + iC) + B_{lij} (d_G - ib_G + jC_G) \right] \\ &\quad - eL^2 s B_{1st} \left[B_{lij;H} (d - ib + iC) + B_{lij} (d_H - ib_H + jC_H) \right], \end{aligned}$$

$$d = \frac{3 A_1 A_2 e}{4 L G^2} (15c^2 - 3), \quad b = \frac{9 A_1 A_2 e c}{2 L G^4}, \quad C = \frac{3}{4} A_1 \sigma e L^2$$

$$R'_{1nm} = \frac{A_1}{8} \frac{R_{1nm}}{n n_3 + m n_4}$$

$$\tilde{R}'_{1nm} = \frac{A_1}{8} \frac{\tilde{R}_{1nm}}{2n_2 + n n_3 + m n_4}$$

And the primes on the summation signs in Equation (33) indicate that the terms R_{100} is excluded. Clearly (33) holds as long as

$$i n_2 + n n_3 + m n_4 > J_2^{1/2} \quad (i = 0, 1) = 0$$

For H_{31rg}^{**} and H_{21rg}^* we have

$$H_{31rg}^{**} = 0 \tag{34}$$

$$W_{21rg}^* = \sum_{n=-2}^4 \sum_{i=-1}^1 \sum_{j=-1,1} \left\{ \hat{S}_{1nij} \sin F_{nij} + \hat{T}_{1nij} \cos F_{nij} \right\} \tag{35}$$

where

$$\hat{S}_{1nij} = \frac{1}{3} \frac{S_{1nij}}{n n_2 + i n_3 + j n_4}$$

$$\hat{T}_{1nij} = \frac{1}{3} \frac{T_{1nij}}{n n_2 + i n_3 + j n_4}$$

And

$$S_{1-ij} = \frac{1}{2} \left\{ (d - ib + jC) \left[B_{ij} (W_{12;G}^* + iW_{12;H}^*) + 2B_{ij;G} W_{12}^* \right] + 2(d_G - ib_G + jC_G) B_{ij} W_{12}^* \right\}$$

$$S_{13ij} = \frac{1}{2} \left\{ (d - ib + jC) \left[B_{1g} (-W_{12;G}^* - iW_{12;H}^*) + 2B_{ij;G} W_{12}^* \right] + 2(d_G - ib_G + jC_G) B_{ij} W_{12}^* \right\}$$

$$S_{1nij} = 0 \quad (n = -2, 0, 1, 2, 4)$$

$$T_{1-2ij} = \frac{1}{2} \left\{ (d - ib + jC) \left[B_{ij} (W_{13;G}^* + iW_{13;H}^*) + 3B_{ij;G} W_{13}^* \right] + 3(d_G - ib_G + jC_G) B_{ij} W_{13}^* \right\}$$

$$T_{10ij} = \frac{1}{2} \left\{ (d - ib + jC) \left[B_{ij} (W_{11;G}^* + iW_{11;H}^*) + B_{ij;G} W_{11}^* \right] + (d_G - ib_G + jC_G) B_{ij} W_{11}^* \right\}$$

$$T_{12ij} = \frac{1}{2} \left\{ (d - ib + jC) \left[B_{ij} (W_{11;G}^* + iW_{11;H}^*) - B_{ij;G} W_{11}^* \right] - (d_G - ib_G + jC_G) B_{ij} W_{11}^* \right\}$$

$$T_{14ij} = \frac{1}{2} \left\{ (d - ib + jC) \left[B_{ij} (W_{13;G}^* + iW_{13;H}^*) - 3B_{ij;G} W_{11}^* \right] - 3(d_G - ib_G + jC_G) B_{ij} W_{13}^* \right\}$$

$$T_{1nij} = 0 \quad (n \text{ odd})$$

$$W_{11}^* = \frac{1}{4} \frac{A_3}{A_2} \frac{e s}{G}$$

$$W_{12}^* = \frac{A_2}{32\mu^2} \frac{1}{1-5c^2} (-1 + 16c^2 - 15c^4) \left(\frac{1}{L^2 G} - \frac{1}{G^3} \right) + \frac{5}{65} \frac{A_4}{A_2} \frac{1}{1-5c^2} (-1 + 8c^2 - 7c^4) \left(\frac{1}{L^2 G} - \frac{1}{G^3} \right)$$

Equation (35) holds true as long as

$$n n_2 + i n_3 + j n_4 > J_2^{1/2}$$

Finally, for H_{31gr}^{**} and H_{21gr}^* , we have

$$H_{31gr}^{**} = 0 \tag{36}$$

$$W_{21gr}^* = \sum_{n=-2}^4 \sum_{i=-1}^1 \sum_{j=-1,1} \left\{ \dot{M}_{1nij} \sin F_{nij} + \dot{N}_{1nij} \cos F_{nij} \right\} \tag{37}$$

where:

$$\dot{M}_{1nij} = \frac{1}{3} \frac{M_{1nij}}{nn_2 + in_3 + jn_4}$$

$$\dot{N}_{1nij} = \frac{1}{3} \frac{N_{1nij}}{nn_2 + in_3 + jn_4}$$

And

$$M_{1-1ij} = \frac{3}{8} A_1 \left\{ B_{ij} \left[\frac{2G}{e} W_2^c - eL^2 (W_{2;G}^c + iW_{2;H}^c) \right] - 2eL^2 B_{ij;G} W_2^c \right\}$$

$$M_{13ij} = \frac{3}{8} A_1 \left\{ B_{ij} \left[\frac{-2G}{e} W_2^c - eL^2 (W_{2;G}^c + iW_{2;H}^c) \right] + 2eL^2 B_{ij;G} W_2^c \right\}$$

$$M_{1nij} = 0 \quad (n = -2, 0, 1, 2, 4)$$

$$N_{1-2ij} = -\frac{3}{8} A_1 \left\{ B_{ij} \left[\frac{9G}{e} W_2^s - 3eL^2 (W_{3;G}^s + iW_{3;H}^s) \right] - 9eL^2 B_{ij;G} W_2^s \right\}$$

$$N_{10ij} = -\frac{3}{8} A_1 \left\{ B_{ij} \left[\frac{G}{e} W_2^s - eL^2 (W_{1;G}^s + iW_{1;H}^s) \right] - eL^2 B_{ij;G} W_1^s \right\}$$

$$N_{12ij} = -\frac{3}{8} A_1 \left\{ B_{ij} \left[\frac{G}{e} W_1^s - eL^2 (W_{1;G}^s + iW_{1;H}^s) \right] - eL^2 B_{ij;G} W_1^s \right\}$$

$$T_{14ij} = -\frac{3}{8} A_1 \left\{ B_{ij} \left[\frac{9G}{e} W_3^s + 3eL^2 (W_{3;G}^s + iW_{3;H}^s) \right] - 9eL^2 B_{ij;G} W_3^s \right\}$$

$$N_{1nij} = 0, \quad (n \text{ odd})$$

Again, Equation (37) holds true as long as the above conditions hold. Collecting we find that

$$H_{31}^{**} = R_{100} \quad \text{at } n, m = 0 \tag{38}$$

$$W_{21}^* = W_{21rr}^* + W_{21rg}^* + W_{21gr}^* \tag{39}$$

4.3.3. Expressions for H_{32}^{**} and ω_{22}^{**}

$$H_{32rr}^{**} = R_{200} \tag{40}$$

$$W_{22rr}^* = \sum_{n=-2}^2 \sum_{m=-2}^2 \left\{ \dot{R}_{1nm} \sin F_{0nm} + \tilde{R}_{1nm} \sin F_{2nm} \right\} \tag{41}$$

where

$$\dot{R}_{2nm} = \frac{1}{3} \frac{R_{2nm}}{nn_3 + mn_4}$$

$$\tilde{R}_{2nm} = \frac{1}{3} \frac{\tilde{R}_{1nij}}{2n_2 + nn_3 + mn_4}$$

$$R_{2nm} = \frac{3}{8} A_1 \sum_{s=-1}^1 \sum_{t=-1,1} \left\{ \alpha_{st} \left[eL^2 (B_{1,n+s,m+t;G} + sB_{1,n+s,m+t;H}) - \frac{G}{e} B_{n+s,m+t} \right] + B_{1,n+s,m+t} [\alpha_{st;G} + (n+s)\alpha_{st;H}] \right\}$$

$$\tilde{R}_{2nm} = \frac{3}{8} A_1 \sum_{s=-1}^1 \sum_{t=-1,1} \left\{ \alpha_{st} \left[-eL^2 (B_{1,n-s,m-t;G} - sB_{1,n-s,m-t;H}) + \frac{G}{e} B_{n-s,m-t} \right] + B_{1,n-s,m-t} [\alpha_{st;G} + (n-s)\alpha_{st;H}] \right\}$$

$$\alpha_{-1-1} = -\frac{3}{2} A_1 e \eta_{-20} (1-c)(1-c_{\odot})$$

$$\alpha_{-11} = -\frac{3}{2} A_1 e \eta_{-20} (1-c)(1+c_{\odot})$$

$$\alpha_{0-1} = -3A_1 e \eta_{-2} s s_{\odot}$$

$$\alpha_{01} = 3A_1 e \eta_{-2} s s_{\odot}$$

$$\alpha_{1-1} = -\frac{3}{2} A_1 e \eta_{-20} (1+c)(1+c_{\odot})$$

$$\alpha_{11} = -\frac{3}{2} A_1 e \eta_{-20} (1+c)(1-c_{\odot})$$

And

$$\eta_{ij} = L^{-i} G^{-j}$$

For H_{32rg}^{**} and H_{22rg}^{**} we have

$$H_{32rg}^{**} = 0 \tag{42}$$

$$W_{22rg}^* = \sum_{n=-2}^4 \sum_{i=-1}^1 \sum_{j=-1,1} \left\{ S_{1nij}^* \sin F_{nij} + T_{1nij}^* \cos F_{nij} \right\} \tag{43}$$

$$S_{2nij}^* = \frac{1}{3} \frac{S_{2nij}}{nn_2 + in_3 + jn_4}$$

$$T_{2nij}^* = -\frac{1}{3} \frac{T_{2nij}}{2n_2 + nn_3 + mn_4}$$

$$S_{2-1ij} = \frac{1}{2} (W_{12;G}^* + iW_{12;H}^*) \alpha_{ij} - W_{12}^* \alpha_{ij;G}$$

$$S_{23ij} = \frac{1}{2} (W_{12;G}^* + iW_{12;H}^*) \alpha_{ij} - W_{12}^* \alpha_{ij;G}$$

$$S_{2nij} = 0 \quad (n = -2, 0, 1, 2, 4)$$

$$T_{1-2ij} = -\frac{1}{2} (W_{13;G}^* + iW_{13;H}^*) \alpha_{ij} - \frac{3}{2} W_{32}^* \alpha_{ij;G}$$

$$T_{20ij} = -\frac{1}{2} (W_{11;G}^* + iW_{11;H}^*) \alpha_{ij} - \frac{1}{2} W_{11}^* \alpha_{ij;G}$$

$$T_{22ij} = -\frac{1}{2} (W_{11;G}^* + iW_{11;H}^*) \alpha_{ij} + \frac{1}{2} W_{11}^* \alpha_{ij;G}$$

$$T_{24ij} = -\frac{1}{2} (W_{11;G}^* + iW_{11;H}^*) \alpha_{ij} + \frac{3}{2} W_{11}^* \alpha_{ij;G}$$

$$T_{2nij} = 0 \quad (n \text{ odd})$$

Finally, for H_{32gr}^{**} and W_{23gr}^* we have

$$H_{32gr}^{**} = 0 \tag{44}$$

$$W_{22gr}^* = \sum_{n=-2}^4 \sum_{i=-1}^1 \sum_{j=-1,1} \left\{ M_{2nij}^* \sin F_{nij} + N_{2nij}^* \cos F_{nij} \right\} \tag{45}$$

where:

$$M_{2nij}^* = \frac{1}{3} \frac{M_{2nij}}{nn_2 + in_3 + jn_4}$$

$$N_{2nij}^* = \frac{1}{3} \frac{N_{2nij}}{nn_2 + in_3 + jn_4}$$

$$M_{2-ij} = \frac{3}{8} A_1 \left\{ B_{ij} \left[eL^2 (H_{22;G}^* + iH_{22;H}^*) - \frac{2G}{e} H_{22}^* \right] + 2eL^2 B_{ij;G} H_{22}^* \right\}$$

$$M_{23ij} = \frac{3}{8} A_1 \left\{ B_{ij} \left[eL^2 (H_{22;G}^* + iH_{22;H}^*) + \frac{2G}{e} H_{22}^* \right] - 2eL^2 B_{ij;G} H_{22}^* \right\}$$

$$M_{2ij} = 0 \quad (n = -2, 0, 1, 2, 4)$$

$$N_{2-2ij} = \frac{3}{8} A_1 \left\{ B_{ij} \left[-eL^2 (H_{23;G}^* + iH_{23;H}^*) + \frac{3G}{e} H_{23}^* \right] - 3eL^2 B_{ij;G} H_{23}^* \right\}$$

$$N_{20ij} = \frac{3}{8} A_1 \left\{ B_{ij} \left[-eL^2 (H_{21;G}^* + iH_{21;H}^*) + \frac{G}{e} H_{21}^* \right] - eL^2 B_{ij;G} H_{21}^* \right\}$$

$$N_{22ij} = \frac{3}{8} A_1 \left\{ B_{ij} \left[eL^2 (H_{21;G}^* + iH_{21;H}^*) + \frac{G}{e} H_{21}^* \right] - eL^2 B_{ij;G} H_{21}^* \right\}$$

$$N_{24ij} = \frac{3}{8} A_1 \left\{ B_{ij} \left[eL^2 (H_{23;G}^* + iH_{23;H}^*) + \frac{3G}{e} H_{21}^* \right] - eL^2 B_{ij;G} H_{21}^* \right\}$$

$$H_{20}^* = \frac{3}{10} \frac{A_2^2}{\mu^2} \left\{ (5 - 10c^2 + 35c^4) \eta_{37} - (4 - 24c^2 + 36c^4) \eta_{46} - (5 - 18c^2 + 5c^4) \eta_{55} \right\} + \frac{3}{32} A_4 (15 - 150c^2 + 137c^4) \eta_{37} - (9 - 90c^2 + 105c^4) \eta_{55},$$

$$H_{21}^* = \frac{3}{2} A_3 e \eta_{55} (-45 + 5S^3)$$

$$H_{22}^* = \frac{3}{8} \frac{A_2^2}{\mu^2} (-1 + 16c^2 - 15c^4) (\eta_{37} - \eta_{55}) + \frac{15}{16} A_4 (-1 + 8c^2 - 7c^4) (\eta_{37} - \eta_{55})$$

$$H_{23}^* = 0$$

4.3.4. Expressions for H_{33}^{**} and w_{23}^*

As $H_{2r}^{**} = 0$ Equation (23) and Equation (26) reduce to

$$\tilde{H}_{33}^* = 2(H_{2g}^{**}; W_{1r}^*)$$

So that

$$H_{33}^{**} = 0$$

$$W_{23}^* = \sum_{i=-1}^1 \sum_{j=-1,1} \dot{\gamma}_{1ij} \sin F_{ij}$$

where:

$$\dot{\gamma}_{1ij} = \frac{1}{3} \frac{\gamma_{1ij}}{n_2 + in_3 + jn_4}$$

$$\gamma_{1ij} = \frac{3}{4} A_1 eL^2 \{H_{20;G}^* + iH_{20;H}^*\} B_{1ij}$$

The elements of transformation are easily obtained using Equations (7), (8) with W replaced by W^* , the unprimed elements replaced by single primed elements, and with the single primed elements replaced by double primed ones.

5. Effects of the Earth’s Shadow

In the proceeding developments, effects of entry into, or exit from, the Earth’s shadow are not taken into considerations.

To account for the shadow effects, we note that:

- 1) The integration processes to find the generators are not (directly) affected since they are more quadrates without integration limits.
- 2) What is affected and, indirectly, affects derivation of the generator is averaging process over the mean anomaly, performed to obtain the transformed Hamiltonian. To account for the shadow, let the time of exit from the shadow be t_1 where the mean anomaly is ℓ_1 and let those corresponding to entrance into the shadow be t_2 and ℓ_2 . Then the averages are to be replaced by definite integrals between these 2 limits.

6. Secular Perturbations and Computation of Position and Velocity

After performing the long period transformation, the equation of motions are now reduced to

$$\frac{dU}{dt} = - \frac{\partial H^{**}}{\partial u^{**}} = 0$$

$$\frac{du}{dt} = \frac{\partial H^{**}}{\partial U^{**}} = \alpha$$

where α are arbitrary constants

$$U^{**} = U^{**}_0$$

$$u^{**} = u^{**}_0 + ct$$

where the constants (u^{**}_0, U^{**}_0) are be determined from the initial conditions.

Let (u_0, U_0) be the values of the elements at a given initial epoch t_0 , then (u^{**}_0, U^{**}_0) can be determined as follows:

- 1) From the elements of the inverse transformations we compute

$$u' = u^{**} + \sum_{n=1}^2 \frac{J_2^n}{n!} u^{** (n)}$$

$$U' = U^{**} + \sum_{n=1}^2 \frac{J_2^n}{n!} U^{** (n)} e$$

$$u = u' + \sum_{n=1}^2 \frac{J_2^n}{n!} u^{(n)}$$

$$U = U' + \sum_{n=1}^2 \frac{J_2^n}{n!} U^{(n)}$$

- 2) Having determined $u = (\ell, g, h, k)$ and $U = (L, G, H, K)$ at time t , we can compute the position and velocity by any known method.

7. Conclusion

Equation (32) reveals that, contrary to previous concept e.g. Kampos (1968) [8], the direct solar radiation pressure produces secular effects at order 3 as long as the satellite is apart from the earth's shadow. This emphasizes the importance of taking the effects of the solar radiation pressure into account in high order artificial satellite theories, particularly for those with high area-to-mass ratio flying at high altitudes.

Again, Equation (41) is valid away from the resonant conditions, *i.e.* as long as $(in_2 + nn_3 + mn_4) > J_2^{1/2}$, ($i = 0, 1$), while (40) provides another secular part of H_3^{**} to be added to R_{100} .

Acknowledgements

The author is grateful to the anonymous referee for helpful comments.

References

- [1] Kamel, A.A. (1971) Lie Transformation and the Hamiltonian of Non-Hamiltonian Systems. *Celestial Mechanics*, **4**, 397-405. <http://link.springer.com/article/10.1007%2FBF01231400>
- [2] Giacaglia, G.E.O. (1972) Perturbation Methods in Non-Linear Systems. Springer-Verlag, New York. <http://dx.doi.org/10.1007/978-1-4612-6400-2>
- [3] Musen, P.J. (1960) The Influence of the Solar Radiation Pressure on the Motion of an Artificial Satellite. *Journal of Geophysical Research*, **6**, 1391-1396. <http://onlinelibrary.wiley.com/doi/10.1029/JZ065i005p01391/abstract>
- [4] Kozai, Y. (1961) Effects of Solar Radiation Pressure on the Motion of an Artificial Satellite. *Smithsonian Astrophys. Obs. Spec. Re.*, **56**, 25-33.
- [5] Brouwer, D. (1962) Dynamics of Satellites (Roy, M., Ed.). Springer, Berlin.
- [6] Hori, G. (1966) Theory of General Perturbation with Unspecified Canonical Variable. *Publications of the Astronomical Society of Japan*, **18**, 287. <http://adsabs.harvard.edu/abs/1966PASJ...18..287H>
- [7] De Moraes, R.V. (1981) Combined Solar Radiation Pressure and Drag Effects on the Orbits of Artificial Satellites. *Celestial Mechanics*, **25**, 281-292. <http://link.springer.com/article/10.1007%2FBF01228965>
- [8] Kampos, B. (1986) Nasa CR1008-Guidance, Flight Mech. and Trajectory Optimization, Vol. IX.
- [9] Sehna, L. (1970) In Dynamics Satellites (Morando, Ed.). Springer, Berlin.
- [10] Sehna, L. (1975) In Satellite Dynamics (Giacalia, Ed.). Springer, Berlin.
- [11] McInnes, C.R. and Brown, J.C. (1990) The Dynamics of Solar Sails with a Non-Point Source of Radiation Pressure *Celestial Mechanics and Dynamical Astronomy*, **49**, 249-264. <http://link.springer.com/article/10.1007%2FBF00049416>
- [12] Feraz-Mello, S. (1972) Analytical Study of the Earth's Shadowing Effects on Satellite Orbits. *Celestial Mechanics*, **5**, 80-101. <http://link.springer.com/article/10.1007%2FBF01227825>
- [13] Anselono, L., Bertotti, B., Farinella, P., Milani, A. and Mobil, A.M. (1983) Orbital Perturbation Due to Radiation Pressure for a Spacecraft of Complex Shape. *Celestial Mechanics*, **29**, 27-43. <http://link.springer.com/article/10.1007%2FBF01358596>
- [14] Geyling, F.T. and Westerman, H.R. (1971) Introduction to Orbital Mech. Addison Wesley.
- [15] Lála, P. (1972) Combined Gravitational and Solar Radiation Pressure Effects on The Semimajor Axis of the Earth's Satellite. *Bulletin of the Astronomical Institute of Czechoslovakia*, **23**, 342. <http://adsabs.harvard.edu/abs/1972BAICz..23..342L>
- [16] McMahan, J.W. (2011) ProQuest Dissertations and Theses. Ph.D. Thesis, University of Colorado at Boulder, Boulder.
- [17] Lücking, C., Colombo, C. and McInnes, C.R. (2012) A Passive Satellite Deorbiting Strategy for Medium Earth Orbit Using Solar Radiation Pressure and the J_2 Effect. *Acta Astronautica*, **77**, 197-206. <http://adsabs.harvard.edu/abs/2012AcAau..77..197L>
<http://dx.doi.org/10.1016/j.actaastro.2012.03.026>
- [18] Cook, D.G. (2001) Master's Thesis, AD-A390187; AFIT/GSP/ENY/01M-01 Graduate School of Engineering and Management.
- [19] Selim, H.H. (1991) The Effect of Solar Radiation Pressure on The Motion of Artificial Satellite. M.Sc. Thesis, Cairo University, Giza.
- [20] Deprit, A. (1969) Canonical Transformations Depending on a Small Parameter. *Celestial Mechanics*, **1**, 12-30. <http://adsabs.harvard.edu/abs/1969CeMec...1...12D>

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