



# Microstructure, Mechanical Properties, Toughness, Wear Characteristics and Fracture Phenomena of Austenitised and Austempered Low-Alloyed Ductile Iron

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## Abstract

Conventional ductile iron (DI) and austempered ductile iron (ADI) alloys were successfully produced. The alloying elements—Ni, Mo, Cr and Mn were added to get as-cast low-alloyed ductile iron (LADI) followed by austempering heat treatment. Hardness was measured for all investigated alloys. A hardness conversion table was deduced for irons investigated. Highest values of calculated quality index (QI) were for ADI alloys implying higher material performance. Impact energy values were attributed to microstructure and tensile properties. Wear characteristics of selected ADI alloys showed comparable values with LADI. Micro-hardness values were used to identify the micro-constituents and the work-hardened layers. Fracture modes were revealed and fracture surface observations were done by SEM. Due to higher toughness and QI, and cheap price, it is suggested that rolling mills may be produced from ADI, specially heat-treated instead of LADI.

## Keywords

Austempered Ductile Iron, Low-Alloyed Ductile Iron, Tensile Properties, Wear Characteristics, Microstructure, Hardness

**Subject Areas:** Composite Material, Fundamentals of Material Science

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## 1. Introduction

Ductile irons, having a spheroidal graphite structure, have been evolving during the last few decades [1], result-

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ing in better combinations of mechanical strength and toughness. A further and recent advance [2] has been the introduction of austempering thermal treatment. The term of austempered ductile iron (ADI) describes a range of irons which have been subjected to an isothermal (specific austempering) heat treatment to produce an essentially bainitic structure in the material [3]. ADI has been one of the most significant developments in cast iron technology since the discovery of ductile iron [4]. Austempering has created a new family of irons with moderate level of toughness and ductility [4]. ADI is currently used widely in the place of cast, forged or low-alloyed steel, and weld components in a number of applications [1]. The commercial use of ADI is increasing continuously nowadays because their mechanical properties can be optimised and thereby enable the grades of standard ASTM A 897-M: 1990 (metric) to be satisfied [5]. However widespread use of this material cannot take place before the effect of composition, as-cast structure and heat-treated alloys are firmly established. Conventional ADI is two types—being alloyed and heat-treated DI [6]. The remarkable combination of strength and ductility has been attributed to differences in the type and amount of micro-constituents arising from variations of composition and austempering processing parameters [7]. The effect of some alloying elements on the microstructure and properties of austenitic ductile iron have been previously clarified by Fatahalla *et al.* [8]. The alloying elements such as nickel (Ni), molybdenum (Mo), chromium (Cr) etc. affect the mechanical properties of ADI [9].

The aim of the present investigation was to imply the examination of the influence of microstructure on the tensile properties, wear, fracture characteristics, and impact and fracture toughness for LADI and ADI. Another aim was to detect the possibility of using a specially heat-treated ADI to replace the low-alloyed steel or low-alloyed ductile.

## 2. Experimental Procedures

### 2.1. Materials and Melting

The materials investigated were produced in a normal production process of El Nasr Castings Co. in Egypt; Fatahalla *et al.* [10]. Pig iron, commercially pure iron and other alloys were melted in a basic high frequency induction furnace with a melt capacity of 6 ton/hr. The normal charge material, which produced conventional DI, was subjected to two austempering treatments, which produced ADI. Additions of the following alloys: FeNi (95% Ni), FeMo (70% Mo), FeCr (70% Cr), and FeMn (70% Mn) were done to obtain LADI in the as-cast condition while the main addition alloy was the Nickel which was controlled at a level of 2 mass %. The melt was then superheated and held at 1773 K for 0.6 ks. It was then tapped into a preheated ladle at 1733 K with a pocket in the base (Sandwich Technique) [10]. It was then inoculated (16% by weight of charge) using Fe-Si-Mg alloy (45, 50 and 5 mass % respectively) having a grain size of 15 - 50 mm. thereafter, inoculated (0.6% by weight of charge) using Fe-Si alloy (80% Si) having a grain size of 0.2 - 3 mm. The inoculant was added to the stream of molten iron as the ladle was being filled for a period of 0.3 ks. The temperature of molten iron before treatment was measured to be 1693 K and after treatment was 1663 K for a period of 0.24 ks. An immersion thermocouple, digital meter stick was used to monitor the melt temperature in the induction furnace and pouring ladle. Spectrographic samples were cast in a copper mould for analysis by emission spectrograph. **Table 1** shows the chemical compositions of the conventional DI and LADI.

### 2.2. Casting Procedure

After spheroidisation treatment, the melt was cast into a Y-block, using green sand moulds. All ingots of conventional DI and LADI, were cast into the same dimensions to maintain the same solidification cooling rate (SCR) and also the casting conditions. The conventional DI ingots were poured from a single preheated ladle at a pouring temperature in the range 1623 - 1643 K for 0.12 ks. The LADI was cast solely under almost the same conditions. Upon completion of pouring, all castings were allowed to cool overnight in the moulds before shake-

**Table 1.** Chemical composition of the conventional ductile-iron and LADI [mass %].

Materials	Elements										
	C	Si	S	P	Mn	Ni	Cr	Mo	Mg	CE	
Conventional ductile-iron	3.57	2.20	0.013	0.036	0.086	0.02	0.02	-	0.040	4.315	
LADI	3.50	1.54	0.009	0.045	0.71	2.22	0.75	0.45	0.042	4.100	

out and sandblasting.

### 2.3. Sampling

It is recommended [1] that most of machining be carried out before heat treatment, depending on the ADI components which will have either a high hardness or high work-hardening rate. The conventional DI Y-block samples were machined into specimens with 0.5 mm finishing allowance. The specimens were cut from the bottom of Y-blocks while the top sections were not used in order to avoid variations in nodule characteristics and porosity. After heat treating, the specimens were machined to final dimensions. The LADI castings got their dimensions by sectioning and grinding in cold working operations.

### 2.4. Heat Treatments

#### 2.4.1. Austempering Treatment

The austempering treatment was performed in a specialised commercial heat treatment shop in the Iron and Steel Company in Egypt. The machined conventional DI and LADI specimens were placed in baskets, which were loaded one at a time into a high-temperature furnace for austenitisation. Following that, the baskets were transferred rapidly into a quenching medium (salt bath) held at the required austempering temperatures. Previous publication [6] indicated that the range for austenitising is; 1123 - 1223 K and for austempering to be; 500 - 723 K. Special heat-treatments performed, in the present investigation. Austenitising and austempering temperatures, used in the present investigation, were carefully selected to be 1143 K and 1203 K for the first and 573 K and 673 K for the latter to fill the gap in literature. The ADI specimens were austenitised either at 1203 K or 1143 K for 3.6 ks or 5.4 ks, respectively. While the LADI specimens were austenitised at 1143 K for 5.4 ks, in a muffle furnace with accuracy  $\pm 5$  K, thereafter, transferred directly, for quenching into a salt bath containing Die Gusse As 140 at one of the austempering temperatures 673 K and 573 K for 3.6 ks and 7.2 ks, respectively. After austempering, all specimens were air-cooled to 300 K.

#### 2.4.2. Tempering

Tempering treatment was performed on the as-cast LADI ingots at 873 K for 7.2 ks to obtain the best combination of strength, hardness and machinability.

### 2.5. Metallographic Characterisation

The microstructural evaluations (including nodules and matrix characteristics) were conducted on small cubes, 20 mm side, taken from each Y-block of the as-cast and as-heat treated castings. Standard experimental techniques for metallography were used [11]. 3% nital was used as an etchant for 8 s to reveal the structural characteristics of the investigated alloys. Observation of the microstructures was carried out using Reichert Me F2 universal optical microscope.

### 2.6. Hardness Tests

Micro- and macro-hardness tests were conducted. The latter type was conducted using three indentation hardness tests, 1) Vickers, 2) Brinell, and 3) Rockwell. All the hardness tests were carried out at ambient temperature (300 K) using Otto Wolpert-Weker universal hardness testing machine. Standard microhardness Vickers test was applied on the specimens of each type to obtain the hardness of its matrix constituents. It was also carried out on longitudinal section perpendicular to the worn surface at different distances. Wolpert high quality digital microhardness tester set at 200 g load for 15 s was used. Preparation of samples for microhardness tests was the same as those for microstructure.

### 2.7. Tensile Test

Tensile tests were performed in accordance with ASTM-E8-1990 and the round tensile specimens were used. The tensile tests were conducted on a motor driven tensiometer machine type W at ambient temperature (300 K) at a strain rate of  $4 \times 10^{-4} \text{ s}^{-1}$  up to fracture. Data collected included ultimate tensile strength ( $R_m$ ), 0.2% proof stress ( $R_{p0.2}$ ), percentage elongation ( $\% A_4$ ) and toughness. Each value was the average of at least three successful tensile tests.

## 2.8. Impact Test

For Charpy impact tests, smooth (unnotched) specimens were used in accordance with ASTM-E23-1990. The capacities of Charpy impact testing machine type RKP300 were 300 J. Impact tests were performed at 300 K and all impact test results were averaged from, at least, three successful tests.

## 2.9. Wear Test

Wear-resistance tests were carried out with a pin-on-ring type apparatus using the TNO-Tribometer under dry sliding conditions in the ambient air. ADI specimens in the form of cylindrical pin of 8 mm diameter and 12 mm height were used. The pin was pressed against the circumference of the ring under a constant load of 265 N for 2.7 ks duration at a constant sliding distance of 2500 m. Mass losses were measured in grams before- and after wear tests with a precision to the fourth decimal by an analytical balance (Mettler). At least three successful tests were performed for each set of conditions and the average mass loss was calculated. After testing, the worn surfaces of pins were examined by optical microscope. Friction coefficient ( $\mu$ ) =  $F_f/F_n$  as  $F_n$  = normal force, and  $F_f$  = friction force was calculated.

## 2.10. Fracture Characteristics

Analysis of fracture surfaces illuminates the role of various microconstituents, and under different heat treatment conditions, in promoting or limiting tensile (static) and impact (dynamic) toughness. Caution was taken for appropriate handling of specimens after fracture, former to any fractographic examination. They were instantly examined, in the SEM, to avoid any accumulation of under humidity on the fracture surfaces. OM Axiovert 405 M-ZEISS was used to explain the fracture behaviour of the two types; tensile and impact fracture.

## 3. Results and Discussion

All kinds of tested-alloys together with their conditions are listed in [Table 2](#).

### 3.1. Microstructural Features

**Nodule characteristics:** As polished specimen-photographs were examined for nodule-characteristics. The differences among the nodule characteristics (count, size and nodularity) were, therefore, detected. [Figure 1\(a\)](#), [Figure 1\(b\)](#) illustrates, the as-polished microstructure for alloy 1 (conventional DI) and 6 (as-cast LADI), respectively. Higher nodule count (87 nodule/mm<sup>2</sup>) and smaller nodule size (50  $\mu$ m) was observed in [Figure 1\(a\)](#) for alloy 1 (DI) relative to that in [Figure 1\(b\)](#) (69 nodule/mm<sup>2</sup> and nodule size 62  $\mu$ m) for alloy 6 (LADI). Comparing the two types of alloys 1 and 6 with ADI and as-treated (LADI) specimens, respectively, it was observed that, there is no remarkable change in the graphite nodule characteristics due to austempering heat treatment. Therefore, alloys 1 and 6 were taken to represent the other types of ADI and LADI specimens, respectively. The value of % nodularity of alloy 1; 95.5% and alloy 6; 93.5% are close to each other. These results can be attributed to the same SCR. These variations of nodule count may stem from the variation in chemical compositions between the two types of ductile-irons [9].

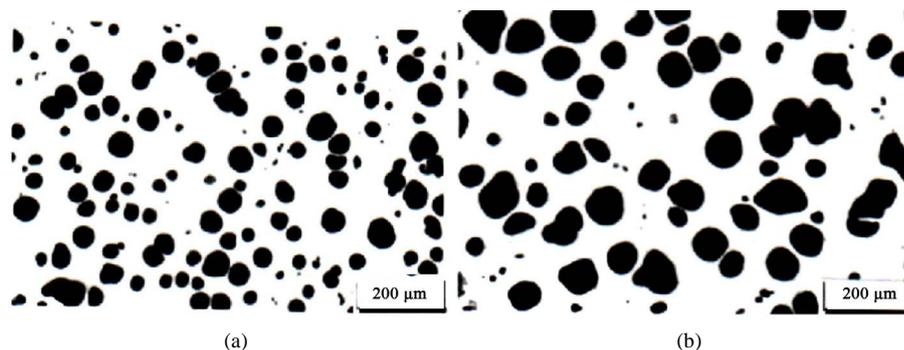
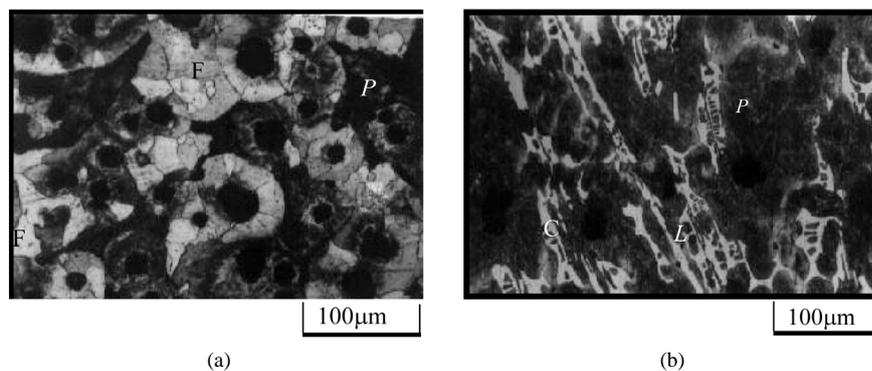
Also, Jenkins [12] stated that, nodule count affects graphite size and shape. Increasing nodule count results in a decrease in nodule size, which improves tensile and fracture properties. The present result of % nodularity satisfies the ASM specification [13], which indicates that for most purposes, a % nodularity of 85% - 100% should be in a fully nodular form. Additionally, these results are confirmed with the results obtained by Grech [14]. He concluded that, the morphology of graphite influences the final properties of an ADI. Whereas, for optimum mechanical properties, the % nodularity of the graphite in castings with a low nodule count should be as high as possible.

### 3.2. Microstructural Features

[Figure 2\(a\)](#) and [Figure 2\(b\)](#) illustrates the matrix micro-constituents after etching the specimens for as-cast-alloys 1 and 6, respectively. The microstructure in [Figure 2\(a\)](#) is a typical bull's eye type [11] with ferrite surrounding the graphite nodules. The microstructure delineates, generally, graphite nodules embedded in a ferritic-pearlitic matrix (48% pearlite). [Figure 2\(b\)](#) represents the microstructure of alloy 6 with Ni content 2.22 mass %.

**Table 2.** Designations used, in the present investigation, for the 3 types (DI, ADI, LADI) of ductile-iron.

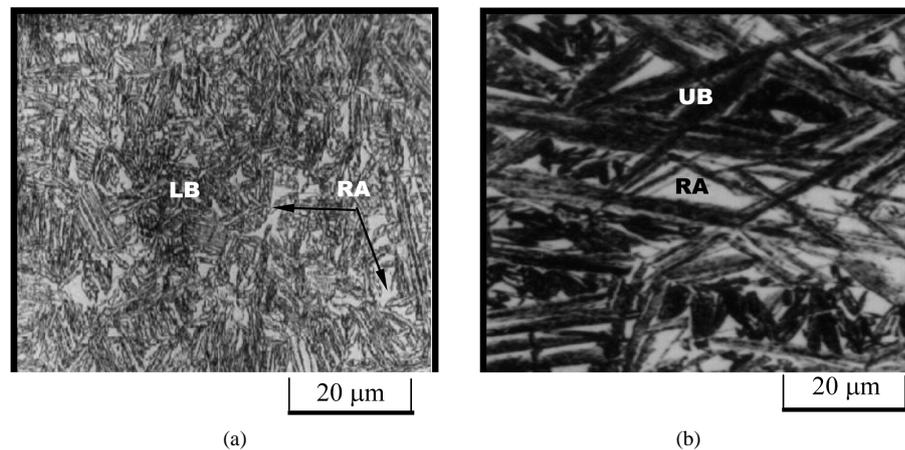
DI	ADI				LADI		
	Austenitising conditions				Austenitising conditions		
	1143 K (5.4 ks)		1203 K (3.6 ks)		1143 K (5.4 ks)		
Conventional ductile-iron	Austempering conditions				As-cast	Austempering conditions	
	573 K (7.2 ks)	673 K (3.6 ks)	573 K (7.2 ks)	673 K (3.6 ks)		573 K (7.2 ks)	673 K (3.6 ks)
Alloy 1	Alloy 2	Alloy 3	Alloy 4	Alloy 5	Alloy 6	Alloy 7	Alloy 8

**Figure 1.** Graphite nodules as observed for as-polished specimens (a) alloy 1 (conventional DI) and (b) alloy 6 (LADI). Optical microscope.**Figure 2.** The observed matrix constituents for (a) alloy 1 (conventional DI), and (b) alloy 6 (as-cast LADI); after etching with nital 2%. Optical microscope.

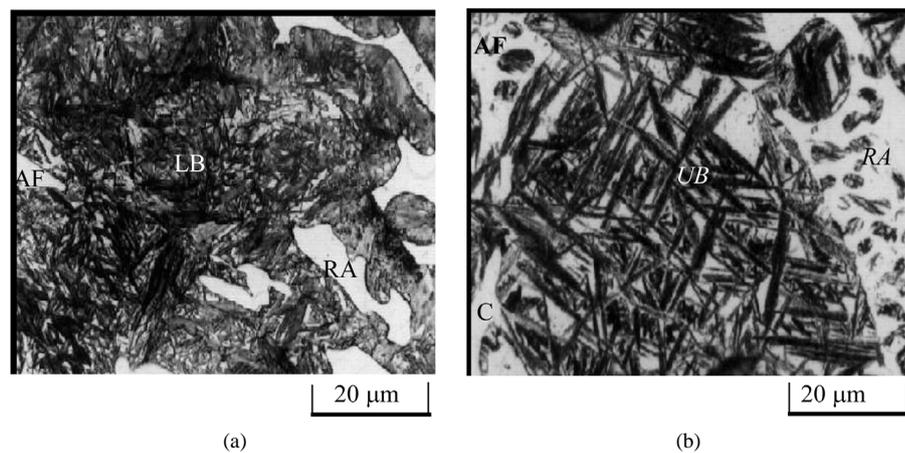
The matrix micro-constituents are mainly pearlite (P), ledeburite (L), and carbide (C) (cf. photo) occurring in the inter-nodular regions. Graphite nodules in a matrix of higher volume fraction of ferrite (52%) provide a ductile-iron with good ductility, impact resistance and tensile strength, as will be discussed later [15]. The microstructure of alloy 1 resembles those obtained previously by Fatahalla [15] and Jenkins [12]. The latter [12] indicated that, increasing nodule count decreases the pearlite content, decreasing strength and increasing elongation. The presence of carbide in the structure of alloy 6 (cf. **Figure 2(b)**) confirms the results of K. M. Ibrahim *et al.* [16] who suggested that the low Ni-alloy additions lead to the formation of intercellular carbides. The previous investigators [16] indicated that repetition of production of bainitic ductile-iron in the as-cast condition based on Ni and Mo could be achieved. While, for the present LADI, Mn and Cr were added in the as-cast condition. Moreover, there was slight difference in volume fraction ( $V_p$ ) of pearlite, which has a higher value than that of ledeburite and carbide in the matrix of alloy 6.

This maybe imputed to the presence of elements that favour pearlite and carbides in the composition, such as Mn and Cr. **Figure 3** reveal the matrix constituents of the ADI after etching for alloys 2, 3 (**Figure 3(a)**, **Figure 3(b)**) and **Figure 4** for alloys 5, 6. Using micro-hardness testing, and comparing the micro-constituents, ob-





**Figure 4.** The observed matrix constituents for the ADI specimens after etching with nital 2%. (a) alloy 4 and (b) alloy 5. Austenitised at 1203K and austempered at (a) 573 K, and (b) 673 K.



**Figure 5.** Matrix constituents of the LADI specimens after etching with nital (a) alloy 7 and (b) alloy 8. All specimens austenitised at 1143 K and austempered at (a) 573 K, and (b) 673 K.

**Table 3.** Hardness conversion table for the 3 types (DI, ADI, and LADI) of ductile-iron.

Iron type Hardness Property	DI		ADI			LADI		
	Alloy 1	Alloy 2	Alloy 3	Alloy 4	Alloy 5	Alloy 6	Alloy 7	Alloy 8
HV (MPa)	1850	4370	2910	4500	3100	4770	6130	4550
HB (MPa)	1780	4160	2870	4310	2920	4460	5600	4350
HR <sub>A</sub>	51.3	-	-	-	-	-	-	-
HR <sub>C</sub>	-	43.0	26.4	44.6	27.3	45.6	56.0	45.0

It is believed that the hardness of the UB observed in **Figure 4(b)** and **Figure 5(b)** together with the more amount of RA resulted in drastic decrease in hardness compared to **Figure 4(a)** and **Figure 5(a)**. The fine structure of LB in **Figure 4(a)** and **Figure 5(a)** is believed to raise the hardness. The slight decrease in hardness for the LADI due to raising the austempering temperature refers to the increase in RA at 673 K relative to its amount at 573 K (cf. **Figure 5(a)**, **Figure 5(b)**). The results of hardness, in the present investigation, confirms those obtained by Korichi Priestner [19] and Abouelala [20] with slight differences. It is to be taken into consideration that the slight differences in chemical composition, austenitising and austempering temperatures between the present and previous investigations are the cause of slight differences in results. Results in **Table 3** for LADI

specimens are supported by those obtained previously [21]. He indicated that the macrohardness values decrease with increasing the austempering temperatures. The lower hardness values are obtained at the austempering temperature of 673 K. The higher hardness of the as-cast LADI given in **Table 3** (4770 HV) refers to the microstructure of **Figure 2(b)** revealing hard pearlite and carbide constituents. This extremely high hardness value of as-cast LADI confirms the results obtained previously by Velez *et al.* [22]. The present results of hardness values for alloys 7 and 8 (LADI) show consistency with the results of Volkov *et al.* [23]. They suggested that by increasing austempering temperatures from 572 K to 663 K, the Brinell hardness values decreased by about 16% to 20% on the average (This decrease per cent is 22% in the present investigation as detected from **Table 3**).

**Microhardness:** **Table 4** lists the microhardness values of the different matrix micro-constituents for the 3 types of alloys under investigation. The hardness of the ferrite, and pearlite phases of alloy 1 confirmed the previous results [24] which showed values of 1700, 2900 MPa for the ferrite and pearlite respectively. J. Aranzabal *et al.* [25] indicated that, the relative microhardness values range of common micro-constituents in cast iron are: Ferrite 1000 - 2500 MPa, austenite 1000 - 4000 MPa, acicular ferrite 3000 - 5000 MPa, austempered (Bainite) 2500 - 6000 MPa and the carbide ranges from 10,000 to 20,000 MPa. The values of microhardness obtained in **Table 4** for the different constituents fall within the indicated ranges of [25].

### 3.4. Tensile Properties

**Table 5** shows the average values of ultimate tensile strength ( $R_m$ ), 0.2% proof stress ( $R_{p0.2}$ ), maximum plastic strain ( $A_4$  %) and toughness, which is calculated as the area under the tensile stress-strain diagram in MPa. **Table 5** delineates that alloy 1 has the lowest  $R_m$  and  $R_{p0.2}$ ; whereas, it exhibits highest  $A_4$ % and toughness “E” compared to ADI and LADI alloys. This high toughness of alloy 1 may stem from the microplasticity associated with ferritic-pearlitic matrix and good % nodularity [28]. One of the interesting results in the present investigation can be noticed from **Table 5** being; comparable  $R_m$ , and  $R_p$  of ADI (alloys 3, 5) to values of LADI (alloy 8). Moreover, alloys 3, 5 (ADI) showed superior ductility and toughness compared to those of LADI (alloy 8). Adding to the previous advantages the high hardness of alloys 3, 5 throws the light on possibility of using this relatively cheap ADI replacing the more expensive LADI in industrial applications such as producing rolling mills and others. Furthermore, the present tensile results confirm those obtained previously [11]-[13].

To evaluate the performance of the present alloys, the quality index (QI), which have been introduced by [12] implying both strength and elongation, is used:

$$(\text{tensile strength ksi})^2 \times (\text{elongation \%}) / 1000 = \text{QI}$$

where QI is a constant, and defined as Quality Index.

The calculated values of QI for the 3 types of ductile-iron are listed in **Table 5**. The highest values of QI is for alloy 3 (ADI); 119.9 and alloy 5 (ADI); 114.5 which, indicates a combination of higher strength and elongation and therefore, higher material performance. The segregation is known to reduce the toughness and ductility [6]. **Table 5** which showed low ductility and toughness for the LADI suggests that the addition of alloying elements like Ni and Mo is not needed as far as the toughness is concerned.

### 3.5. Impact Properties

**Table 6** shows the impact fracture toughness (energy absorbed in J) of the 3 types investigated of ductile-iron. The highest value is observed for the conventional DI (alloy 1) and the lowest values refer to LADI (alloys 6, 7, and 8). For the sake of comparison between the energy absorbed until fracture in a static test (tensile) and a dynamic test (impact) the values of toughness (area under stress strain diagram in MPa) was multiplied by the impact-test-volume to get the toughness in Joule units. The values listed in **Table 6** indicate that the energy required to fracture the tensile specimen (static test) is higher than that obtained by the impact specimen (dynamic test) for the 3 types of ductile-iron. This decrease in energy through impact testing reflects the high rate of loading in impact test compared to tensile test. It is also interesting to point out that although the impact toughness of LADI (alloys 7, 8) was relatively very low, however, its toughness showed values 10 times or more the impact toughness (cf. **Table 6**).

This phenomenon reflects the sensitivity of these LADI alloys to strain rate testing [30]. The present results of energy absorbed for ADI specimens can be explained by that obtained by Ratto *et al.* [31]. They indicated that, in the low range of austempering temperatures ductile-iron develops moderate toughness and low impact

**Table 4.** Microhardness values HV<sub>200</sub> (MPa) for matrices of the constituents of the 3 types of ductile-iron.

Iron type Matrix Constituents	DI		ADI			LADI		
	Alloy 1	Alloy 2	Alloy 3	Alloy 4	Alloy 5	Alloy 6	Alloy 7	Alloy 8
Ferrite (F)	1715	-	-	-	-	-	-	-
Pearlite (P)	2775	-	-	-	-	3800	4000	3610
Retained austenite (RA)	--	3560	3400	3670	3480	--	3890	3765
Acicular ferrite (AF)	--	-	-	-	-	3960	4430	3690
Upper bainite (UB)	--	-	3775	-	3880	--	--	3950
Lower bainite (LB)	--	4795	-	4920	-	--	5380	--
Ledeburite (L)	--	-	-	-	-	6440	6710	--
Carbide (C)	--	-	-	-	-	8030	8820	7660

Hardness units are MPa.

**Table 5.** Tensile properties and Quality Index of the 3 types of ductile-iron.

Iron type Tensile properties and Quality Index (QI)	DI		ADI			LADI		
	Alloy 1	Alloy 2	Alloy 3	Alloy 4	Alloy 5	Alloy 6	Alloy 7	Alloy 8
R <sub>m</sub> (MPa)	518	961	844	980	890	998	1370	979
R <sub>p0.2</sub> (MPa)	380	653	549	679	587	749	1092	666
A <sub>4</sub> (%)	14.2	3.4	7.8	3.0	7.1	2.3	1.1	2.9
Toughness* (MPa)	57.4	20.6	48.9	17.4	47.2	14.1	9.5	16.7
Quality Index(QI)	77.8	63.8	119.9	58.8	114.5	47.0	42.3	56.8

\*Toughness is the area under the curve of the tensile stress-strain diagram (dimensions of the tensile test specimen: gauge diameter = 4 mm, and gauge length = 20 mm).

**Table 6.** Comparison between the values of impact energy and toughness (in energy units) along with the ratios between them for the 3 types of ductile-iron.

Iron type property	DI		ADI			LADI		
	Alloy 1	Alloy 2	Alloy 3	Alloy 4	Alloy 5	Alloy 6	Alloy 7	Alloy 8
E <sub>impact</sub> * (J)	130.0	55.7	88.0	49.5	80.4	4.7	3.5	5.5
E** (J)	160.0	70.4	127.6	60.5	114.4	48.4	33.0	55.0

\*E<sub>impact</sub> = Actual energy measured in impact test in J. \*\*E = Area under tensile stress-strain diagram (MPa) × volume of impact specimen (0.010 × 0.010 × 0.055 cubic metre) (J).

strength. It is believed that, the low value of energy absorbed for alloy 6 (as cast LADI-cf. **Table 6**) referred to the effect of alloy additions; Ni, Mo, Mn, and Cr and its segregation, and the presence of carbides. Kobayashi [27] concluded that, the increase in Mn content resulted in dropping of impact value, due to increasing of the V<sub>f</sub> of pearlite and segregation of Mn. Moreover, the carbides increase rapidly with increasing Mo addition, *i.e.*; the carbides and the segregation of Mo to cell boundaries cause significant loss of impact energy. Marrow and Çetinel [32] indicated that, the decline in impact toughness maybe due to the segregation of carbides around the ferrite/austenite grain boundaries as the austempering temperature exceeds 623 K. Authors of [33] noted that, the impact energy values are more sensitive to these structural changes than, for example, % elongation.

### 3.6. Wear

**Wear characteristics:** **Table 7** lists the values of mass loss and coefficient of friction ( $\mu$ ) of the 3 types of iron;

**Table 7.** Wear characteristics and hardness of the 3 types of ductile-iron.

Iron type Wear Char. and macrohardness	DI		ADI				LADI	
	Alloy 1	Alloy 2	Alloy 3	Alloy 4	Alloy 5	Alloy 6	Alloy 7	Alloy 8
Mass loss (mg)	1491	168	614	151	579	63	22	97
Coefficient of friction ( $\mu$ )	0.48	0.28	0.31	0.27	0.30	0.24	0.21	0.26
HB (MPa)	1780	4160	2870	4310	2920	4460	5600	4350

DI, ADI, and LADI under investigation. Conventional DI (alloy 1) showed the highest value of mass loss and hence, the lowest wear resistance material. This may refer to the relatively high volume fraction of ferrite being 52% (cf. **Figure 2(a)**) and in consistence with the suggestion given before by Ping *et al.* [33]. In general, reviewing the results in **Table 7** we can observe inverse proportionality between the values of hardness and those for mass loss. This confirms that wear resistance is a hardness dependent property in consistence with results and conclusion of [34].

It can also be seen (cf. **Table 7**) that although the LADI (alloys 7, 8) showed the highest wear resistance, however, moderate wear resistance are observed for the ADI (alloys 2, 4). Therefore, LADI showed about twice higher values of wear resistance compared to ADI (alloys 2, 4), however, it is a compromise between required properties and alloy price. M. C. Jeng [34] stated that, the wear resistance absolutely depends on the hardness of the matrix structure, thus confirming the present results in **Table 7**. The coefficient of friction ( $\mu$ ) of alloy 1 is high because of the tendency to adhesion of soft ferrite matrix. The lower value of  $\mu$  for ADI and LADI specimens maybe attributed to the difficulty of adhesion of the bainite matrix in ADI and hard carbide matrix in LADI. This phenomenon of decreasing “ $\mu$ ” is supported by Hemanth [35], who reported that, it should be possible to reduce ( $\mu$ ) by increasing the flow stress of the base metal by alloying or heat treatment.

**Work-hardened layers due to wear:** The work-hardened layers in the different alloys of the present investigation are seen in **Figure 6**. It is noticed that the higher the microhardness value the less thickness of the work-hardened layer is observed. For instance, for the conventional DI-alloy 1-(lowest hardness) the thickness of the work-hardened layer is about 260  $\mu\text{m}$  while for the LADI-alloy 7-(highest hardness) the thickness is about 100  $\mu\text{m}$ . Other investigated alloys showed intermediate values of work-hardened layer thicknesses. It is also interesting to note that the work-hardened layer of the ADI (alloy 2) is more or less very similar in behaviour to that of the LADI (alloy 8) with a thickness of about 160  $\mu\text{m}$  for both. Again it is suggested that ADI with special heat treatments (austenitising and austempering) can be a material of industrial use.

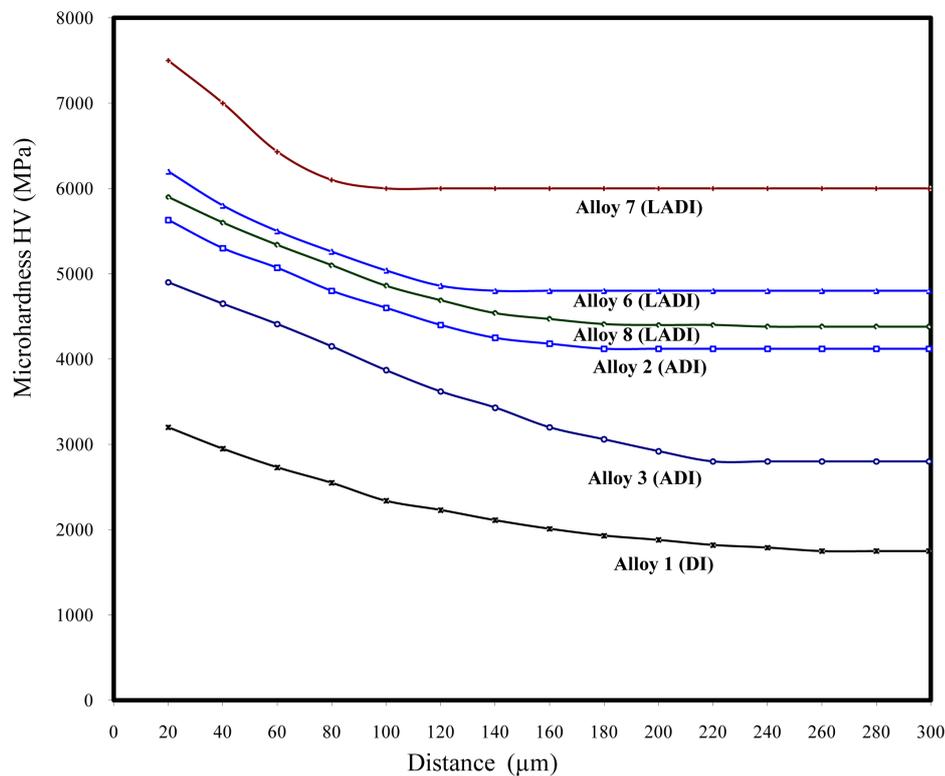
### 3.7. Fracture Surface Observations

Fractographic study was carried out on selected-samples of the investigated alloys. For the sake of comparison alloys 2, 3 (ADI) and alloys 7, 8 (LADI) were selected for fractography since they were subjected to the same austenitising temperature of 1143 K and the same austempering temperatures (573, 673 K). Additionally, the as-cast DI (alloy 1) and as-cast LADI (alloy 6) fracture surfaces were investigated.

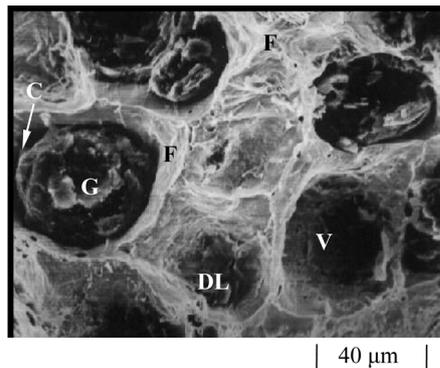
**Failure in tension:** Scanning electron microscopy was used to reveal the fracture surfaces for all investigated alloys (from **Figure 7** to **Figure 10**). **Figure 7** reveals the features of the fracture surface of the conventional DI (alloy 1). The ferrite areas, in the interior of the matrix, showed ductile patterns (symbol “F”). Sometimes small areas of dendritic lobes are also seen on the fracture surface (symbol “DL”). The graphite-nodules are pulled out leaving behind a vacant (symbol “V”). Graphite nodules (symbol “G”) are observed in their sites surrounded by ferrite. It is interesting to note that there is an observable clearance between the nodules and the surrounding ferrite (symbol “C”). This clearance “C” reflects the amount of plastic deformation that occurred prior to fracture. The cavities were generated due to decohesion occurred at graphite/ferrite interface at very low strain level due to the fact that the graphite nodules are weakly bonded to the ferrite matrix surrounding them [36].

**Figure 8** reveals the fractographs of ADI type (alloys 2, 3). **Figure 8(a)** shows cleavage pattern in the brittle area indicated, and small clearance “C” between the graphite-nodule and its cavity. These features reflect the relatively brittle mode of failure. Again vacant sites are seen “V”.

Small area of ductile features can be seen in the **Figure 8(a)** occurring in the RA regions. The phenomenon of white small particles on the surface of graphite nodules in DI was related to carbon precipitation during heat



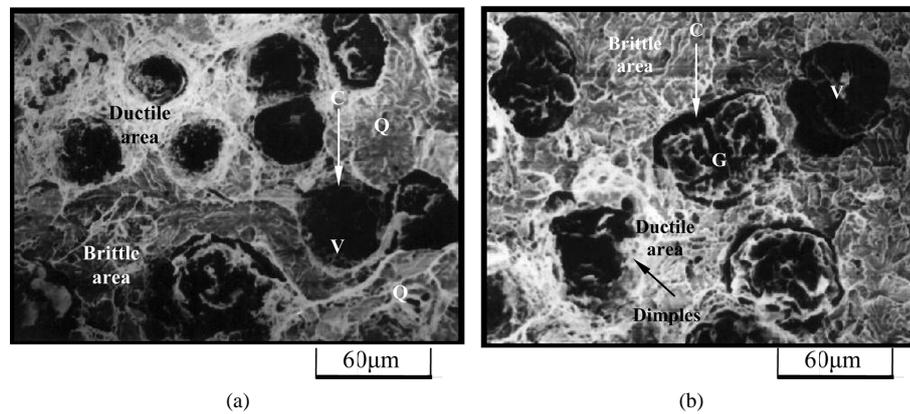
**Figure 6.** Variation of microhardness with distance from worn-surface revealing the thicknesses of the work-hardened layers for the 3 types of alloys under investigation (DI, ADI, and LADI).



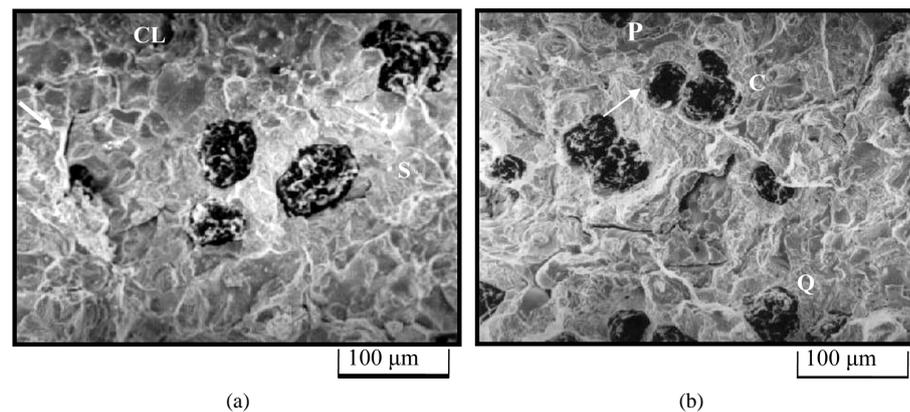
**Figure 7.** SEM-fracture surface of as-cast conventional DI (alloy 1) after failure in tension.

treatment as was indicated by N. Fatahalla *et al.* [37]. The white precipitates on graphite nodules shown in **Figure 8(b)** are due to the excess carbon of retained austenite in the matrix. The change of lower bainite to upper bainite due to raising the austempering temperature from 573 K to 673 K was reflected on the fracture characteristics in **Figure 8(b)** (alloy 3) relative to that observed in **Figure 8(a)** (alloy 2). Since the amount of RA was increased and, additionally, the ductility of upper bainite is higher than that of lower bainite, the ductile mode of fracture covered much larger areas on the fracture surface in **Figure 8(b)** relative to **Figure 8(a)**. Brittle patterns areas are less than that in **Figure 8(a)** and it is to be noted that dimples can clearly be seen at the RA area neighbouring to nodule (cf. **Figure 8(b)**) reflecting moderate ductile mode of fracture in consistence with literature [38].

**Figure 9(a)**, **Figure 9(b)** reveals the fractography of the LADI (alloys 7, 8). Both photos indicate a brittle mode of fracture. However, the clearance between the graphite-nodule and its corresponding cavity is much smaller in **Figure 9(a)** (alloy 7) relative to that of **Figure 9(b)** (alloy 8). This corresponds to the higher hardness



**Figure 8.** SEM-Fracture surfaces of the type ADI austenitised at 1143 K and austempered at (a) 573 K -alloy 2-, (b) 673 K -alloy 3-.



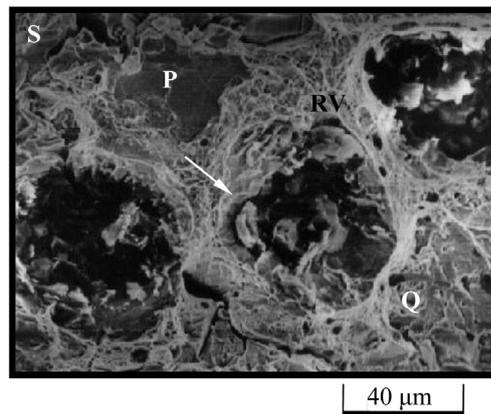
**Figure 9.** SEM-Fracture surfaces of LADI austenitised at 1143 K, and austempered at (a) 573 K -alloy 7-, and (b) 673 K -alloy 8-after failure in tension.

and strength of alloy 7 relative to that of alloy 8. Features in **Figure 9(a)** reveal cleavage pattern and faceted regions (symbol “S”) indicating a brittle mode of fracture. In **Figure 9(b)** the features showed quasi-cleavage and pearlitic areas, and the clearance was relatively wider than that for alloy 7. The relatively larger amount of RA and the more ductile upper bainite do affect the fracture pattern. The effect of some alloying elements on the fracture behaviour of DI was previously investigated [39] supporting the present results.

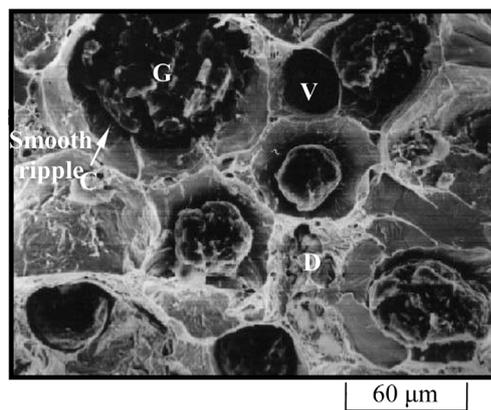
**Figure 10** shows the fracture surface of the as-cast LADI (alloy 6). Fragmentation of graphite can be observed, large areas of quasi-cleavage “Q” and faceted pattern “S”, and pearlite regions “P”. Very tight clearance “C” (indicated by a white arrow) between graphite-nodules and its corresponding cavities reflecting the brittleness of the alloy.

**Failure in impact:** **Figure 11** shows the fracture surface of the conventional as-cast DI (alloy 1) after failure in impact testing (high rate of loading). The surface features still resemble a ductile mode of fracture. Smooth ripple pattern surrounding the graphite nodules in the ferrite regions is observed (cf. **Figure 11**). Additionally, areas of dimples are also shown in the figure. Again we can observe vacant sites “V” referring to pulled nodules and some fragmentation of graphite. The relatively large clearance between nodule and its cavity indicates the degree of plasticity prior to fracture, this result confirms that obtained previously by Fatahalla *et al.* [40].

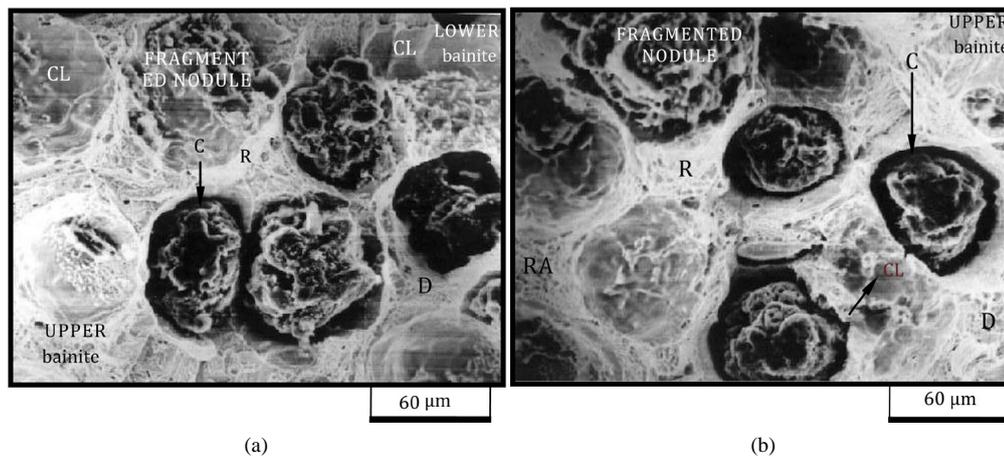
**Figure 12** shows the fracture surfaces of ADI (alloys 2, 3) after failure in impact testing. Both photos reveal a mixed mode of fracture (ductile + brittle). However, in **Figure 12(a)**, cleavage “CL” pattern is revealed in the upper portion (brittle characteristic) while smooth ripple pattern “R” is seen in the lower area of the same photo (ductile feature). In **Figure 12(b)** most of the fracture surface is characterised by a smooth ripple pattern “R” plus regions of dimples “D” reflecting higher ductility. The clearance “C” between the nodules and their corresponding cavities are much tolerated in **Figure 12(b)** rather than that observed in **Figure 12(a)**, reflecting plastic-



**Figure 10.** SEM-Fracture surface of as-cast LADI (alloy 6) after failure in tension.



**Figure 11.** SEM-Fracture surface of as-cast DI (alloy 1) after failure in impact.



**Figure 12.** SEM-Fracture surfaces of ADI type austenitised at 1143 K and austempered at (a) 573 K -alloy 2-, and (b) 673 K-alloy 3-after failure in impact.

ity and higher ductility in the latter. Fragmented graphite nodules are observed in both photos of **Figure 12**. Wolkov *et al.* [41] examined the behaviour of ADI under dynamic load in terms of the fracture mechanics. They found that, as “brittleness” is considered in terms of energy, they ought to classify high strength bainitic ductile-irons as brittle fracture materials.

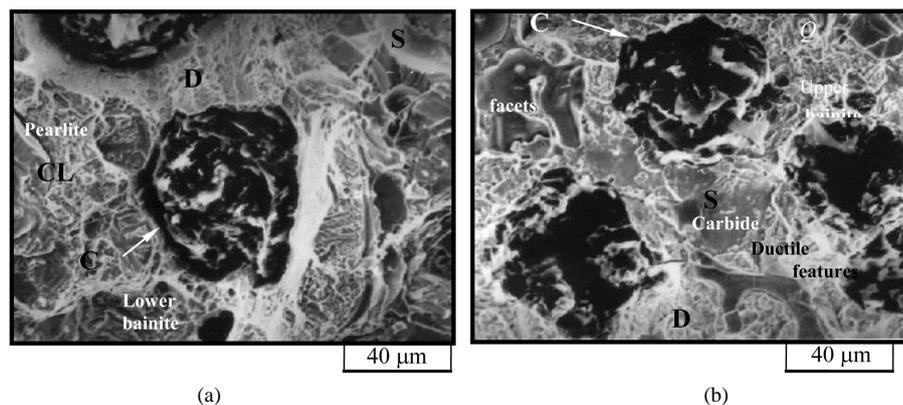
**Figure 13** shows the fracture surfaces of LADI (alloys 7, 8) fractured in impact. All features in both photos of **Figure 13** reveal tendency to brittleness. However, **Figure 13(b)** reveals a bit of ductile fracture features such as

ripple “R” and dimple “D” patterns in the ductile features region of **Figure 13(b)**. Although a small area in **Figure 13(a)** (alloy 7) showed dimple pattern, however, most of the fracture surface is covered with cleavage “CL”, carbide “S”, and hard lower bainite regions. On the other hand, facets and carbide areas were also detected in **Figure 13(b)**. The effect of high strain rate loading (impact) on the fracture phenomenon of LADI (alloys 7, 8) can be clearly seen if we compare the photos in **Figure 13** with those in **Figure 9** (tension).

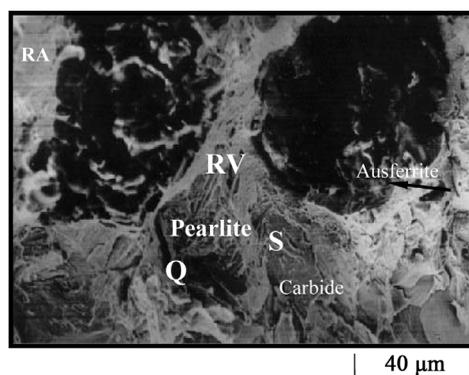
**Figure 14** shows the fracture surface of the as-cast LADI (alloy 6) after failure in an impact test. The river pattern “RV” of ausferrite in between two graphite nodules, can be seen. Also, quasi-cleavage features “Q” that occur on unspecified planes of pearlite and the flat facets “S” of carbide indicate the brittleness of the alloy. Tightness between the nodules and their corresponding cavities, *i.e.* small clearance “C” is another evidence of less plastic deformation prior to fracture. Alloy 6 (LADI) has better tensile strengths, macrohardness and lower impact energy than that of alloy 1 which refers to the blocky carbides located around the graphite nodules in the former. Brittleness and low toughness of alloy 6 (LADI) is believed to refer to segregation of the alloying elements; Ni, Mn, Mo, and Cr and formation of carbides that occurred in the matrix.

#### 4. Conclusions

- 1) Successful production of conventional ductile iron (DI), austempered DI, and low-alloyed DI was under the same experimental conditions.
- 2) Microstructures of all types of DI (8 groups of specimens) were revealed, therefore, investigated for different constituents and correlated with chemical compositions.
- 3) A hardness conversion table was introduced implying Brinell, Vicker’s, and Rockwell hardness for the investigated 3 types of ductile iron.
- 4) The microhardness values for different microconstituents were determined. Microhardness profile was drawn with distance from worn surface to detect the work-hardened layer thicknesses.



**Figure 13.** SEM-Fracture surfaces of LADI austenitised at 1143 K and austempered at (a) 573 K-alloy 7-, (b) 673 K-alloy 8-after failure in an impact test.



**Figure 14.** SEM-Fracture surface of as-cast LADI (alloy 6) after failure in impact.

5) Tensile properties (ultimate, proof, elongation %, and toughness) were determined for all investigated types of ductile-iron.

6) Impact toughness was measured for all alloys under investigation, and these values were compared with toughness values obtained from stress-strain diagrams. Thus the effect of strain rate loading on toughness was investigated.

7) Some wear characteristics were investigated implying weight loss and coefficient of friction for all alloys.

8) Fracture surface observation was conducted using the SEM to reveal the mode of fracture and features of the surface. Correlation of fracture phenomena with other obtained results was introduced.

9) The suitable properties of the ADI alloys, specially heat-treated, compared to that of the LADI, suggested the possibility of using the cheap former alloys replacing the expensive latter type to produce, for instance, rolling mills and other industrial products.

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