

Analytic Solution of MHD Stagnation Point Flow over a Stretching Permeable Surface with Effects of Viscous Dissipation and Joule Heating

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Abstract

A mathematical analysis is investigated to obtain an analytic solution of magneto hydrodynamic stagnation-point flow towards permeable stretching surface with viscous dissipation and joule heating. In the presence of uniform suction, a transverse magnetic field normal to the surface is applied when the surface is stretched in its own plane with a velocity proportional to the distance from the stagnation-point. The governing nonlinear momentum and energy equations are solved by homotopy analysis method (HAM) to obtain the complete analytic solution and a good agreement is found. The convergence region shows the validity of the HAM solutions. It is observed that the velocity at a point increases/decreases more with increase in the magnetic parameter when the free stream velocity is greater/less than the stretching velocity in presence of suction. An interesting result of the analysis is that, in the presence of suction parameter, the temperature increases with the increase in magnetic parameter at a certain distance from the stretching surface. Near stagnation point on the surface, heat flows from the surface to the fluid and far from the stagnation-point, heat flows from the fluid to surface due to combining effect of ohmic dissipation and strain energy inside the boundary layer.

Keywords

MHD, Suction, Stretching Surface, Homotopy Analysis Method

1. Introduction

The laminar flow and heat transfer of an incompressible fluid over a stretching surface have significant applica-

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tions in several technologies of process industry. A number of manufacturing industries process and include both metal and polymer sheets during the production of sheeting materials. This kind of motion of a fluid close to the rigid or stretching surface can be investigated from a standard boundary layer equation. The quality of the final product depends to a large extent on the rate of heat transfer at the stretching surface. It is, therefore, of interest to know the fluid behavior over a stretching surface which determines the rate of cooling. Hiemenz [1] first analyzed an exact solution of the stagnation point flow of a viscous fluid over a rigid surface. The axisymmetric three-dimensional cases for the rigid surface are investigated by Homann [2]. The steady flow of an incompressible viscous fluid over an elastic sheet was studied by Crane [3] where the flow being caused solely by stretching of sheet in its own plane. The temperature distribution in the flow over a stretching surface subject to uniform heat flux was studied by Dutta et al. [4]. The steady two-dimensional orthogonal stagnation point flow of an incompressible viscous fluid towards a stretching surface which is held at constant temperature, was investigated by Mohapatra and Gupta [5]. Later, Mohapatra and Gupta [6] extended his work to study the momentum and heat transfer characteristic of stagnation-point flow of a viscoelastic fluid over a stretching surface. Hassanien [7] analyzed the similarity solution for the flow and heat transfer of a viscoelastic fluid over a stretching sheet. The temperature distribution for the stagnation point flow of a micro polar fluid past a stretching sheet was investigated by Nazar et al. [8].

In the region of adverse pressure gradient the flow in a boundary layer separates. Due to separation, it leads to increase in drag on the body immersed in the flow and adversely affects heat transfer from the surface of the body. Flow separation can be prevented by applying suction to remove the directed fluid particle from the boundary layer. On the other hand, the wall shear stress and hence friction drag is reduced by blowing. Wall suction and blowing have many applications such as turbine blade cooling, transition delay and prevention separation. The effect of suction on the Hiemenz problem was studied by Schlichting and Bussmann [9]. An approximate solution to the problem of uniform suction is given by Ariel [10]. The effect of uniform suction on the Homann problem where the flat plate oscillates in its own plane is considered by Weidman and Mahalingam [11]. The effect of uniform suction or injection on the two- or three-dimensional stagnation-point flow of a conducting fluid was given by Attia [12] [13] in the presence of an externally applied uniform magnetic field. The effect of uniform suction or injection on two-dimensional stagnation point flow towards a stretching surface with heat generation was given by Attia and Seddeek [14].

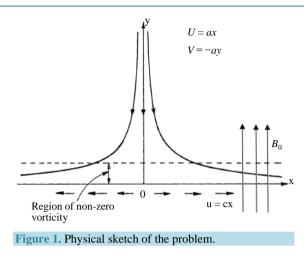
Electrically conducting fluid like a liquid metal, electrolyte or plasma under the influence of magnetic field induces electric current. This type of fluid has many applications in industries, for example, to drive flow, induce stirring, levitation or control heat transfer and turbulence. An exact similarity solution to the MHD boundary layer equations for the steady two-dimensional flow of an electrically conducting incompressible fluid due to the stretching of a plane elastic surface in the presence of a uniform transverse magnetic field was given by Pavlov [15]. Chakrabarti and Gupta [16] studied the MHD flow of a Newtonian fluid over a stretching sheet. In the presence of uniform transverse magnetic field Andersson [17] investigated the MHD flow of a viscoelastic fluid past a stretching surface. Liao [18] developed Homotopy analysis method (HAM) which is one of the most successful and efficient methods to solve a non-linear differential equation. This method is based on a fundamental concept in topology, *i.e.*, Homotopy (Hilton [19]) which is widely used in numerical techniques (Grigolyuk and Shalashilin [20]).

In this present work, HAM is employed to find the new analytic solution of the effect of uniform transverse magnetic field as well as suction or blowing on a steady two-dimensional orthogonal stagnation point flow of an electrically conducting fluid towards a permeable stretching surface. For a very small magnetic Reynolds number the induced magnetic field can be neglected. We assume the wall and free stream temperature to be constant.

2. Flow Analysis

Consider the two dimensional stagnation point flow of a viscous incompressible electrically conducting fluid impinging orthogonally to a permeable stretching surface coinciding with plane y = 0, the being in the region y > 0. Two equal and opposite forces are applied to the sheet along the *x*-axis such that the surface is stretched keeping the origin fixed as shown in **Figure 1**. The flow is permeated by a uniform magnetic field of strength B_0 applied transverse to the plate. The plate is assumed to be electrically non-conducting and is subject to uniform suction or blowing.

Let u and v are the velocity components along x and y directions, respectively. The velocity in the free flow is given by



$$U(x) = ax, \quad V(y) = -ay \tag{1}$$

where a > 0 is proportional to the free stream velocity far away from the surface. Under these assumptions, along with the Boussinesq approximation and boundary layer approximations, the governing equations for the steady two-dimensional stagnation point flows by introducing Lorentz force are

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0 \tag{2}$$

$$u\frac{\partial u}{\partial x} + v\frac{\partial u}{\partial y} = U\frac{\partial U}{\partial x} + v\frac{\partial^2 u}{\partial y^2} + \frac{\sigma B_0^2}{\rho} (U(x) - u)$$
(3)

$$u\frac{\partial T}{\partial x} + v\frac{\partial T}{\partial y} = \kappa \frac{\partial^2 T}{\partial y^2} + \frac{\mu}{\rho c_P} \left(\frac{\partial u}{\partial y}\right)^2 + \frac{\sigma}{\rho c_P} B_0^2 \left(u - ax\right)^2$$
(4)

where T represents the temperature and ρ , μ , ν , σ , κ and c_p are the density, the viscosity, the kinematic viscosity, these electric conductivity of the fluid, thermal diffusivity and the specific heat of the fluid at constant pressure, respectively, and U(x) is horizontal component of the inviscid potential flow velocity over the body surface. At the stretching surface, the velocity in the x direction is equal to the moving surface velocity cx, where c is a positive constant. So boundary conditions are

$$u = cx, \quad v = -v_0, \quad T = T_W \quad \text{at} \quad y = 0$$
 (5)

$$u \to U(x), \quad T \to T_{\infty} \quad \text{as} \quad y \to \infty$$
 (6)

where v_0 is the constant velocity at the plate with $v_0 > 0$ for suction and $v_0 < 0$ for blowing and T_w and T_{∞} are constants with $T_W > T_{\infty}$.

To examine the flow regime, we introduce the following similarity transformation

$$u = cxf'(\eta), \quad v = -\sqrt{vc}f(\eta), \quad \theta(x,\eta) = \frac{T(x,y) - T_{\infty}}{T_{W} - T_{\infty}}, \quad \eta = y\sqrt{\frac{c}{v}}$$
(7)

Here prime denotes differentiation with respect to η . Using above equation, the continuity Equation (2) is satisfied automatically and Equations (3) and (4) becomes

$$f''' + ff'' - f'^{2} + H^{2}\left(\frac{a}{c} - f'\right) + \frac{a^{2}}{c^{2}} = 0$$
(8)

$$xf'\frac{\partial\theta}{\partial x} - f\frac{\partial\theta}{\partial \eta} = \frac{\kappa}{\nu}\frac{\partial^2\theta}{\partial \eta^2} + \frac{c^2x^2(f'')^2}{c_P(T_W - T_\infty)} + \frac{\sigma B_0^2 c x^2 \left(f' - \frac{a}{c}\right)^2}{\rho c_P(T_W - T_\infty)}$$
(9)

where $H = B_0 \sqrt{\frac{\sigma}{\rho c}}$ is the Hartman Number. Setting

$$\theta(x,\eta) = \theta_0(\eta) + \frac{cx^2}{\nu} \theta_1(\eta)$$
(10)

In Equation (9) and equating the coefficients of x^0 and x^2 we obtain the following equations for $\theta_0(\eta)$ and $\theta_1(\eta)$:

$$\theta_0'' + \Pr f \theta_0' = 0 \tag{11}$$

$$\theta_1'' + \Pr\left(f\theta_1' - 2f'\theta_1\right) = -\Pr.E.\left(f''\right)^2 - \Pr.E.H^2\left(f' - \frac{a}{c}\right)^2$$
(12)

where $E = vc/c_p (T_w - T_{\infty})$, is the Eckert number which is a measure of viscous dissipation.

The corresponding boundary conditions (5) and (6) can be written as

$$f(0) = A, \quad f'(0) = 1, \quad f'(\infty) = \frac{a}{c}$$
 (13)

$$\theta_0(0) = 1, \quad \theta_0(\infty) = 0 \tag{14}$$

$$\theta_1(0) = 0, \quad \theta_1(\infty) = 0 \tag{15}$$

Here $A = \pm \frac{v_0}{\sqrt{cv}}$ is the suction parameter (A > 0 for suction and A < 0 for injection). For boundary layer

flow, the wall skin friction τ_w is given by

$$\tau_{w} = \left(\mu \frac{\partial u}{\partial y}\right)_{y=0}$$
(16)

Using U(x) = ax as a characterized velocity, the skin friction coefficient C_f can be defined as

$$C_f = \frac{\tau_w}{\rho U^2} \tag{17}$$

Substituting (7) and (16) in (17) we get

$$C_f \sqrt{Re_x} = f''(0) \tag{18}$$

where $Re_x = xU/\nu$ is the local Reynolds number.

3. Homotopy Analysis Solution

The velocity $f(\eta)$ and the temperature $\theta_0(\eta)$, $\theta_1(\eta)$ can be represented in terms of a set of base function

$$\left\{\eta^{m} \exp\left(-n \left|\frac{a}{c} - 1\right| \eta\right) : m \ge 0, n \ge 0 \text{ are integers}\right\}$$
(19)

as

$$f(\eta) = \sum_{n=0}^{\infty} \sum_{m=0}^{\infty} a_{n,m} \eta^m \exp\left(-n \left|\frac{a}{c} - 1\right| \eta\right)$$
(20)

$$\theta_0(\eta) = \sum_{n=0}^{\infty} \sum_{m=0}^{\infty} b_{n,m} \eta^m \exp\left(-n \left| \frac{a}{c} - 1 \right| \eta\right)$$
(21)

and

$$\theta_1(\eta) = \sum_{n=0}^{\infty} \sum_{m=0}^{\infty} c_{n,m} \eta^m \exp\left(-n \left| \frac{a}{c} - 1 \right| \eta\right)$$
(22)

where $a_{n,m}$, $b_{n,m}$ and $c_{n,m}$ are the coefficients. Looking into the boundary condition and the base function we choose the initial guess $f_0(\eta)$, $\theta_{0,0}(\eta)$ and $\theta_{1,0}(\eta)$ as

$$f_0(\eta) = A + \frac{a}{c}\eta - \left[1 - \exp\left(-\left|\frac{a}{c} - 1\right|\eta\right)\right] \operatorname{sgn}\left(\frac{a}{c} - 1\right)$$
(23)

$$\theta_{0,0}(\eta) = \exp\left(-\left|\frac{a}{c} - 1\right|\eta\right)$$
(24)

and

$$\theta_{1,0}(\eta) = \exp\left(-\left|\frac{a}{c} - 1\right|\eta\right) \left[1 - \exp\left(-\left|\frac{a}{c} - 1\right|\eta\right)\right]$$
(25)

The auxiliary linear operators L_f , L_{θ_0} and L_{θ_1} can be taken as

$$L_{f}\left[f\left(\eta;q\right)\right] = \frac{\mathrm{d}^{3}f\left(\eta;q\right)}{\mathrm{d}\eta^{3}} + \left|\frac{a}{c} - 1\right|\frac{\mathrm{d}^{2}f\left(\eta;q\right)}{\mathrm{d}\eta^{2}}$$
(26)

$$L_{\theta_0}\left[\theta(\eta;q)\right] = L_{\theta_1}\left[\theta(\eta;q)\right] = \frac{\mathrm{d}^2\theta(\eta;q)}{\mathrm{d}\eta^2} + \left|\frac{a}{c} - 1\right|\frac{\mathrm{d}\theta(\eta;q)}{\mathrm{d}\eta}$$
(27)

where $q \in [0,1]$ is an embedding parameter. The auxiliary linear operators L_f , L_{θ_0} and L_{θ_1} satisfies

$$L_{f}\left[C_{1}+C_{2}\eta+C_{3}\exp\left(-\left|\frac{a}{c}-1\right|\eta\right)\right]=0$$
(28)

$$L_{\theta_0}\left[C_4 + C_5 \exp\left(-\left|\frac{a}{c} - 1\right|\eta\right)\right] = 0$$
⁽²⁹⁾

$$L_{\theta_{1}}\left[C_{6}+C_{7}\exp\left(-\left|\frac{a}{c}-1\right|\eta\right)\right]=0$$
(30)

where C_i , (i=1-7) are arbitrary constants.

Corresponding to the governing differential Equations (8), (11) and (12), we define the non-linear operators as

$$N_{f}\left[\hat{f}(\eta;q)\right] = \frac{\partial^{3}\hat{f}(\eta;q)}{\partial\eta^{3}} + \hat{f}\frac{\partial^{2}\hat{f}(\eta;q)}{\partial\eta^{2}} - \left(\frac{\partial\hat{f}(\eta;q)}{\partial\eta}\right)^{2} - H^{2}\frac{\partial\hat{f}(\eta;q)}{\partial\eta} + H^{2}\left(\frac{a}{c}\right) + \frac{a^{2}}{c^{2}}$$
(31)

$$N_{\theta_0} \left[\hat{\theta}(\eta; q), \hat{f}(\eta; q) \right] = \frac{1}{\Pr} \frac{\partial^2 \hat{\theta}(\eta; q)}{\partial \eta^2} + \hat{f} \frac{\partial \hat{\theta}(\eta; q)}{\partial \eta}$$
(32)

$$N_{\theta_{1}}\left[\hat{\theta}(\eta;q),\hat{f}(\eta;q)\right] = \frac{1}{\Pr}\frac{\partial^{2}\hat{\theta}(\eta;q)}{\partial\eta^{2}} + \hat{f}\frac{\partial\hat{\theta}(\eta;q)}{\partial\eta} - 2\hat{\theta}\frac{\partial\hat{f}(\eta;q)}{\partial\eta} + E\left(\frac{\partial^{2}\hat{f}(\eta;q)}{\partial\eta^{2}}\right)^{2} + EH^{2}\left(\frac{\partial\hat{f}(\eta;q)}{\partial\eta} - \frac{a}{c}\right)^{2} (33)$$

If h_f , h_{θ_0} , h_{θ_1} denotes the non-zero auxiliary parameters and $G_f(\eta)$, $G_{\theta_0}(\eta)$, $G_{\theta_1}(\eta)$ denotes the auxiliary function, we construct the zero-th order deformation equations

$$(1-q)L_{f}\left[\hat{f}(\eta;q)-f_{0}(\eta)\right]=h_{f}G_{f}(\eta)N_{f}\left[\hat{f}(\eta;q)\right]$$
(34)

$$(1-q)L_{\theta_0}\left[\hat{\theta}_0(\eta;q) - \theta_0(\eta)\right] = h_{\theta_0}G_{\theta_0}(\eta)N_{\theta_0}\left[\hat{\theta}_0(\eta;q)\right]$$
(35)

$$(1-q)L_{\theta_{1}}\left[\hat{\theta}_{1}(\eta;q)-\theta_{1}(\eta)\right]=h_{\theta_{1}}G_{\theta_{1}}(\eta)N_{\theta_{1}}\left[\hat{\theta}_{1}(\eta;q)\right]$$
(36)

subject to the boundary conditions,

$$\hat{f}(0;q) = A, \quad \frac{\partial \hat{f}(\eta;q)}{\partial \eta} \bigg|_{\eta=0} = 1, \quad \frac{\partial \hat{f}(\eta;q)}{\partial \eta} \bigg|_{\eta\to\infty} = \frac{a}{c}$$
(37)

$$\hat{\theta}_{0}(0;q) = 1, \quad \hat{\theta}_{1}(0;q) = 0, \quad \hat{\theta}_{0}(\eta;q)\Big|_{\eta \to \infty} = 0, \quad \hat{\theta}_{1}(\eta;q)\Big|_{\eta \to \infty} = 0$$
(38)

where

$$G_{f}(\eta) = G_{\theta_{0}}(\eta) = G_{\theta_{1}}(\eta) = \exp\left(-\left|\frac{a}{c}-1\right|\eta\right)$$
(39)

When q = 0 and q = 1 we have

$$\hat{f}(\eta;0) = f_0, \quad \hat{f}(\eta;1) = f(\eta)$$
(40)

$$\hat{\theta}_0(\eta;0) = \theta_{0,0}, \quad \hat{\theta}(\eta;1) = \theta_0(\eta) \tag{41}$$

$$\hat{\theta}_1(\eta;0) = \theta_{1,0}, \quad \hat{\theta}(\eta;1) = \theta_1(\eta) \tag{42}$$

Note that when the embedding parameter q increases from 0 to 1, $\hat{f}(\eta;q)$, $\hat{\theta}_0(\eta;q)$ and $\hat{\theta}_1(\eta;q)$ varies from the initial approximations $f_0(\eta)$, $\theta_{0,0}(\eta)$ and $\theta_{1,0}(\eta)$ to the exact solution $f(\eta)$, $\theta_0(\eta)$ and $\theta_1(\eta)$ respectively. By Taylor's theorem we express $\hat{f}(\eta;q)$, $\hat{\theta}_0(\eta;q)$ and $\hat{\theta}_1(\eta;q)$ in the power series of the embedding parameter q.

$$\hat{f}(\eta;q) = f_0(\eta) + \sum_{m=1}^{\infty} f_m(\eta) q^m$$
(43)

$$\hat{\theta}_0(\eta;0) = \theta_{0,0}(\eta) + \sum_{m=1}^{\infty} \theta_{0,m}(\eta) q^m$$
(44)

$$\hat{\theta}_{1}(\eta;0) = \theta_{1,0}(\eta) + \sum_{m=1}^{\infty} \theta_{1,m}(\eta) q^{m}$$

$$\tag{45}$$

where

$$f_m(\eta) = \frac{1}{m!} \frac{\partial^m f(\eta; q)}{\partial \eta^m} \bigg|_{q=0}$$
(46)

$$\theta_{0,m}(\eta) = \frac{1}{m!} \frac{\partial^m \theta_0(\eta; q)}{\partial \eta^m} \bigg|_{q=0}$$
(47)

$$\theta_{1,m}(\eta) = \frac{1}{m!} \frac{\partial^m \theta_1(\eta;q)}{\partial \eta^m} \bigg|_{q=0}$$
(48)

If the auxiliary parameters h_f , h_{θ_0} and h_{θ_1} are properly chosen such that the series converges at q = 1 then we have

$$f(\eta) = f_0(\eta) + \sum_{m=1}^{\infty} f_m(\eta), \qquad (49)$$

$$\theta_0(\eta) = \theta_{0,0}(\eta) + \sum_{m=1}^{\infty} \theta_{0,m}(\eta), \qquad (50)$$

$$\theta_{1}(\eta) = \theta_{1,0}(\eta) + \sum_{m=1}^{\infty} \theta_{1,m}(\eta).$$
(51)

Differentiating the zero-th-order Equations (34), (35), (36) m-times with respect to the embedding parameter q, then setting q = 0, and finally dividing by m!, we have m-th-order deformation equation

$$L_{f}\left\lfloor f_{m}\left(\eta\right)-\chi_{m}f_{m-1}\left(\eta\right)\right\rfloor = h_{f}G_{f}\left(\eta\right)R_{m}^{f}\left(\eta\right)$$

$$\tag{52}$$

$$L_{\theta_{0}}\left[\theta_{0,m}\left(\eta\right)-\chi_{m}\theta_{0,m-1}\left(\eta\right)\right]=h_{\theta_{0}}G_{\theta_{0}}\left(\eta\right)R_{m}^{\theta_{0}}\left(\eta\right)$$
(53)

$$L_{\theta_{l}}\left[\theta_{l,m}\left(\eta\right)-\chi_{m}\theta_{l,m-1}\left(\eta\right)\right]=h_{\theta_{l}}G_{\theta_{l}}\left(\eta\right)R_{m}^{\theta_{l}}\left(\eta\right)$$
(54)

subject to the boundary conditions

$$f_m(0) = 0, \quad f'_m(0) = 0, \quad f'_m(\infty) = 0$$
 (55)

$$\theta_{0,m}(0) = 0, \quad \theta_{0,m}(\infty) = 0 \tag{56}$$

$$\theta_{1,m}(0) = 0, \quad \theta_{1,m}(\infty) = 0 \tag{57}$$

where

$$R_{m}^{f}(\eta) = f_{m-1}^{\prime\prime\prime} + \sum_{n=0}^{m-1} f_{n}f_{m-1-n}^{\prime\prime} - f_{n}^{\prime}f_{m-1-n}^{\prime} - H^{2}f_{m-1}^{\prime} + (1-\chi_{m})\left(H^{2}\frac{a}{c} + \frac{a^{2}}{c^{2}}\right),$$
(58)

$$R_m^{\theta_0}(\eta) = \frac{1}{\Pr} \theta_{0,m-1}'' + \sum_{n=0}^{m-1} f_n \theta_{0,m-1-n}'' , \qquad (59)$$

$$R_{m}^{\theta_{1}}(\eta) = \frac{1}{\Pr}\theta_{1,m-1}'' - 2\frac{a}{c}EH^{2}f_{m-1}' + EH^{2}\frac{a^{2}}{c^{2}}(1-\chi_{m}) + \sum_{n=0}^{m-1}f_{n}\theta_{1,m-1-n}' - 2f_{n}'\theta_{1,m-1-n} + Ef_{n}'f_{m-1-n}'' + EH^{2}f_{n}'f_{m-1-n}''$$
(60)

and

$$\chi_m = \begin{cases} 0, & m \le 1\\ 1, & m > 1 \end{cases}$$
(61)

Let $f_m^*(\eta)$, $\theta_{0,m}^*(\eta)$, $\theta_{1,m}^*(\eta)$ denote the special solution of Equations (52), (53) and (54). We have

$$f_{m}^{*}(\eta) = \chi_{m} f_{m-1}(\eta) + h_{f} L_{f}^{-1} \Big[G_{f}(\eta) R_{m}^{f}(\eta) \Big],$$
(62)

$$\theta_{0,m}^{*}(\eta) = \chi_{m}\theta_{0,m-1}(\eta) + h_{\theta_{0}}L_{\theta_{0}}^{-1}\Big[G_{\theta_{0}}(\eta)R_{m}^{\theta_{0}}(\eta)\Big],$$
(63)

$$\theta_{1,m}^{*}\left(\eta\right) = \chi_{m}\theta_{1,m-1}\left(\eta\right) + h_{\theta_{1}}L_{\theta_{1}}^{-1}\left[G_{\theta_{1}}\left(\eta\right)R_{m}^{\theta_{1}}\left(\eta\right)\right].$$
(64)

where L_f^{-1} , $L_{\theta_0}^{-1}$, $L_{\theta_0}^{-1}$ denotes the inverse operator of L_f , L_{θ_0} , L_{θ_1} respectively. Thus the solution of the *m*-th-order deformation equations as

$$f_{m}(\eta) = f_{m}^{*}(\eta) + C_{1} + C_{2}\eta + C_{3}\exp\left(-\frac{|a|}{c} - 1|\eta\right)$$
(65)

$$\theta_{0,m}(\eta) = \theta_{0,m}^*(\eta) + C_4 + C_5 \exp\left(-\left|\frac{a}{c} - 1\right|\eta\right)$$
(66)

$$\theta_{1,m}\left(\eta\right) = \theta_{1,m}^{*}\left(\eta\right) + C_{6} + C_{7} \exp\left(-\left|\frac{a}{c} - 1\right|\eta\right)$$
(67)

Here the integral constants C_i , $(i=1,2,3,\dots,7)$ are determined by the boundary conditions (55), (56) and (57) as

$$C_{1} = -C_{3} - f_{m}^{*}(0), \quad C_{2} = 0, \quad C_{3} = \frac{1}{\left|\frac{a}{c} - 1\right|} \frac{\partial f_{0,m}^{*}(\eta)}{\partial \eta} \bigg|_{\eta = 0}$$
(68)

$$C_4 = 0, \quad C_5 = -\theta_{0\,m}^*(0), \quad C_6 = 0, \quad C_7 = -\theta_{1\,m}^*(0)$$
 (69)

Therefore, it is easy to solve linear non-homogeneous Equations (52), (53) and (54) subject to the boundary conditions (55), (56) and (57) by using MATHEMATICA one after other in the order $m = 1, 2, 3, \cdots$

4. Convergence of Homotopy Solutions

We observed that the Equations (52), (53) and (54) consist of the auxiliary parameters h_f , h_{θ_0} and h_{θ_1} . These auxiliary parameters are important in controlling the convergence of series solutions the functions $f(\eta)$, $\theta_0(\eta)$ and $\theta_1(\eta)$. For this analysis, the admissible values of h_f , h_{θ_0} and h_{θ_1} curves are plotted for 12th-order of approximations. Figure 2 shows the admissible values of h_f of f''(0) for 12th-order of approximation with a/c = 0.1 for several values of H. It is observed that h_f curves have parallel lines segment that correspond to the regions $-1.5 \le h_f \le -0.3$, $-1.0 \le h_f \le -0.2$ and $-0.6 \le h_f \le -0.3$ for H = 0,1,2, respectively. To analyze the admissible range of h_{θ_0} and h_{θ_1} of $\theta'_0(0)$ and $\theta'_1(0)$ are displayed in Figure 3 for 12th-order of approximation with a/c = 0.1, H = 2 and $\Pr = 5.0$. It is observed that the converging ranges are $-2.3 \le h_{\theta_0} \le -0.5$ and $-1.8 \le h_{\theta_1} \le -0.7$. From these figures we have examined that by choosing the values of convergence control parameter from these range we will get the convergent result up to higher decimal place.

5. Results and Discussion

The nonlinear boundary value problem given by Equations (8), (11) and (12) with boundary conditions (13), (14), and (15) cannot be solved in closed form. So these equations are solved by Homotopy Analysis Method (HAM) to get an analytical solution. Extensive calculation has been performed by using symbolic computation software MATHEMATICA to obtain the velocity and temperature profiles for various values of physical parameter such as Suction parameter A, Hartmann Number H, Prandtl number Pr, Eckert number E, (which characterizes viscous dissipation in the flow) and the parameter a/c (ratio of external velocity and stretching velocity). Figure 4 shows the variation of horizontal velocity component with distance from the surface for several values of Hartmann Number H in the presence of suction parameter A = 1.0 and a/c = 0.1 *i.e.*, stretching velocity more than the straining velocity. It can be viewed that horizontal velocity decreases with increase in Hartmann Number H. It is interesting that the flow has inverted boundary layer structure because of stretching velocity cx exceeds the velocity of external stream ax (*i.e.* a/c < 1). Whereas, in the presence of suction parameter A = 1.0 and a/c = 3, the velocity increases with increase in H, as shown in Figure 5.

Table 1 shows the value of dimensionless wall shear stress *i.e.*, skin-friction coefficient (see Equation (18)) for several values of H and a/c in the presence of suction. It gives that for fixed value of suction parameter A = 1, the magnitude of the wall shear stress increases with increase in H and a/c. In order to have correctness of our analytical result, we have examined the comparison result in the value of -f''(0) for different values of A and H in the present study and with the corresponding values by Mahapatra *et al.* [21] (shown in Table 1).

It is observed that during the motion of electrically conduction fluid, a certain amount of energy is stored up in the fluid as strain energy and some energy is lost due to viscous and ohmic dissipation. So in the fluid flow, energy is balanced by the internal energy, the conduction of heat, the convection of heat with the flow, the generation of heat with the flow, the generation of heat through viscous and ohmic dissipation, the strain (or deformation) energy in the fluid and energy due to wall suction effect. At a distance X (Dimensionless length) from stagnation-point, the temperature distribution consists of two functions, first part of temperature distribution denoted by $\theta_0(\eta)$ which depends on the Hartmann Number and suction parameter and second part denoted by $\theta_1(\eta)$ which depends on viscous dissipation and Joule Heating.

In the presence of suction parameter A = 1.0, the variation of temperature distribution $\theta_0(\eta)$ with η is displayed in **Figure 6** for several values of Hartman Number *H* with Pr = 5.0, a/c = 0.1. It is observed that for fixed value of Pr, A = 1.0, the temperature at point increases with increase in the Hartman Number when a/c < 0.1. But **Figure 7** shows that when a/c > 1, $\theta_0(\eta)$ decreases with increase in *H*. In the presence of wall

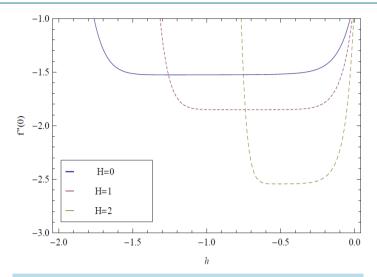


Figure 2. The h_f curve of f''(0) at the 12th order of approximation.

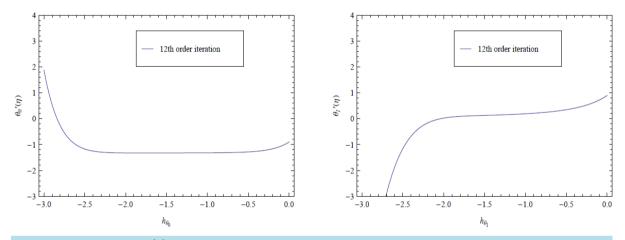


Figure 3. The h curve of $\theta'(0)$ at the 12th order of approximation.

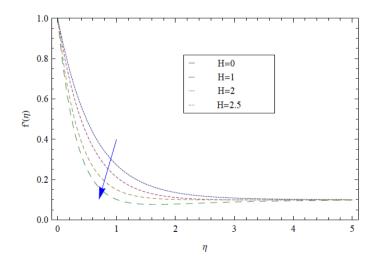


Figure 4. Variation of $f'(\eta)$ with η for several values of H with a/c = 0.1, A = 1.0.

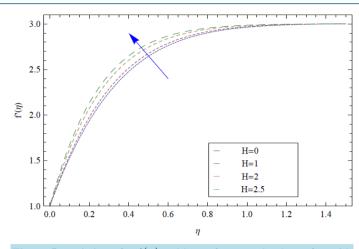


Figure 5. Variation of $f'(\eta)$ with η for several values of H with a/c = 3.0, A = 1.0.

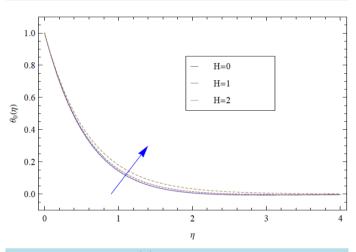


Figure 6. Variation of $\theta_0(\eta)$ with η for several values of *H* with a/c = 0.1, A = 1.0.

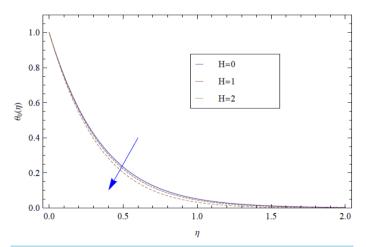


Figure 7. Variation of $\theta_0(\eta)$ with η for several values of *H* with a/c = 3.0, A = 1.0.

Table 1. Values of $f''(0)$.				
a/c	Н	Α	T. R. Mahapatra et al. [2001]	Present Result
<u>.</u>	0	0	-0.9643	-0.964443
	2	0	-2.0198	-2.01671
0.1	3	0	-2.8222	-2.82039
0.1	0	1	-	-1.50512
	2	1	-	-2.53697
	3	1	-	-3.26112
3.0	0	0	4.6800	4.68192
	2	0	6.0952	6.09427
	3	0	7.4891	7.48876
	0	1	-	5.94725
	2	1	-	7.33498
	3	1	-	8.74062

suction, viscous and ohmic dissipation and stored deformation energy in the flow (E = 0), the variation of the temperature distribution given by $\theta_1(\eta)$ with η is shown in **Figure 8**, for several values of *H* with Pr = 5.0 and a/c > 1. It is observed that a novel result, that in the presence of suction parameter A = 1.0, $\theta_1(\eta)$ increases with the increase in Hartmann Number *H* where as in the absence of suction parameter Mahapatra *et al.* [22] shows that in the vicinity of the stretching surface $\theta_1(\eta)$ increases with increase in *H* but after a certain distance from the stretching plate, the trend reverse. **Figure 9** depicts that the variation of $\theta_1(\eta)$ with η for several values of *H* with Pr = 5.0 and A = 1.0 when a/c = 0.1.

The dimensionless rate of heat transfer at the surface $\eta = 0$ given $-\theta'(X, 0)$ from Equation (10) is derived as

$$-\theta'(X,0) = -\theta'_0(0) - X^2 \theta'_1(0)$$
(70)

where $X = \left(\frac{c}{v}\right)^{1/2} x$. The dimensionless heat flux at the surface $-\theta'(X,0)$ is computed from Equation (70) and displayed in **Figure 10** and **Figure 11** corresponding two distinct locations (X = 0.1) and (X = 2.0) on the stretching surface. It is observed from **Figure 10** that the heat flows from the surface to fluid at the small value X = 0.1 (for several values of *H*), because $-\theta'(X,0)$. On the side, for X = 0.1, it is seen in **Figure 11** that the heat flow from the fluid to the stretching surface for several value of H because $-\theta'(X,0)$ is negative. It is more interesting that heat flux is more in the presence of suction. This novel result can be interpreted physically that for small value of X = 0.1, the term $-X^2\theta'_1(X,0)$ becomes small and both ohmic dissipation and strain energy in the flow are not playing important roles to generate heat. So for small value of X, heat flows from the surface to fluid since $T_w > T_\infty$. But for the large value of X = 3.0, the term $-X^2\theta'_1(X,0)$ has significant role due combine effect of ohmic dissipation and strain energy inside the boundary layer which causes to generate sufficient heat inside the boundary layer. Thus, the temperature of the fluid near the surface exceeds the surface temperature T_w . So heat flows from fluid to surface even when $T_w > T_\infty$. In this case it is also observed that heat flux increase with suction for fixed value of *H*.

Since, for small value of X, heat flows from the surface to the fluid and for the large value of X, heat flows from the fluid to surface, so it can be observed in Figure 12 that at a certain value of $X = X_0$ on the stretching surface, heat flux vanishes. It is observed from this figure that value of X_0 decreases with increase in H or E. This interesting result admits of a physical interpretation. Near stagnation point, heat flows from the surface to the fluid but due to increase in the Hartmann number and Ekart Number, ohmic dissipation and strain energy will increase and that will cause to generate the heat inside the boundary layer. So that the value of X_0 will decrease with increase in H or E.

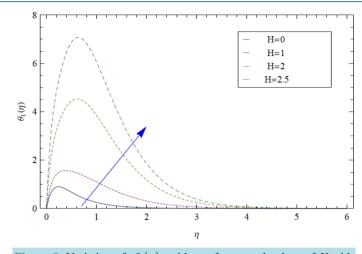


Figure 8. Variation of $\theta_1(\eta)$ with η for several values of *H* with a/c = 3.0, Pr = 5.0, E = 2.0 and A = 1.0.

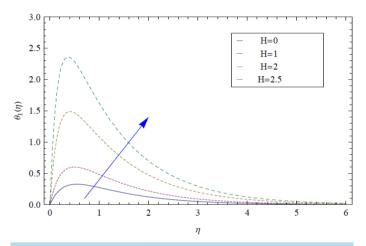


Figure 9. Variation of $\theta_1(\eta)$ with η for several values of *H* with a/c = 0.1, Pr = 5.0, E = 2.0 and A = 1.0.

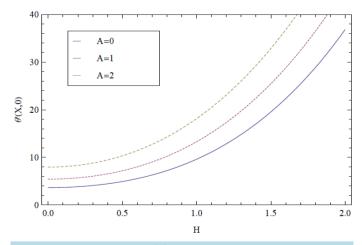


Figure 10. Variation of $\theta'(0)$ with *H* for several values of *A* at X = 2.0 with a/c = 0.1, Pr = 5.0 and E = 2.0.

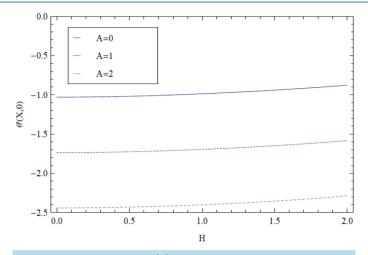
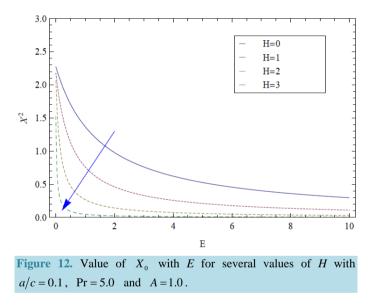


Figure 11. Variation of $\theta'(0)$ with *H* for several values of *A* at X = 0.1 with a/c = 0.1, Pr = 5.0 and E = 2.0.



6. Conclusion

The present study describes the effect of viscous dissipation and joule heating on magneto hydrodynamic stagnation-point flow towards permeable stretching surface. To analyze the new analytic solution of this problem, we apply homotopy analysis method. By this powerful and newly developed technique, the convergence series solutions are obtained. To validate the analytic solutions of velocity distribution and temperature distribution using HAM method, we have computed the convergence regions. The HAM solutions have an excellent agreement. A novel result of this problem is that the temperature increases with the increase in Hartmann Number Hat a certain distance from the stretching surface in the presence of suction parameter. Heat flows from the surface to the fluid at near stagnation point on the surface and on the hand far from the stagnation-point, heat flows from the fluid to surface due to combining effect of ohmic dissipation and strain energy inside the boundary layer.

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