# Thermal Effect on Vibration of Parallelogram Plate of Bi-Direction Linearly Varying Thickness 

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Received September 13, 2010; revised November 3, 2010; accepted November 6, 2010


#### Abstract

In this paper, the effect of thermal gradient on the vibration of parallelogram plate with linearly varying thickness in both direction having clamped boundary conditions on all the four edges is analyzed. Thermal effect on vibration of such plate has been taken as one-dimensional distribution in linear form only. An approximate but quiet convenient frequency equation is derived using Rayleigh-Ritz technique with a two-term deflection function. The frequencies corresponding to the first two modes of vibration of a clamped parallelogram plate have been computed for different values of aspect ratio, thermal gradient, taper constants and skew angle. The results have been presented in tabular forms. The results obtained in this study are reduced to that of unheated parallelogram plates of uniform thickness and have generally been compared with the published one.


Keywords: Parallelogram Plate, Vibration, Thermal Gradient, Linearly Thickness, Both Directions

## 1. Introduction

Parallelogram plates have quite a good number of applications in modern structures. This type of plates can be found frequently in modern constructions in the form of reinforced slabs or stiffened plates. Such structures are widely used as floor in bridges, ship hulls, buildings etc. these plates are also used in the construction of wings, tails and fins of rockets and missiles.

In the modern time, people have started taking lot of interest in effect of temperature on solids, as it has a lot of role in space technology, high-speed atmospheric flights and in nuclear energy applications. In the modern technology, mechanical parts of different machines have to operate high temperature, which effect efficiency. The reason for it is that during heating up period of structures exposed to high intensity heat fluxes, the material properties under go significant vibrations.
Notable contributions [1-10] are available on vibration of skew plate but none of them considered the thermal effect. It is well known [11] that in the presence of a constant thermal gradient, the elastic coefficients of homogeneous material become functions of the space vari-
able. Tomar and Gupta [12-13] have considered the effect of thermal gradient of the frequency of an orthotropic plate of variable thickness. Bhatnagar and Gupta [14-15] have studied the effect of thermal gradient on vibration of visco-elastic plate of variable thickness. Singh and Saxena [16] have studied the transverse vibration of skew plates with variable thickness. Gupta and Khanna [17] have studied the vibration of visco-elastic rectangular plate with linearly thickness variation in both directions. Gupta, Kumar and Gupta [18] have analyzed on vibration of visco-elastic parallelogram plate with parabolic thickness variation. Free vibration of super elliptical plates with constant and variable thickness by Ritz method has been discussed by Bambill, Maize and Rossi [19]. Effect of thermal gradient on vibration of non-homogeneous visco-elastic elliptical plate of variable thickness has been analyzed by Gupta and Kumar [20]. Li [21] discussed the vibration analysis of rectangular plate with general elastic boundary supports. Sakiyama, Haung, Matuda and Morita [22] solved the problem of free vibration of orthotropic square plate with square hole.

The aim of the present study is to find the effect of linear thermal gradient on vibration of a clamped paralle-
logram plate with linearly varying thickness in both directions. All the edges are taken as clamped. The Ray-leigh-Ritz technique has been used to determine the frequency equation of the plate. The frequency to the first two modes of vibration is obtained for a clamped parallelogram plate for various values of aspect ratio $(a / b)$, thermal gradient ( $\alpha$ ), taper constants ( $\beta_{1}, \beta_{2}$ ) and skew angle $(\theta)$.

## 2. Analysis and Equation of Motion

A parallelogram plate R (axb) with skew angle $\theta$ is shown in Figure 1. The skew plate is assumed to be nonuniform, thin and isotropic. The skew co-ordinate are related as

$$
\begin{equation*}
\xi=x-y \tan \theta \text { and } \eta=y \sec \theta \tag{1}
\end{equation*}
$$

The boundaries of the plate in oblique co-ordinate are

$$
\begin{equation*}
\xi=0, \xi=a \text { and } \eta=0, \eta=b \tag{2}
\end{equation*}
$$

For free vibration of the parallelogram plate, the displacement is assumed to be of the form

$$
\begin{equation*}
w(\xi, \eta, t)=W(\xi, \eta) \sin \omega t \tag{3}
\end{equation*}
$$

where $W(\xi, \eta)$ is the maximum displacement at time $t$ and $\omega$ is the angular frequency. The maximum kinetic energy, T , and the strain energy, V in the plate when it is executing transverse vibration mode shape $W(\xi, \eta)$ are [1]

$$
\begin{equation*}
T=\frac{1}{2} \rho \omega^{2} \cos \theta \iint h W^{2} d \xi d \eta \tag{4}
\end{equation*}
$$

and

$$
\begin{align*}
V= & \frac{1}{2 \cos ^{3} \theta} \iint D\left[W_{, \xi \xi^{2}}-4 \sin \theta W_{, \xi \xi} W_{, \xi \eta}\right. \\
& +2\left(\sin ^{2} \theta+v \cos ^{2} \theta\right) W_{, \xi \xi} W_{, \eta \eta}  \tag{5}\\
& +2\left(1+\sin ^{2} \theta-v \cos ^{2} \theta\right) W_{, \xi \eta^{2}} \\
& \left.-4 \sin \theta W_{, \xi \eta} W_{, \eta \eta}+W_{, \eta \eta^{2}}\right] d \xi d \eta
\end{align*}
$$

A comma followed by a suffix denotes partial differential with respect to that variable. Here D is the flexural rigidity.

Assume that plate is subjected to a study one dimensional temperature along the length i.e. in $\xi$-direction as


Figure 1. Plate in $x-y$ plane.

$$
\begin{equation*}
\tau=\tau_{0}\left(1-\frac{\xi}{a}\right) \tag{6}
\end{equation*}
$$

where $\tau$ is the temperature excess above the reference temperature at any point at a distance $\frac{\xi}{a}$ and $\tau_{0}$ at $\xi=\mathrm{a}$.

The temperature dependence of the modules of elasticity is given by

$$
\begin{equation*}
E(\tau)=E_{0}(1-\gamma \tau) \tag{7}
\end{equation*}
$$

where $\mathrm{E}_{0}$ is the Young Modules at $\tau=0$.
Using (6) and (7) one obtains

$$
\begin{equation*}
E(\xi)=E_{0}\left(1-\alpha\left(1-\frac{\xi}{a}\right)\right) \tag{8}
\end{equation*}
$$

where $\alpha=\gamma \tau_{0} \quad(0 \leq \alpha<1)$, is a parameter known as temperature gradient.

The thickness variation of the parallelogram plate is assumed to be linear in both directions

$$
\begin{equation*}
h=h_{0}\left(1+\beta_{1}\left(\frac{\xi}{a}\right)\right)\left(1+\beta_{2}\left(\frac{\eta}{b}\right)\right) \tag{9}
\end{equation*}
$$

where $\beta_{1}$ and $\beta_{2}$ are taper constants in $\xi$-direction and $\eta$-direction respectively. And $\mathrm{h}_{0}=\mathrm{h}$ when $\xi, \eta,=0$.

Using(8) and (9) in Equations (4) and (5) one gets

$$
\begin{equation*}
T=\frac{1}{2} h_{0} \rho \omega^{2} \int_{0}^{b} \int_{0}^{a}\left(1+\beta_{1}\left(\frac{\xi}{a}\right)\right)\left(1+\beta_{2}\left(\frac{\eta}{b}\right)\right) W^{2} d \xi d \eta \tag{10}
\end{equation*}
$$

and

$$
\begin{align*}
V= & \frac{E_{0} h_{0}^{3}}{24\left(1-v^{2}\right) \cos ^{4} \theta} \int_{0}^{b} \int_{0}^{a}\left(1-\alpha\left(1-\frac{\xi}{a}\right)\right)\left(1+\beta_{1}\left(\frac{\xi}{a}\right)\right)^{3}\left(1+\beta_{2}\left(\frac{\eta}{b}\right)\right)^{3} \\
& \times\left[W_{, \xi \xi^{2}}-4\left(\frac{a}{b}\right) \sin \theta W_{, \xi \xi} W_{, \xi \eta}+2\left(\frac{a}{b}\right)^{2}\left(\sin ^{2} \theta+\cos ^{2} \theta\right) W_{, \xi \xi} W_{, \eta \eta}\right.  \tag{11}\\
& \left.+2\left(\frac{a}{b}\right)^{2}\left(1+\sin ^{2} \theta-v \cos ^{2} \theta\right) W_{, \xi \eta^{2}}-4\left(\frac{a}{b}\right)^{3} \sin \theta W_{, \xi \eta} W_{, \eta \eta}+\left(\frac{a}{b}\right)^{4} W_{, \eta \eta^{2}}\right] d \xi d \eta
\end{align*}
$$

## 3. Solution and Frequency Equation

In using the Rayleigh-Ritz technique, one requires that the maximum strain energy must be equal to the maximum kinetic energy. It is, therefore, necessary for the problem under consideration that

$$
\begin{equation*}
\delta(V-T)=0 \tag{12}
\end{equation*}
$$

for arbitrary variations of W satisfying relevant geometric boundary conditions.

For a parallelogram plate clamped along all the four edges the boundary conditions are $\mathrm{W}=\mathrm{W}_{, \xi}=0$ at $\xi=0$, a and

$$
\begin{equation*}
W=W_{, \eta}=0 \text { at } \eta=0, b \tag{13}
\end{equation*}
$$

and corresponding two term deflection function is taken as [1].

$$
\begin{align*}
W(\xi, \eta)= & \left(\frac{\xi^{2}}{a^{2}}\right)\left(\frac{\eta^{2}}{b^{2}}\right)\left(1-\frac{\xi}{a}\right)^{2}\left(1-\frac{\eta}{b}\right)^{2}  \tag{14}\\
& \times\left[A_{1}+A_{2}\left(\frac{\xi}{a}\right)\left(\frac{\eta}{b}\right)\left(1-\frac{\xi}{a}\right)\left(1-\frac{\eta}{b}\right)\right] .
\end{align*}
$$

Now Equation (12) becomes after using Equation (14)

$$
\begin{equation*}
\delta\left(V_{1}-\lambda^{2} T_{1}\right)=0 \tag{15}
\end{equation*}
$$

where

$$
\begin{aligned}
V_{1}= & \frac{1}{\cos ^{4} \theta} \int_{0}^{b} \int_{0}^{a}\left(1-\alpha\left(1-\frac{\xi}{a}\right)\right)\left(1+\beta_{1} \frac{\xi}{a}\right)^{3}\left(1+\beta_{2} \frac{\eta}{b}\right)^{3} \\
& \times\left[W_{, \xi \xi^{2}}-4\left(\frac{a}{b}\right) \sin \theta W_{, \xi \xi} W_{, \xi \eta}\right. \\
& +2\left(\frac{a}{b}\right)^{2}\left(\sin ^{2} \theta+\cos ^{2} \theta\right) W_{, \xi \xi} W_{, \eta \eta} \\
& +2\left(\frac{a}{b}\right)^{2}\left(1+\sin ^{2} \theta-v \cos ^{2} \theta\right) W_{, \xi \eta^{2}} \\
& \left.-4\left(\frac{a}{b}\right)^{3} \sin \theta W_{, \xi \eta} W_{, \eta \eta}+\left(\frac{a}{b}\right)^{4} W_{, \eta \eta^{2}}\right] d \xi d \eta \\
& T_{1}=\int_{0}^{b} \int_{0}^{a}\left(1+\beta_{1}\left(\frac{\xi}{a}\right)\right)\left(1+\beta_{2}\left(\frac{\eta}{b}\right)\right) W^{2} d \xi d \eta
\end{aligned}
$$

and

$$
\lambda^{2}=\frac{12 a^{4} \omega^{2} \rho\left(1-v^{2}\right)}{E_{0} h_{0}^{2}} \text { is a frequency parameter. }
$$

Equation (15) involves the unknown $\mathrm{A}_{1}$ and $\mathrm{A}_{2}$ arising due to the substitution of $\mathrm{W}(\xi, \eta)$ from Equation (14). These unknowns are to be determined from Equation (15) for which

$$
\begin{equation*}
\frac{\partial}{\partial A_{n}}\left(V_{1}-\lambda^{2} T_{1}\right)=0, n=1,2 \tag{16}
\end{equation*}
$$

The above equation simplifies to

$$
\begin{equation*}
b_{n 1} A_{1}+b_{n 2} A_{2}=0, n=1,2 \tag{17}
\end{equation*}
$$

where $b_{n 1}, b_{n 2}(n=1,2)$ involve parametric constants and the frequency parameter.

For a non-trivial solution the determinant of the coefficients of Equation (17) must be zero.

Therefore one gets the frequency equation as

$$
\left|\begin{array}{ll}
b_{11} & b_{12}  \tag{18}\\
b_{21} & b_{22}
\end{array}\right|=0
$$

From Equation (18), one can obtained the quadratic equation in $\lambda^{2}$ from which two values of $\lambda^{2}$ can found.

## 4. Result and Discussion

The frequency Equation (18) is quadratic in $\lambda^{2}$ from which two roots can be determined. The frequency parameter $\lambda$ corresponding to the first two modes of vibration of clamped parallelogram plate have been computed for various values of temperature gradient $(\alpha)$, aspect ratio $(\mathrm{a} / \mathrm{b})$, taper constants $\left(\beta_{I}\right),\left(\beta_{2}\right)$ and skew angle $(\theta)$. These results are summarized in Tables 1-5.

For numerical computation the value of Poisson's ratio $v$ is taken 0.3.

Table 1 contains the value of frequency parameter of a clamped parallelogram plate for different values of thermal constant $(\alpha)$ and for fixed aspect ratio $(a / b)=1.0$ for the first two modes of vibration for values of taper constants ( $\beta_{1}=0.0$ and $\beta_{2}=0.0, \beta_{1}=0.0$ and $\beta_{2}=0.6, \beta_{1}=$ 0.4 and $\beta_{2}=0.6$ ) and two values of skew angle $(\theta)$. It can be seen from the table that as thermal constant increase, frequency parameter decreases in all the cases. Also effect of thermal constant is more for $\theta=45^{\circ}$ in comparison to $\theta=30^{\circ}$. Table 2 comprises the value of frequency parameter of a clamped parallelogram plate for different values of aspect ratio $(a / b)$ for the first two modes of vibration for fixed value of taper constant ( $\beta_{1}=0.4$ and $\beta_{2}$ $=0.6)$ and fixed thermal constant $(\alpha=0.2)$ for two values of skew angle $(\theta)$. It is noted from the table that as aspect ratio is increasing, frequency parameter increasing in all the cases and effect of aspect ratio is more for $\theta=45^{\circ}$ in comparison to $\theta=30^{\circ}$.

Table 3 gives the values of frequency parameter of a clamped parallelogram plate for various values of taper constant $\left(\beta_{I}\right)$ for the first two modes of vibration for fixed value of taper constant $\left(\beta_{2}=0.0,0.6\right)$, thermal gradient $(\alpha=0.4)$, aspect ratio $(\mathrm{a} / \mathrm{b}=1.0)$ and for two values of skew angle $(\theta)$. It is observed from the table that as taper constant is increasing, frequency parameter is also increasing.

Table 4 shows the values of frequency parameter of a clamped parallelogram plate for various values of taper

Table 1. Frequency parameter $\lambda$ of a clamped parallelogram plate for $\mathbf{a} / \mathbf{b}=\mathbf{1 . 0}$.

| $\theta=30^{\circ}$ |  |  |  |  |  |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $\alpha$ | $\beta_{1}=\beta_{2}=0.0$ |  |  |  | $\beta_{1}=0.0, \beta_{2}=0.6$ |  | $\beta_{1}=0.4, \beta_{2}=0.6$ |  |
|  |  | I mode |  | II mode | I mode | II mode | I mode | II mode |
| 0.0 |  | 61.4283 |  | 236.5564 | 67.2606 | 258.9453 | 81.1022 | 318.4257 |
| 0.2 |  | 58.7339 |  | 226.1332 | 63.7993 | 245.2736 | 77.4094 | 304.5002 |
| 0.4 |  | 55.9060 |  | 215.2066 | 59.9905 | 230.7934 | 73.5239 | 289.9084 |
| 0.6 |  | 52.9215 |  | 203.6964 | 56.0027 | 215.3417 | 69.4106 | 274.5449 |
| 0.8 |  | 49.7499 |  | 191.4977 | 51.7080 | 198.6919 | 65.0218 | 258.2730 |
| $\theta=45^{\circ}$ |  |  |  |  |  |  |  |  |
| $\beta_{1}=\beta_{2}=0.0$ |  |  |  |  | $\beta_{1}=0.0, \beta_{2}=0.6$ |  | $\beta_{1}=0.4, \beta_{2}=0.6$ |  |
|  | I mode |  | II mode |  | I mode | II mode | I mode | II mode |
|  | 79.5005 |  | 302.5225 |  | 105.7948 | 401.7948 | 127.3479 | 499.3421 |
|  | 75.4208 |  | 286.9980 |  | 100.1417 | 380.3892 | 121.5072 | 477.7926 |
|  | 71.1074 |  | 270.5844 |  | 94.2215 | 357.7051 | 115.3581 | 455.2276 |
|  | 66.5149 |  | 253.1086 |  | 87.9034 | 333.4815 | 108.8434 | 431.4886 |
|  | 61.5808 |  | 234.3329 |  | 81.0943 | 307.3547 | 101.8854 | 406.3723 |

Table 2. Frequency parameter $\lambda$ of a clamped parallelogram plate for $\boldsymbol{\beta}_{1}=0.4, \boldsymbol{\beta}_{2}=0.6, \alpha=0.2$.

|  | $\theta=30^{\circ}$ |  |  | $\theta=45^{\circ}$ |
| :---: | :---: | :---: | :---: | :---: |
| $\mathrm{a} / \mathrm{b}$ | I mode | II mode | I mode | II mode |
| 0.5 | 51.8014 | 206.8539 | 121.7636 | 316.5569 |
| 1.0 | 77.4094 | 304.5002 | 121.5072 | 477.7926 |
| 1.5 | 129.6335 | 514.0130 | 201.7746 | 804.2516 |
| 2.0 | 207.3843 | 827.5726 | 319.2430 | 1283.8269 |
| 2.5 | 309.2726 | 1237.4373 | 472.4142 | 1906.5696 |

Table 3. Frequency parameter $\lambda$ of a clamped parallelogram plate for $\mathbf{a} / \mathbf{b}=1$.

| $\beta_{1}$ | $\theta=30^{\circ}$ |  |  |  | $\theta=45^{\circ}$ |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | $\alpha=0.4, \beta_{2}=0.0$ |  | $\alpha=0.4, \beta_{2}=0.6$ |  | $\alpha=0.4, \beta_{2}=0.0$ |  | $\alpha=0.4, \beta_{2}=0.6$ |  |
|  | I mode | II mode | I mode | II mode | I mode | II mode | I mode | II mode |
| 0.0 | 45.2395 | 174.2904 | 59.9905 | 230.7934 | 71.1074 | 270.5843 | 94.2216 | 357.7051 |
| 0.2 | 50.4341 | 194.2100 | 66.6139 | 260.3359 | 79.2771 | 301.4143 | 104.5958 | 407.2275 |
| 0.4 | 55.9060 | 215.2067 | 73.5239 | 289.9084 | 87.8845 | 333.8754 | 115.3581 | 455.2276 |
| 0.6 | 61.5759 | 237.0058 | 80.6690 | 319.7503 | 96.8045 | 367.5542 | 126.4684 | 502.6380 |
| 0.8 | 67.3896 | 259.413 | 88.0023 | 349.9568 | 105.9515 | 402.1589 | 137.8679 | 549.9366 |

Table 4. Frequency parameter $\lambda$ of a clamped parallelogram plate for $\mathbf{a} / \mathbf{b}=1$.

| $\beta_{2}$ | $\theta=30^{\circ}$ |  |  |  | $\theta=45^{\circ}$ |  |  |  |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | $\alpha=0.4, \beta_{l}=0.0$ |  | $\alpha=0.4, \beta_{l}=0.6$ |  | $\alpha=0.4, \beta_{l}=0.0$ |  | $\alpha=0.4, \beta_{l}=0.6$ |  |
|  | I mode | II mode | I mode | II mode | I mode | II mode | I mode | II mode |
| 0.0 | 45.2395 | 174.5793 | 61.5759 | 237.0058 | 71.1074 | 270.5005 | 96.8045 | 367.5542 |
| 0.2 | 49.8864 | 192.1109 | 67.6357 | 266.7037 | 78.3851 | 298.0087 | 106.2491 | 416.8518 |
| 0.4 | 54.8293 | 211.0362 | 74.0234 | 293.7256 | 86.1307 | 327.2656 | 116.1637 | 460.8392 |
| 0.6 | 59.9405 | 230.7934 | 80.6690 | 319.7503 | 94.2216 | 357.7051 | 126.4684 | 502.6380 |
| 0.8 | 65.3156 | 251.1870 | 87.5153 | 345.6646 | 102.5688 | 389.1129 | 137.0843 | 543.8896 |

constant $\left(\beta_{2}\right)$ for the first two modes of vibration for fixed value of taper constant $\left(\beta_{I}=0.0,0.6\right)$, thermal gradient $(\alpha=0.4)$, aspect ratio $(\mathrm{a} / \mathrm{b}=1.0)$ and for two values of skew angle $(\theta)$. It is observed from the table that as taper constant is increasing, frequency parameter is also increasing.

Table 5 contains the values of frequency parameter of a clamped parallelogram plate for various values of skew angle $(\theta)$ for the first two modes of vibration for fixed value of taper constant ( $\beta_{1}=0.0, \beta_{2}=0.0$ and $\beta_{1}=0.4 \beta_{2}$ $=0.6$ ), thermal gradient $(\alpha=0.2)$ and aspect ratio $(\mathrm{a} / \mathrm{b}=$ 1.0). It is observed from the table that as skew angle is increasing, frequency parameter is also increasing.

## 5. Conclusions

The results for a uniform isotropic clamped rectangular plate are compared with the results published by the authors [1] and found to be in close agreement.

After comparing, the authors conclude that as the skew angle increases, the frequency parameter increases. Further, it is interesting to note that effect of taper in $\xi$ -direction is small in comparisons of $\eta$-direction. Therefore, engineers are provided with a method to develop plates in a manner so that they can fulfill the requirements.

Table 5. Frequency parameter $\lambda$ of a clamped parallelogram plate for $\mathbf{a} / \mathrm{b}=1$.

| $\theta$ | $\alpha=0.2, \beta_{1}=\beta_{2}=0.0$ |  | $\alpha=0.2, \beta_{1}=0.4 \beta_{2}=0.6$ |  |
| :---: | :---: | :---: | :---: | :---: |
|  | I mode | II mode | I mode | II mode |
| $0^{\circ}$ | 34.1524 | 133.6542 | 55.4962 | 212.4850 |
| $30^{\circ}$ | 47.9838 | 184.8624 | 77.4094 | 304.5002 |
| $45^{\circ}$ | 75.4208 | 286.9980 | 121.5072 | 477.7926 |
| $60^{\circ}$ | 157.4119 | 593.0757 | 253.5286 | 993.6003 |

## 6. References

[1] A. W. Leissa, "Vibration of Plates," NASA SP-160, 1969.
[2] A. W. Leissa, "Recent Studies in Plate Vibration 1981-1985, Part-II Complicating Effect," The Shock and Vibration Digest, Vol. 19, No. 3, 1987, pp. 10-24. doi: 10.1177/058310248701900304
[3] A. W. Leissa, "Recent Studies in Plate Vibration 1982-1985, Part-I Classical Theory," The Shock and Vibration Digest, Vol. 19, 1987, pp. 11-18. doi:10.1177/ 058310248701900204
[4] B. Singh and S. Chakraverty, "Flexural Vibration of Skew Plates Using Boundary Characteristic Orthogonal Polynomials in Two Variables," Journal of Sound and Vibration, Vol. 173, No. 2, 1994, pp. 157-178. doi: 10.1006/jsvi.1994.1224
[5] P. S. Nair and S. Durvasula, "Vibration of Skew Plates," Journal of Sound and Vibration, Vol. 26, 1973, pp. 1-20. doi:10.1016/S0022-460X(73)80201-9
[6] R. M. Orris and M. Petyt, "A Finite Element Study of Vibration of Trapezoidal Plates," Journal of Sound and Vibration, Vol. 27, No. 3, 1973, pp. 325-344. doi: 10.1016/S0022-460X(73)80349-9
[7] P. A. A. Laura, R. H. Gutierrez and R. B. Bhat, "Transverse Vibration of a Trapezoidal Cantilever Plate of Variable Thickness," AIAA Journal, Vol. 27, No. 7, 1989, pp. 921-922. doi:10.2514/3.10201
[8] K. M. Liew and M. K. Lim, "Transverse Vibration of Trapezoidal Plates of Variable Thickness: Symmetric Trapezoids," Journal of Sound and Vibration, Vol. 165, No. 1, 1993, pp. 45-67. doi:10.1006/jsvi.1993.1242
[9] K. Y. Lam, K. M. Liew and S. T. Chow, "Free Vibration Analysis of Isotropic and Orthotropic Triangular Plates," International Journal of Mechanical Sciences, Vol. 32, No. 5, 1990, pp. 455-464. doi:10.1016/0020-7403(90) 90172-F
[10] T. Sakiyama and M. Huang, "Free Vibration Analysis of Right Triangular Plates with Variable Thickness," Journal of Sound and Vibration, Vol. 234, No. 5, 2000, pp. 841-858. doi:10.1006/jsvi.2000.2903
[11] N. J. Hoff, "High Temperature Effect in Aircraft Struc-
tures," Pergamon Press, New York, p. 62.
[12] J. S. Tomar and A. K. Gupta "Thermal Effect of Frequencies of an Orthotropic Rectangular Plate of Linearly Varying Thickness," Journal of Sound and Vibration, Vol. 90, No. 3, 1983, pp. 325-331. doi:10.1016/0022-460X(83) 90715-0
[13] J. S. Tomar and A. K. Gupta, "Thermal Effect of Axis Symmetric Vibration of an Orthotropic Circular Plate of Variable Thickness," American Institute of Aeronautics and Astronautics Journal, Vol. 22, No. 7, 1984, pp. 10151017.
[14] N. S. Bhatnagar and A. K. Gupta, "Thermal Effect on Vibration of Visco-Elastic Elliptic Plate of Variable Thickness," Proceedings of International Conference on Modeling and Simulation, Melbourne, 1987, pp. 424-429.
[15] N. S. Bhatnagar and A. K. Gupta, "Vibration Analysis of Visco-Elastic Circular Plate Subjected to Thermal Gradient," Modeling, Simulation and Control, B, AMSE Press 15, Cairo, 1988, pp. 17-31.
[16] B. Singh and V. Saxena, "Transverse Vibration of Skew Plates with Variable Thickness," Journal of Sound and Vibration, Vol. 206. No. 1, 1997, pp. 1-13. doi:10.1006/ jsvi.1997.1032
[17] A. K. Gupta and A. Khanna, "Vibration of Visco-Elastic Rectangular Plate with Linearly Thickness Variation in
both Directions," Journal of Sound and Vibration, Vol. 301, No. 3-5, 2007, pp. 450-457. doi:10.1016/j.jsv. 2006. 01.074
[18] A. K. Gupta, Anuj Kumar and Y.K. Gupta, "Vibration of Visco Elastic Parallelogram Plate with Parabolic Thickness Variation," Applied Mathematics, Vol. 1, No. 2, 2010, pp. 128-136. doi:10.4236/am.2010.12017
[19] D. V. Bambill, S. Maize and R. E. Rossi "Free Vibration of Super Elliptical Plates with Constant and Variable Thickness," Journal of Sound and Vibration, Vol. 329, No. 21, 2010, pp. 4578-4580. doi:10.1016/j.jsv.2010.05. 007
[20] A. K. Gupta and Lalit Kumar "Effect of Thermal Gradient on Vibration of Non-Homogeneous Visco-Elastic Elliptic Plate of Variable Thickness," Meccanica, Vol. 44, No. 5, 2009, pp. 507-518. doi:10.1007/s11012-008-9184-9
[21] W. L. Li, "Vibration Analysis of Rectangular Plate with General Elastic Boundary Supports," Journal of Sound and Vibration, Vol. 273, No. 3, 2004, pp. 619-635. doi: 10.1016/S0022-460X(03)00562-5
[22] T. Sakiyama, M. Haung, H. Matuda and C. Morita, "Free Vibration of Orthotropic Square Plate with a Square Hole," Journal of Sound and Vibration, Vol. 259, No. 1, 2003, pp. 66-80. doi:10.1006/jsvi.2002.5181

