

Numerical Modelling of Coal Combustion Processes in the Vortex Furnace with Dual-Port Loading

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ABSTRACT

The work is devoted to numerical simulation of pulverized-coal combustion processes in the vortex furnace which is a prospective design of a boiler unit for thermal power plants. New modification of this design characterized by additional tangential-injection nozzle located at the bottom of combustion chamber has been studied. Numerical results for the case of Siberian brown coal combustion in this vortex furnace with dual-port loading are presented, including 3-D aerodynamic structure, the fields of temperatures, radiated heat fluxes, species and dispersed phase concentrations, and NO_x emissions.

Keywords: Vortex Furnace, Pulverized-Coal Combustion, Numerical Modelling

1. Introduction

One of prospective technologies to improve the heat transfer efficiency and ecological performance of boiler units for thermal power plants utilizing low-grade coals is the use of furnace devices with vertical jet pulverized-coal combustion. As a kind of these devices, the so-called vortex furnace has been considered. The vortex furnace design shown in **Figure 1** usually comprises the vortex combustion chamber in the lower part of the furnace, the diffuser, and the cooling chamber exited with discharge flue. With this the pulverized coal-air mixture is injected tangentially (alongside the main vortex flow in the combustion chamber) via rectangular nozzle located in the upper part of combustion chamber, i.e. via the upper loading port. For the initial vortex furnace design with only upper nozzle injection, the complex experimental and numerical study of interior aerodynamics and turbulent flow features has been carried out at laboratory-scale vortex furnace model in isothermal formulation earlier by authors [1]. In the presented work a new modification of this design, proposed also by authors [2] and characterized by additional tangential-injection nozzle located horizontally at the bottom of combustion chamber (bottom loading port), has been studied numerically with comprehensive combustion model for the full-scale furnace configuration. In this new vortex furnace design with dual-port loading, the stable swirl flow inside the combustion chamber is formed by two tangential jets of coal-air mixture released from the upper and

the bottom nozzles. The other distinguishing features of new vortex furnace design are: the ratio of the diffuser throat width to effective diameter D_0 of vortex combustion chamber is $H_x = 0.4$; the centers of semi-orifices (comprising the frontal and back walls in XY -section of the vortex chamber) are displaced between each other strictly along vertical line (**Figure 1**).

The following advantages of new vortex furnace design can be indicated: – the compact mass-dimensional parameters, – the distributed (between the upper and the bottom nozzles) tangential input of air or fuel-air mixture, which also provides more flexibility to control the furnace working regime, – the horizontal direction of the axis of rotation which facilitates the completeness of coal

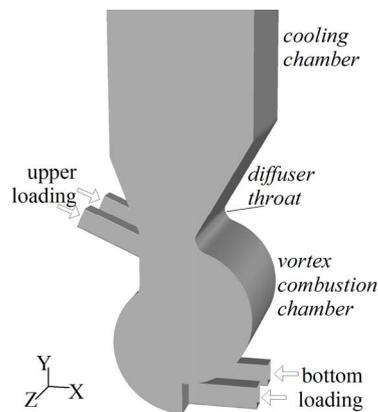


Figure 1. Scheme of the vortex furnace new design (upper part of the cooling chamber is not shown).

burnout process, – the high-temperature combustion of coals of wide quality range in the regime of liquid slag removal in the vortex chamber. Obviously the vortex transfer appears to be the aerodynamic basis of entire combustion process in the studied vortex furnace. With this the appropriate aerodynamic structure can enhance the combustion efficiency, stability of temperature and heat flux distributions, and reduce toxic emissions. Therefore the purpose of the presented work is to investigate the features of two-phase turbulent reacting flow inside the furnace with aim to estimate and justify the new design and operating parameters for prospective boiler units with vortex furnace.

2. Mathematical Model of Aerodynamics and Coal Combustion Processes

Flow in the vortex furnace is subsonic, turbulent and essentially three-dimensional due to the constructional features of the furnace. The mathematical formulation of the steady-state 3-D turbulent two-phase reacting flow is based on Euler-Lagrange approach. The Favre-averaged governing equations of momentum, enthalpy and the species concentration transport for carrier gas phase are closed with “realizable” modification of k - ε turbulence model [3]. This model, compared to the “standard” k - ε model [4], provides more accurate predictions for swirl flows (which is essential for the vortex furnace) and demonstrates good numerical robustness.

For description of the heat-mass transfer and combustion processes, the carrier gas phase is represented by mixture of the following species: $\{O_2, CH_nO_m, CO, CO_2, H_2O, N_2\}$. Here CH_nO_m denotes the generalized formula for hydrocarbons released during the volatiles’ pyrolysis. Coefficients m and n in CH_nO_m formula were calculated from elementary composition of the coal which is presumed to consist of the volatiles, ash, wet and free carbon. Berezhovskij brown coal (rank “2B”) of siberian Kansk-Achinsk field has been taken in simulations, with the following data of its proximal analysis: $W = 33\%$, $A_0 = 4.7\%$, $C_{f0} = 44.2\%$, $O_{V0} = 14.4\%$, $H_{V0} = 3.1\%$, $N_{V0} = 0.4\%$, $S_0 = 0.2\%$. Its volatiles content related to dry ash-free fraction is $V^{daf} = 48\%$, and burnout heat release $Q = 15.66$ MJ/kg. Then the coefficients in CH_nO_m were estimated as: $n = 3.037$, $m = 0.889$, and volatiles’ molar weight = 29.3 g/mol. The rate of the hydrocarbon gas-phase burning reaction in the turbulent flow has been determined according to eddy-dissipation model [5]. The radiation energy transport equation is based on the known P - I approximation of the spherical harmonics method, with this the gas-phase absorption coefficients are calculated according to the weighted sum of gray gases model [6], while the dispersed phase absorption and scattering coefficients are determined on the basis of the optically large particles approximation [7].

Within Lagrangian particle-tracking framework, ODEs of mass, momentum and heat transfer for representative clusters of polydispersed coal particles are solved along each trajectory calculated via random walk model to account for turbulent dispersion of particles. The rate of heterogeneous coke combustion reaction has been modeled by diffusion-kinetic relations [7]. To account for polydispersity, the clusters of fresh coal particles were split into ~ 100 fractions according to Rosin-Rammler size distribution law, with this the coal milling dispersity (specified as mass fraction of remainder on 90 μm sieve) was taken as $R_{90} = 0.15$ and polydispersity degree $\eta = 1.2$.

To predict the nitric oxides emission in the vortex furnace, at the stage of post-processing of main solution data, the transport equations for the concentrations of $\{NO, HCN, NH_3\}$ species have been solved. With this the following NOx formation mechanisms are taken into account: a) extended “thermal” N_2 oxidation mechanism of Ya.B.Zeldovich; b) “prompt” NO formation in reaction of N_2 with hydrocarbon radicals following C. P. Fenimore mechanism; c) “fuel” NO formation mechanism. Also the factors of NO reduction are accounted: – via “reburn-mechanism” in reaction with CH_i radicals; – via heterogeneous consumption of NO on the surface of coke particles.

3. Boundary Conditions and Numerical Algorithm

Because the studied vortex furnace has a symmetry plane S_{XY} (in the section $z_{symm}=3$ m) between two nozzles (see **Figure 1**), the mirror symmetry of the flow is assumed in relation to this plane, thus to save computational resources only half of the furnace volume was considered in simulations. The bounding dimensions of computational domain were taken as: $x_{max}=D_0=6$ m, $y_{max}=36$ m, $z_{max}=z_{symm}=3$ m. The upper nozzle angle of inclination to horizon (XZ plane) was $\beta=24^\circ$. The unstructured computational grid built for finite-volume discrimination of equations consists of 351784 hexahedral cells. The “sticking” boundary conditions are set for velocity components at walls, and “enhanced wall treatment” method [8] is applied for near-wall turbulence modeling. A uniform profile of mean flowrate velocity is prescribed at inlet sections of rectangular nozzles, with this the intensity of inlet turbulent pulsations is assumed equal to 5%.

The known Marsha boundary conditions [7] for the radiation energy transport equation have been set at walls, with emissivity of the furnace walls taken as $\varepsilon_w = 0.8$ and the temperature of superheated water-vapor inside the heat-exchange tubes presumed at 390°C . Also to set boundary conditions for radioactive and convective heat transfer at the heat-exchange surfaces, the integral heat transfer coefficient α_w through the walls (from ash- de-

position layer surface to water-vapor inside the tubes) has been prescribed at different heat-exchange screens (walls) in the furnace cooling chamber. Its values specified at these surfaces, as well as the average values of the screens thermal efficiency Ψ_{av} obtained in computations, are given in **Table 1**.

The mass flow rate of pulverized coal loaded per computed half-section of the vortex furnace volume was $G_{coal}=3.75$ kg/s, the overall air excess coefficient at furnace inlets was set typically for the vortex furnace design as 1.15. With this the primary stream of coal-air mixture at inlet temperature of 180°C was loaded through the upper nozzle only, while the secondary dry air stream at 320°C was supplied through the bottom nozzle. Also the ratio of gas-phase flow rates through the upper and the bottom nozzles $\gamma=G_{upper}/G_{bottom}$ was taken as $\gamma=3$ – as it has been revealed in computations, this parameter influences the aerodynamic structure formed by two tangentially injected jets inside the vortex combustion chamber.

For numerical approximation of convective terms of Favre-averaged momentum equations the second-order upwind scheme [8] has been applied. The numerical solution at each “global” iteration for the gas phase equations is obtained according to PISO algorithm [9] for pressure-velocity decoupling, while the interphone exchange of mass, momentum and heat transfer is accounted following the Particle-Source-In-Cell method [10].

4. Computational Results

Typical flow structure in the studied vortex furnace is represented in **Figure 2** in the form of velocity magnitude is contours in XY -section ($z = 1.6$ m) across the nozzle center (zoomed view of only the vortex chamber and diffuser is shown). The following features of the vortex furnace aerodynamics can be seen, such as the “glove-scheme” flow structure when the upper inlet jet (injected tangentially from the nozzle) evolving alongside the main vortex flow in the combustion chamber is being crossed and surrounded by the upstream flow moving out of this chamber towards the diffuser part of the furnace. Near the center of vortex combustion chamber (but shifted in upper-right direction) a region of vortex core is clearly

Table 1. Average heat transfer coefficients (α_w) and thermal efficiencies (Ψ_{av}) at heat-exchange surfaces.

| Heat-exchange surface location | Specified value of α_w , $\text{W}/(\text{m}^2\cdot\text{K})$ | Computed value of Ψ_{av} |
|--|--|-------------------------------|
| Screen “A” in section $z=3$ m | 250 | 0.590 |
| Two-sided screen “B” in section $z=0$ | 200 | 0.483 |
| Other screens (front, rear and ceiling in the cooling chamber) | 150 | 0.415 |

seen. The upstream flow inside the cooling chamber demonstrates noticeable spatial non-uniformity (such as Coanda effect).

This Coanda effect can be also seen from **Figure 3** where the temperature field in the new vortex furnace is demonstrated. High-temperature level inside the vortex combustion chamber (reaching 1880°C in the near-wall reaction zone, see **Figure 3**) provides the stable regime of liquid slag removal. This is also confirmed from **Figure 4** where the profile of temperature averaged over horizontal XZ -sections is depicted along furnace height (solid line in **Figure 4**). Also the dashed line there indicates the profile of maximum temperature values along furnace height. With this the estimated heat release rate per unit volume inside the combustion chamber is ~ 800 kW/m^3 .

The sharp downward trough in averaged temperature profile (at ~ 7 m height) in **Figure 4** indicates the cold jet stream evolved from the upper nozzle (see also **Figure 3**). Then in the cooling chamber the temperature decreases rapidly along the furnace height, and the temperature

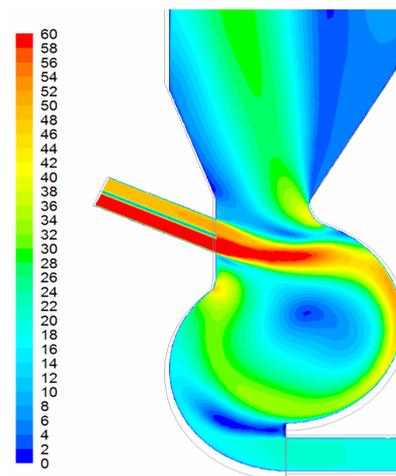


Figure 2. Contours of velocity magnitude in the new vortex furnace (section across nozzle center $z=1.6$ m), m/s.

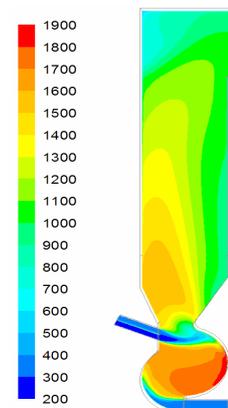


Figure 3. Temperature field in the new vortex furnace (section across nozzle center $z = 1.6$ m), $^\circ\text{C}$.

field becomes more uniform as the difference between maximum and averaged values becomes smaller. This improved performance appears due to efficient heat transfer from the furnace volume to heat-exchange screens. Indeed the profiles of the screens thermal efficiency Ψ (defined as $\Psi=Q_{res}/Q_{inc}$ where Q_{res} is the resulting heat flux at wall surface and Q_{inc} – the incident heat flux) shown in **Figure 5** demonstrate rather good thermal efficiency values of screens in the cooling chamber. The dashed line there denotes values at the middle of screen “A” (mounted in section $z=3$ m) and the solid line – at the middle of two-sided screen “B” (in section $z=0$). Values Ψ_{av} averaged over each screen surface area (shown in **Table 1**) also indicate the range $0.41 < \Psi_{av} < 0.59$ which is better than a typical level of ~ 0.4 for most furnaces.

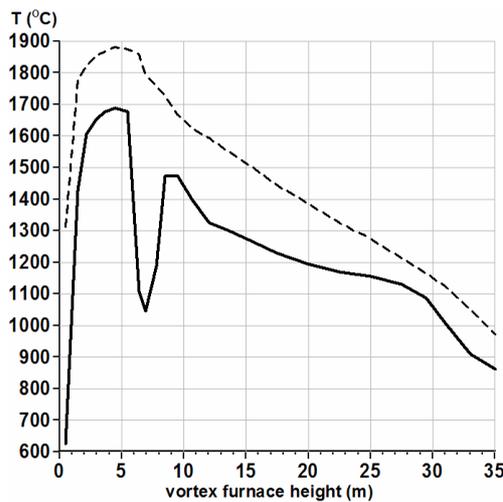


Figure 4. Averaged (solid line) and maximum (dashed line) temperature distribution along vortex furnace height, °C.

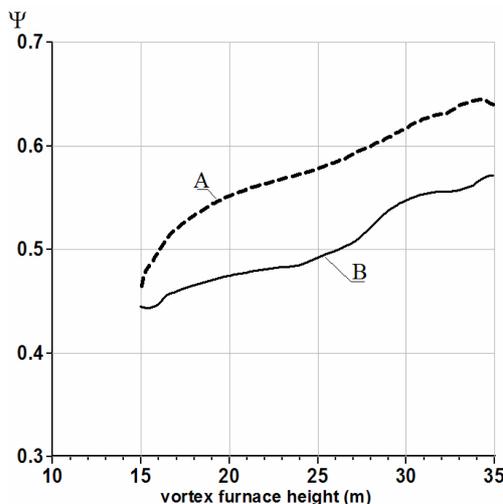


Figure 5. Heat-exchange screens thermal efficiency Ψ distribution along furnace height in the cooling chamber; dashed line – screen “A”, solid line – screen “B”.

The contours of resulting heat fluxes (at heat-exchange screens) are shown in **Figure 6**. The profiles of O_2 , CO , NO concentrations averaged over horizontal XZ -sections are plotted in **Figure 7** along furnace height.

The following integral parameters have been obtained in the vortex furnace outflow section (before exhaust flue): mean temperature $T_{mean} = 980^\circ C$, maximum temperature $T_{max} = 1055^\circ C$, coke burnout incompleteness coefficient $q_4 \sim 1.3\%$, averaged volumetric concentrations $[O_2] = 2.47\%$, $[CO] = 14$ ppm, $[NO] = 307$ ppm. It is seen that NO_x emission level remains contained within acceptable limits – notwithstanding the high temperature level inside the vortex combustion chamber – mainly due to the effect of recirculation of combustion products

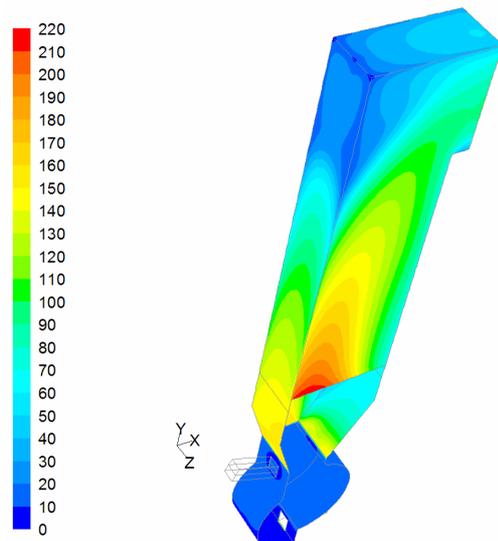


Figure 6. Heat fluxes absorbed at the surface of heat-exchange screens, kW/m^2 .

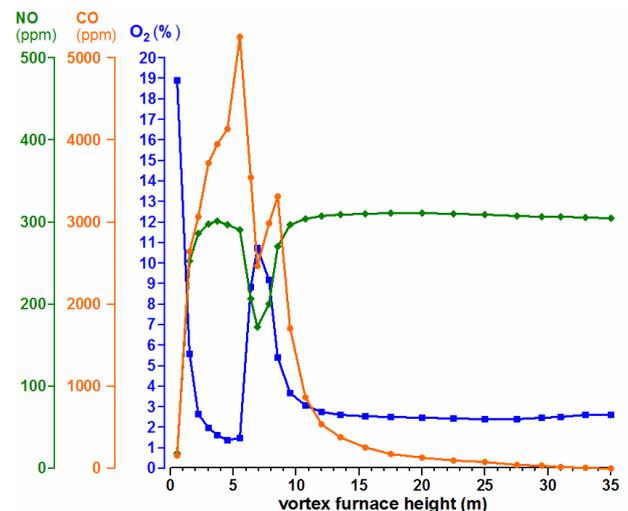


Figure 7. Distributions of O_2 , CO , NO concentrations (by volume), averaged over horizontal XZ -sections, along vortex furnace height.

towards reaction zones inside the vortex chamber. Also the improved thermal efficiency of heat-exchange screens in the cooling chamber has been demonstrated.

5. Conclusions

The numerical simulation of 3-D aerodynamics and pulverized brown coal combustion processes in the new vortex furnace configuration with dual-port loading has been performed, and detailed flow field information including the flow structure, the fields of temperatures, radiated heat fluxes, species and dispersed phase concentrations have been obtained. With this it has been shown that the integral heat engineering and ecological parameters of new vortex furnace meet the ranges corresponding to engineering practice.

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REFERENCES

- [1] V. V. Salomatov, D. V. Krasinsky, Yu. A. Anikin, I. S. Anufriev, O. V. Sharypov and Kh. Enkhjargal, "Experimental and Numerical Investigation of Aerodynamic Characteristics of Swirling Flows in a Model of the Swirling-type Furnace of a Steam Generator," *J. of Engineering Physics and Thermophysics*, Vol. 85, No. 2, 2012, pp. 282-293. [doi:10.1007/s10891-012-0651-8](https://doi.org/10.1007/s10891-012-0651-8)
- [2] N. V. Golovanov, V. E. Nakoryakov, A. P. Burdukov, V. V. Salomatov and A. A. Dorozhkov, "Vortex Furnace," *Russian Federation patent No.2042084*, 1995 (in Russian).
- [3] T.-H. Shih, W. W. Liou, A. Shabir, Z. Yang and J. Zhu "A New k- ϵ Eddy-Viscosity Model for High Reynolds Number Turbulent Flows – Model Development and Validation," *Computers&Fluids* –Vol. 24, No. 3, 1995, pp. 227-238. [doi:10.1016/0045-7930\(94\)00032-T](https://doi.org/10.1016/0045-7930(94)00032-T)
- [4] B. E. Launder and D. B. Spalding, "Lectures in Mathematical Models of Turbulence," London (England): Academic Press, 1972.
- [5] B. F. Magnussen, "The Eddy Dissipation Concept," IEA, 1989.
- [6] T. F. Smith, Z. F. Shen and J. N. Friedman "Evaluation of Coefficients for the Weighted Sum of Gray Gases Model," *Proceedings of XX-th National ASME-AIChE Heat Transfer Conference*, Milwaukee, USA, August 2-5, 1981.
- [7] E. P. Volkov, L. I. Zaichik and V. A. Pershukov, "Modelling the Solid Fuel Combustion," Moscow: Nauka, 1994 (in Russian).
- [8] FLUENT 6.3 User's Guide, Fluent Inc., 2006.
- [9] R. I. Issa, "Solution of Implicitly Discretized Fluid Flow Equations by Operator Splitting," *Journal of Computational Physics*, Vol. 62, No. 1, 1986, pp. 40-65. [doi:10.1016/0021-9991\(86\)90099-9](https://doi.org/10.1016/0021-9991(86)90099-9)
- [10] C. T. Crowe, M. P. Sharma and D. E. Stock, "The Particle-Source-In-Cell (PSI-CELL) model for gas-droplet flows," *ASME Journal of Fluids Engineering*, Vol. 99, No. 2, 1977, pp. 325-332. [doi:10.1115/1.3448756](https://doi.org/10.1115/1.3448756)