

The Bending of Rectangular Deep Beams with Fixed at Both Ends under Uniform Load

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Abstract

Considering the effects of the beam section rotation, shear deformation of the adjacent section and transverse pressure, derived the new equation of rectangular section deep beams, and gives the basic solution of deep beams [1]. And discussed at the bending problems of deep rectangular beams with fixed at both ends under uniform load, based on the equations given in this paper, application of reciprocal law, doing numerical calculation in Matlab platform, compare with the results of ANSYS finite element analysis [2].

Keywords: The Bending, Rectangular Section, Deep Beams, The Basic Solution, Uniform Load, Fixed at Both Ends, Reciprocal Method

1. The Bending of Rectangular Deep Beams under Uniform Load

1.1. The Derivation of New Equations of the Rectangular Deep Beams

1.1.1. The Derivation of the Transverse Pressure σ_z

The elastic mechanics equations in z direction is

$$\frac{\partial \tau_{xz}}{\partial x} + \frac{\partial \tau_{yz}}{\partial y} + \frac{\partial \sigma_z}{\partial z} = 0 \quad (1)$$

The τ_{yz} of straight beam which is in **Figure 1** is 0, and then the Formula (1) change into

$$\frac{\partial \tau_{xz}}{\partial x} + \frac{\partial \sigma_z}{\partial z} = 0 \quad (2)$$

According to the material mechanics knowledge [3], there is

$$\tau_{xz} = \frac{Q}{2J_y} \left(\frac{h^2}{4} - z^2 \right) \quad (3)$$

For the rectangular cross-section as shown in **Figure 1**, there is

$$J_y = \frac{bh^3}{12} \quad (4)$$

And then

$$\tau_{xz} = \frac{6Q}{bh^3} \left(\frac{h^2}{4} - z^2 \right) \quad (5)$$

Calculating the Formula (2) to get

$$\sigma_z = -\int \frac{\partial \tau_{xz}}{\partial x} dx + C \quad (6)$$

Paying attention to

$$\frac{\partial \tau_{xz}}{\partial x} = \frac{6}{bh^3} \frac{\partial Q}{\partial x} \left(\frac{h^2}{4} - z^2 \right) \quad (7)$$

Due to

$$\frac{\partial Q}{\partial x} = -q \quad (8)$$

q is the load strength of unit length along the x direction. Put Formulae (7) and (8) into (6) to get

$$\sigma_z = \int \frac{6q}{bh^3} \left(\frac{h^2}{4} - z^2 \right) dz + C \quad (9)$$

when

$$z = -\frac{h}{2}$$

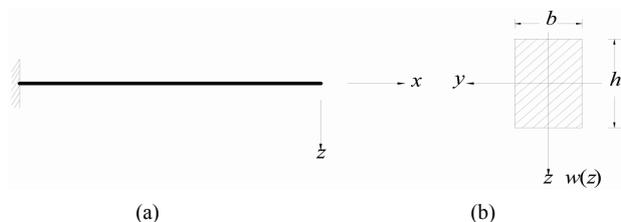


Figure 1. Straight beam of rectangular cross section.

$$\sigma_z = -\frac{q}{b} = \frac{6q}{bh^3} \left(-\frac{1}{3} \times z^3 + \frac{h^2}{4} \cdot z \right) + C \quad (10)$$

Reduction is

$$-\frac{q}{b} = -\frac{q}{2b} + C, C = -\frac{q}{2b} \quad (10)$$

At last, getting:

$$\sigma_z = -\frac{3q}{4b} \left[\frac{2}{3} - 2\frac{z}{h} + \frac{8}{3} \left(\frac{z}{h} \right)^3 \right] \quad (11)$$

After calculation, we can get that when $z = h/2$, $\sigma_z = 0$.

1.1.2. The Derivation of the Moment M_x

Introducing the concept of average corner ω_x . Making the ω_x is the average corner of cross section around x axis, the ω_x is:

$$b \int_{-\frac{h}{2}}^{\frac{h}{2}} \sigma_x u' dz = M_x \omega_x \quad (12)$$

In the formula, u' is the axial displacement of the straight beam. Because

$$\sigma_x = \frac{12M_x z}{bh^3} \quad (13)$$

Put Formula (13) into (12) to get

$$b \int_{-\frac{h}{2}}^{\frac{h}{2}} \frac{12M_x z}{bh^3} u' dz = M_x \omega_x$$

There is

$$\omega_x = \frac{12}{h^3} \int_{-\frac{h}{2}}^{\frac{h}{2}} u' z dz \quad (14)$$

Using the Hooke's law to get

$$\frac{\partial u'}{\partial x} = \frac{1}{E} (\sigma_x - \nu \sigma_z) \quad (15)$$

Put Formula (11) into (15) to get

$$\sigma_x = E \frac{du'}{dx} + \nu \sigma_z = E \frac{du'}{dx} - \frac{3q\nu}{4b} \left[\frac{2}{3} - 2\frac{z}{h} + \frac{8}{3} \left(\frac{z}{h} \right)^3 \right] \quad (16)$$

So the expression of the moment M_x is

$$\begin{aligned} M_x &= b \int_{-\frac{h}{2}}^{\frac{h}{2}} \sigma_x z dz \\ &= b \int_{-\frac{h}{2}}^{\frac{h}{2}} z \left\{ E \frac{du'}{dx} - \frac{3q\nu}{4b} \left[\frac{2}{3} - 2\frac{z}{h} + \frac{8}{3} \left(\frac{z}{h} \right)^3 \right] \right\} dz \\ &= EJ \frac{d\omega_x}{dx} + \int_{-\frac{h}{2}}^{\frac{h}{2}} (-1) \frac{3q\nu}{4} \left[\frac{2}{3} - 2\frac{z}{h} + \frac{8}{3} \left(\frac{z}{h} \right)^3 \right] z dz \\ &= EJ \left(\frac{d\omega_x}{dx} + \frac{6\nu}{5Ehb} q \right) \end{aligned} \quad (17)$$

After this, we establish the relationships between M_x and ω_x

1.1.3. The Derivation of ω_x

According to the shear hooker law [4], there is

$$\frac{\partial u'}{\partial z} + \frac{\partial w'}{\partial x} = \frac{6Q}{Gbh^3} \left(\frac{h^2}{4} - z^2 \right) \quad (18)$$

On both sides of Formula (18) are by $\frac{6}{bh^3} \left(\frac{h^2}{4} - z^2 \right)$,

and doing definite integration between $z = -\frac{h}{2}$ and

$z = \frac{h}{2}$ to get:

$$\int_{-\frac{h}{2}}^{\frac{h}{2}} \frac{\partial u'}{\partial z} \frac{6}{bh^3} \left(\frac{h^2}{4} - z^2 \right) dz + \int_{-\frac{h}{2}}^{\frac{h}{2}} \frac{\partial w'}{\partial x} \frac{6}{bh^3} \left(\frac{h^2}{4} - z^2 \right) dz \quad (19)$$

$$= \frac{Q}{G} \int_{-\frac{h}{2}}^{\frac{h}{2}} \left(\frac{6}{bh^3} \right)^2 \left(\frac{h^2}{4} - z^2 \right)^2 dz$$

Calculating the first part of Formula (19) to get

$$\begin{aligned} &\int_{-\frac{h}{2}}^{\frac{h}{2}} \frac{\partial u'}{\partial z} \frac{6}{bh^3} \left(\frac{h^2}{4} - z^2 \right) dz \\ &= \frac{6}{bh^3} \int_{-\frac{h}{2}}^{\frac{h}{2}} \left\{ \frac{\partial}{\partial z} \left[u' \left(\frac{h^2}{4} - z^2 \right) \right] - u' \frac{\partial}{\partial z} \left(\frac{h^2}{4} - z^2 \right) \right\} dz \\ &= \frac{12}{bh^3} \int_{-\frac{h}{2}}^{\frac{h}{2}} u' z dz \end{aligned}$$

The Formula (14) is paid attention to get

$$\int_{-\frac{h}{2}}^{\frac{h}{2}} \frac{\partial u'}{\partial z} \frac{6}{bh^3} \left(\frac{h^2}{4} - z^2 \right) dz = \omega_x \frac{1}{b} \quad (20)$$

Introducing the concept of average deflection w . w is the average deflection of various of points along the height of straight beam [5]. The w is

$$w \cdot Q = \int_{-\frac{h}{2}}^{\frac{h}{2}} \frac{6Q}{bh^3} \left(\frac{h^2}{4} - z^2 \right) b w' dz$$

Calculating to get

$$w = \frac{3}{2h} \int_{-\frac{h}{2}}^{\frac{h}{2}} w' \left[1 - 4 \left(\frac{z}{h} \right)^2 \right] dz \quad (21)$$

And then

$$\frac{\partial w}{\partial x} = \int_{-\frac{h}{2}}^{\frac{h}{2}} \frac{\partial w'}{\partial x} \frac{6}{h^3} \left(\frac{h^2}{4} - z^2 \right) dz \quad (22)$$

Putting Formulae (20) and (22) into (19) to get

$$\omega_x \frac{1}{b} + \frac{\partial w}{\partial x} \frac{1}{b} = \frac{Q}{G} \int_{-\frac{h}{2}}^{\frac{h}{2}} \left(\frac{6}{bh^3} \right)^2 \left(\frac{h^2}{4} - z^2 \right)^2 dz \quad (23)$$

Given $G = \frac{E}{2(1+\nu)}$, after calculating to get

$$\omega_x = -\frac{\partial w}{\partial x} + \frac{12}{5} \cdot \frac{1+\nu}{Ehb} Q_x \tag{24}$$

After this, we establish the relationships between ω_x , w and Q_x .

1.1.4. The Establishment of Equilibrium Differential Equation

According to the material mechanics knowledge, there is

$$\frac{dQ_x}{dx} = -q \tag{25}$$

$$\frac{dM_x}{dx} = Q \tag{26}$$

$$\frac{d^2 M_x}{dx^2} + q = 0 \tag{27}$$

Putting Formula (25) into (28) to get

$$M_x = -EJ \frac{d^2 w}{dx^2} + (1+\nu) \frac{h^2}{5} \frac{dQ_x}{dx} + \frac{h^2}{10} \nu q \tag{28}$$

Putting Formula (25) into (28) to get

$$M_x = -EJ \frac{d^2 w}{dx^2} - \frac{qh^2}{10} (2+\nu) \tag{29}$$

After this, we establish the relationships between M_x and w .

Putting Formula (26) into (29) to get

$$Q_x = -EJ \frac{d^3 w}{dx^3} - \frac{h^2}{10} (2+\nu) \frac{dq}{dx} \tag{30}$$

After this, we establish the relationships between Q_x and w .

Putting Formula (30) into (24) to get

$$\begin{aligned} \omega_x &= -\frac{dw}{dx} + \frac{12}{5} \cdot \frac{1+\nu}{Ehb} Q_x \\ &= -\frac{dw}{dx} - \frac{12J}{5hb} (1+\nu) \frac{d^3 w}{dx^3} - \frac{6}{25} \cdot \frac{h}{Eb} (1+\nu) (2+\nu) \frac{dq}{dx} \end{aligned} \tag{31}$$

After this, we establish the relationships between ω_x and w .

Putting Formula (27) into (29) to get

$$EJ \frac{d^4 w}{dx^4} = q - \frac{h^2}{10} (2+\nu) \frac{d^2 q}{dx^2} \tag{32}$$

The Formula (32) is the equilibrium differential equation of the deep Beams under uniform load.

After that, the Formula (32) should be analyzed. Assuming that

$$q = q_1 + q_2 + q_3 \tag{33}$$

In the Formula (33), q_1 is the uniform load and q_2 is the concentrated load of the point of η which can be expressed as

$$q_2 = P\delta(x-\eta) \tag{34}$$

In the Formula (34), $\delta(x-\eta)$ is the one-dimensional Delta function of the point of η and q_3 is the concentrated couples of the point of η which can be expressed as

$$q_3 = M\delta'(x-\eta) \tag{35}$$

The control equation of the deep beams under uniform load q , concentrated load P and concentrated couples M is

$$\begin{aligned} EJ \frac{d^4 w}{dx^4} &= q - \frac{h^2}{10} (2+\nu) \frac{d^2 q}{dx^2} + P\delta(x-\eta) \\ &\quad - \frac{Ph^2}{10} (2+\nu) \frac{d^2 \delta(x-\eta)}{dx^2} + M\delta'(x-\eta) \\ &\quad - \frac{Mh^2}{10} (2+\nu) \frac{d^2 \delta'(x-\eta)}{dx^2} \end{aligned}$$

1.2. The Basic Solution of the Bending of Rectangular Deep Beams

Considering the boundary conditions of the deep rectangular beam as shown in **Figure 2**.

The deep rectangular beams as shown in **Figure 3** with both ends simply supported under the one-dimensional Delta function $\delta(x-\eta)$ is considered as the basic system. The solution of the basic system is the basic solution of

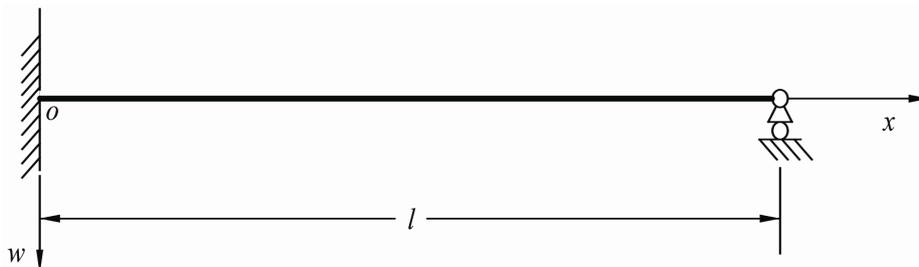


Figure 2. Rectangular beam with different conditions.

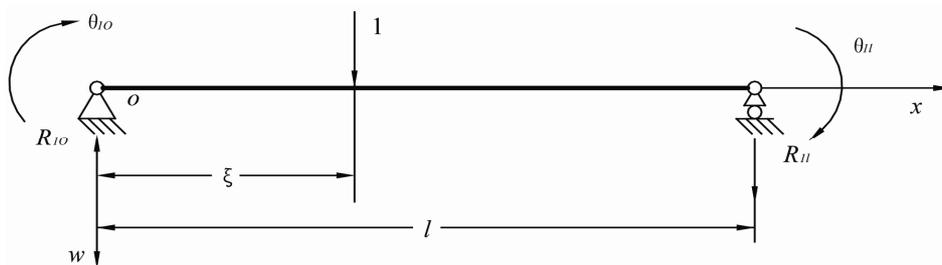


Figure 3. Fictitiously basic system for deep beams of the rectangular cross section.

deep beams. Theoretically, any straight beam under transverse uniform load q can be considered as the basic system, the solution of which can be considered as the actual system. However, the deep rectangular beams with both ends simply supported under transverse uniform load should be chosen as the basic system, for the solution of which is simple.

The control equation of the deflection functions is

$$EJ \frac{d^4 w}{dx^4} = \delta(x - \xi) \tag{37}$$

In the Formula (37), $\delta(x - \eta)$ is the one-dimensional Delta function of the point of $(\xi, 0)$ which is

$$\delta(x - \xi) = \begin{cases} 0, & x \neq \xi; \\ \infty, & x = \xi \end{cases}$$

For any a, b which satisfy the condition of $a < \xi < b$, $\delta(x - \eta)$ has the following properties

- 1) $\int_a^b \delta(x - \xi) dx = 1$
- 2) $\int_a^b f(x) \delta(x - \xi) dx = f(\xi)$
- 3) $\int_a^b f(x) \delta^{(n)}(x - \xi) dx = (-1)^n f^{(n)}(\xi)$

For the straight slender beam which is not considered the shearing deformation, in the Formula (37), $\delta(x - \eta)$ is the transverse unit concentrated load of the point of $(\xi, 0)$. However, for deep beams under the bending, $\delta(x - \eta)$ is the one-dimensional Delta function of the point of $(\xi, 0)$ which dose not have No mechanical significance is called unit concentrated load. And then the rectangular deep beam in **Figure 3** is the basic system, the solution of which is the basic solution.

For the research, the function of δ and its derivative of Fourier coefficient are provided in **Table 1**.

If w_1 will be taken as single heavy trigonometric series, means that $w_1 = \sum_{m=1,2}^{\infty} A_m \sin \frac{m\pi x}{l}$. And then $\delta(x - \eta)$ will be spread out into a sine trigonometric series

$$\delta(x - \xi) = \frac{2}{l} \sum_{m=1,2}^{\infty} \sin \frac{m\pi \xi}{l} \sin \frac{m\pi x}{l} \tag{38}$$

Table 1. δ function and the fourier coefficients of its derivative with different orders.

f(x)	sine series		cosine series	
	$b_m (m \geq 1)$	a_0	$a_m (m \geq 1)$	
$\delta(x - \xi)$	$(2/l) \sin(m\pi\xi/l)$	$2/l$	$(2/l) \cos(m\pi\xi/l)$	
$\delta'(x - \xi)$	$-(2m\pi/l^2) \cos(m\pi\xi/l)$	0	$(2m\pi/l^2) \sin(m\pi\xi/l)$	
$\delta''(x - \xi)$	$-(2m^2\pi^2/l^3) \sin(m\pi\xi/l)$	0	$-(2m^2\pi^2/l^3) \cos(m\pi\xi/l)$	
$\delta'''(x - \xi)$	$(2m^3\pi^3/l^4) \cos(m\pi\xi/l)$	0	$-(2m^3\pi^3/l^4) \sin(m\pi\xi/l)$	

Putting Formula (38) into (37) to get

$$A_m = \frac{2l^3}{(m\pi)^4 EJ} \sin \frac{m\pi \xi}{l} \tag{39}$$

The basic solution can be got easily

$$w_1(x, \xi) = \sum_{m=1,2}^{\infty} \frac{2l^3}{(m\pi)^4 EJ} \sin \frac{m\pi \xi}{l} \sin \frac{m\pi x}{l} \tag{40}$$

For calculating the flexible crankshaft equation of the actual system, boundary corner expression which is in the form of sine series of the actual system is provided as follow

$$\theta_{l0} = \left[\frac{dw_1(x, \xi)}{dx} \right]_{x=0} = \sum_{m=1,2}^{\infty} \frac{2l^2}{(m\pi)^3 EJ} \sin \frac{m\pi \xi}{l} \tag{41}$$

$$\theta_{ll} = \left[\frac{dw_1(x, \xi)}{dx} \right]_{x=l} = \sum_{m=1,2}^{\infty} \frac{(-1)^m 2l^2}{(m\pi)^3 EJ} \sin \frac{m\pi \xi}{l} \tag{42}$$

And boundary corner expression which is in the form of polynomial of the actual system is provided as follow

$$\theta_{l0} = \left[\frac{dw_1(x, l - \xi)}{dx} \right]_{x=0} = \frac{1}{6EJl} (l - \xi) \xi (2l - \xi) \tag{43}$$

$$\theta_{ll} = \left[\frac{dw_1(l - x, \xi)}{dx} \right]_{x=l} = -\frac{1}{6EJl} (l - \xi) \xi (l + \xi) \tag{44}$$

2. The Bending of Rectangular Deep Beams with Both Ends Simply Supported, Fixed at Both Ends under Uniform Load

2.1. The Bending of Rectangular Deep Beams with Both Ends Simply Supported

The actual system of rectangular deep beams with both ends simply supported, fixed at both ends under uniform load as shown in **Figure 4**.

The corresponding control equation is

$$EJ \frac{d^4 w}{dx^4} = q - \frac{h^2}{10}(2+\nu) \frac{d^2 q}{dx^2} \quad (45)$$

The basic system as shown in **Figure 3**, taking the left boundary corner with polynomial form as follow

$$\theta_{l0} = \left[\frac{dw_1(x, l-\xi)}{dx} \right]_{x=0} = \frac{1}{6EJ} (l-\xi)\xi(2l-\xi) \quad (46)$$

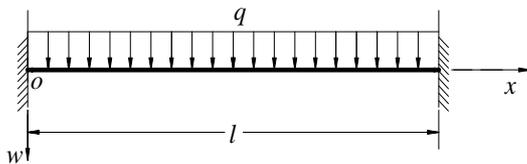
Taking the right boundary corner with polynomial form as follow

$$\theta_{l1} = \left[\frac{dw_1(l-x, \xi)}{dx} \right]_{x=l} = -\frac{1}{6EJ} (l-\xi)\xi(l+\xi) \quad (47)$$

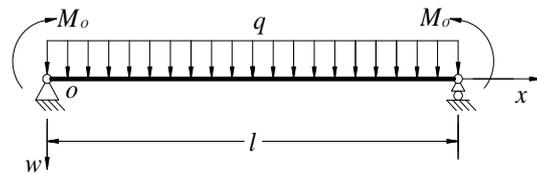
Considering the reciprocal method between the basic system and the actual system (as shown in **Figure 4**), to get the buckling line equation of the actual system

$$\begin{aligned} w(\xi) &= \int_0^l \left[q - \frac{h^2}{10}(2+\nu) \frac{d^2 q}{dx^2} \right] w_1(x, \xi) dx \\ &\quad + M_0 \theta_{l0} - M_0 \theta_{l1} \\ &= \sum_{m=1,3,5}^{\infty} \frac{4ql^4}{m^3 \pi^5 EJ} \left[\frac{1}{m^2} + \frac{1}{10}(2+\nu) \left(\frac{h}{l} \right)^2 \pi^2 \right] \\ &\quad \times \sin \frac{m\pi\xi}{l} + M_0 \frac{1}{2EJ} (l\xi - \xi^2) \end{aligned} \quad (48)$$

$w(\xi)$ is taken a derivatives for ξ to get



(a)



(b)

Figure 4. Actual system of deep beams of the rectangular cross section with two edges fixed under uniformly distributed load.

$$M_0 = -\frac{2}{l} \sum_{m=1,3,5}^{\infty} \left\{ \frac{4ql}{\pi^2} \left[\left(\frac{l}{m\pi} \right)^2 - \frac{12J}{5bh}(1+\nu) \right] \left[\frac{1}{m^2} + \frac{1}{10}(2+\nu) \left(\frac{h}{l} \right)^2 \pi^2 \right] + \frac{24}{25}(1+\nu)(2+\nu) \frac{qhJ}{bl} \right\} \quad (54)$$

$$\begin{aligned} \frac{dw(\xi)}{d\xi} &= \sum_{m=1,3,5}^{\infty} \frac{4ql^3}{m^2 \pi^4 EJ} \left[\frac{1}{m^2} + \frac{1}{10}(2+\nu) \left(\frac{h}{l} \right)^2 \pi^2 \right] \\ &\quad \times \cos \frac{m\pi\xi}{l} + M_0 \frac{1}{2EJ} (l-2\xi) \end{aligned} \quad (49)$$

$w(\xi)$ is taken three derivatives for ξ to get

$$\begin{aligned} \frac{d^3 w(\xi)}{d\xi^3} &= -\sum_{m=1,3,5}^{\infty} \frac{4ql}{\pi^2 EJ} \left[\frac{1}{m^2} + \frac{1}{10}(2+\nu) \left(\frac{h}{l} \right)^2 \pi^2 \right] \\ &\quad \times \cos \frac{m\pi\xi}{l} \end{aligned} \quad (50)$$

$$\frac{dq(\xi)}{d\xi} = \sum_{m=1,3,5}^{\infty} \frac{4q}{l} \cos \frac{m\pi\xi}{l} \quad (51)$$

Putting Formulae (49)-(51) into (31) to get

$$\begin{aligned} \omega_{\xi} &= -\sum_{m=1,3,5}^{\infty} \frac{4ql^3}{m^2 \pi^4 EJ} \left[\frac{1}{m^2} + \frac{1}{10}(2+\nu) \left(\frac{h}{l} \right)^2 \pi^2 \right] \cos \frac{m\pi\xi}{l} \\ &\quad - M_0 \frac{(l-2\xi)}{2EJ} + \frac{12J}{5hb}(1+\nu) \sum_{m=1,3,5}^{\infty} \frac{4ql}{\pi^2 EJ} \\ &\quad \cdot \left[\frac{1}{m^2} + \frac{1}{10}(2+\nu) \left(\frac{h}{l} \right)^2 \pi^2 \right] \cos \frac{m\pi\xi}{l} \\ &\quad - \frac{6}{25} \cdot \frac{h}{Eb}(1+\nu)(2+\nu) \sum_{m=1,3,5}^{\infty} \frac{4q}{l} \cos \frac{m\pi\xi}{l} \end{aligned} \quad (52)$$

By the left of the beam boundary conditions to get the corner that should be 0, and then

$$\begin{aligned} &\sum_{m=1,3,5}^{\infty} \frac{4ql^3}{m^2 \pi^4 EJ} \left[\frac{1}{m^2} + \frac{1}{10}(2+\nu) \left(\frac{h}{l} \right)^2 \pi^2 \right] + M_0 \frac{l}{2EJ} \\ &\quad - \frac{12J}{5hb}(1+\nu) \sum_{m=1,3,5}^{\infty} \frac{4ql}{\pi^2 EJ} \left[\frac{1}{m^2} + \frac{1}{10}(2+\nu) \left(\frac{h}{l} \right)^2 \pi^2 \right] \\ &\quad + \frac{6}{25} \frac{h}{Eb}(1+\nu)(2+\nu) \sum_{m=1,3,5}^{\infty} \frac{4q}{l} = 0 \end{aligned} \quad (53)$$

Solving to Formula (53) to get (54) (see below).

Putting Formula (54) into (48) to get (55) (see below).

2.2. Numerical Calculation

As a numerical calculation example, we make that the span of beam is $l = 1$ m, the height of beam is h , the width of beam is $b = 2h/3$, depth-span ratio $h/l = 0.1, 0.2, \dots, 0.8, 0.9$, Elastic modulus $E = 2.06 \times 10^{11}$ Pa, Poisson's ration is 0.3 to calculate the deflection value of beams with different depth-span ratio at various points along the y direction. In this example, the beam is divided into ten copies along the y direction ($\xi/l = 0.0, 0.1, \dots, 1.0$), and there is the $h/l = 0.1, 0.2, 0.3, 0.4, \dots, 0.8, 0.9$.

2.3. Finite Element Simulation

We use software of ANSYS Programming to calculate

the deflection value of beams along the y direction. The deflection value of beams with different depth-span ratio at various points along the y direction is list in the **Tables 3-5**. Considering the symmetry of the boundary, the uniform loading across the entire span and the symmetry of deflection value of the beams, every table only list deflection value of half of the beam.

2.4. Analysis of the Results

The deflection value of beams with different depth-span ratio at various points along the y direction is list in the **Tables 6-8**.

The numerical solution and finite element calculation values of the deflection of beams with different depth-span ratio at various points of the 1/2 cross section are respectively list in the **Tables 2** and **4**. In this paper, the error between ANSYS finite element solution and the

Table 2. Finite element deflection values at $b/2$ of deep beam of $h/l = 0.1$.

Layer \ ξ/l	0.0	0.1	0.2	0.3	0.4	0.5
1	0.0000	7.7052×10^{-7}	2.1498×10^{-6}	3.5719×10^{-6}	4.5999×10^{-6}	4.9717×10^{-6}
2	0.0000	7.6152×10^{-7}	2.1490×10^{-6}	3.5770×10^{-6}	4.6085×10^{-6}	4.9815×10^{-6}
3	0.0000	7.5435×10^{-7}	2.1484×10^{-6}	3.5809×10^{-6}	4.6152×10^{-6}	4.9891×10^{-6}
4	0.0000	7.4954×10^{-7}	2.1481×10^{-6}	3.5839×10^{-6}	4.6202×10^{-6}	4.9947×10^{-6}
5	0.0000	7.4688×10^{-7}	2.1481×10^{-6}	3.5859×10^{-6}	4.6234×10^{-6}	4.9983×10^{-6}
6	0.0000	7.4643×10^{-7}	2.1486×10^{-6}	3.5870×10^{-6}	4.6249×10^{-6}	4.9999×10^{-6}
7	0.0000	7.4824×10^{-7}	2.1494×10^{-6}	3.5872×10^{-6}	4.6247×10^{-6}	4.9996×10^{-6}
8	0.0000	7.5226×10^{-7}	2.1507×10^{-6}	3.5866×10^{-6}	4.6228×10^{-6}	4.9974×10^{-6}
9	0.0000	7.5841×10^{-7}	2.1524×10^{-6}	3.5849×10^{-6}	4.6192×10^{-6}	4.9931×10^{-6}
10	0.0000	7.6692×10^{-7}	2.1543×10^{-6}	3.5823×10^{-6}	4.6138×10^{-6}	4.9868×10^{-6}
11	0.0000	7.7724×10^{-7}	2.1564×10^{-6}	3.5785×10^{-6}	4.6065×10^{-6}	4.9783×10^{-6}
average	0.0000	7.5748×10^{-7}	2.1505×10^{-6}	3.5824×10^{-6}	4.6163×10^{-6}	4.9900×10^{-6}

$$\begin{aligned}
 w(\xi) &= \int_0^l \left[q - \frac{h^2}{10}(2+\nu) \frac{d^2 q}{dx^2} \right] w_1(x, \xi) dx + M_0 \theta_{l0} - M_0 \theta_{l1} \\
 &= \sum_{m=1,3,5}^{\infty} \frac{4ql^4}{m^3 \pi^5 EJ} \left[\frac{1}{m^2} + \frac{1}{10}(2+\nu) \left(\frac{h}{l} \right)^2 \pi^2 \right] \sin \frac{m\pi\xi}{l} - \frac{(l\xi - \xi^2) 2}{2EJ l} \\
 &\quad \cdot \sum_{m=1,3,5}^{\infty} \left\{ \frac{4ql}{\pi^2} \left[\left(\frac{l}{m\pi} \right)^2 - \frac{12J}{5bh}(1+\nu) \right] \left[\frac{1}{m^2} + \frac{1}{10}(2+\nu) \left(\frac{h}{l} \right)^2 \pi^2 \right] + \frac{24}{25}(1+\nu)(2+\nu) \frac{qhJ}{bl} \right\}
 \end{aligned} \tag{55}$$

Table 3. Finite element deflection values at $b/2$ of deep beam of $h/l = 0.3$.

Layer \ ξ/l	0.0	0.1	0.2	0.3	0.4	0.5
1	0.0000	2.8094×10^{-8}	5.7797×10^{-8}	8.4560×10^{-8}	1.0280×10^{-7}	1.0924×10^{-7}
2	0.0000	2.7056×10^{-8}	5.7763×10^{-8}	8.5192×10^{-8}	1.0383×10^{-7}	1.1040×10^{-7}
3	0.0000	2.6315×10^{-8}	5.7756×10^{-8}	8.5699×10^{-8}	1.0464×10^{-7}	1.1131×10^{-7}
4	0.0000	2.6000×10^{-8}	5.7880×10^{-8}	8.6186×10^{-8}	1.0534×10^{-7}	1.1209×10^{-7}
5	0.0000	2.6064×10^{-8}	5.8202×10^{-8}	8.6723×10^{-8}	1.0600×10^{-7}	1.1279×10^{-7}
6	0.0000	2.6446×10^{-8}	5.8758×10^{-8}	8.7352×10^{-8}	1.0667×10^{-7}	1.1346×10^{-7}
7	0.0000	2.7120×10^{-8}	5.9559×10^{-8}	8.8082×10^{-8}	1.0734×10^{-7}	1.1412×10^{-7}
8	0.0000	2.8159×10^{-8}	6.0590×10^{-8}	8.8899×10^{-8}	1.0802×10^{-7}	1.1476×10^{-7}
9	0.0000	2.9649×10^{-8}	6.1808×10^{-8}	8.9757×10^{-8}	1.0866×10^{-7}	1.1532×10^{-7}
10	0.0000	3.1630×10^{-8}	6.3139×10^{-8}	9.0586×10^{-8}	1.0918×10^{-7}	1.1574×10^{-7}
11	0.0000	3.3889×10^{-8}	6.4475×10^{-8}	9.1283×10^{-8}	1.0949×10^{-7}	1.1591×10^{-7}
average	0.0000	2.8220×10^{-8}	5.9793×10^{-8}	8.7665×10^{-8}	1.0654×10^{-7}	1.1319×10^{-7}

Table 4. Finite element deflection values at $b/2$ of deep beam of $h/l = 0.5$.

Layer \ ξ/l	0.0	0.1	0.2	0.3	0.4	0.5
1	0.0000	7.5279×10^{-9}	1.4375×10^{-8}	2.0080×10^{-8}	2.3839×10^{-8}	2.5147×10^{-8}
2	0.0000	7.1201×10^{-9}	1.4376×10^{-8}	2.0345×10^{-8}	2.4256×10^{-8}	2.5614×10^{-8}
3	0.0000	6.9645×10^{-9}	1.4416×10^{-8}	2.0574×10^{-8}	2.4600×10^{-8}	2.5997×10^{-8}
4	0.0000	7.0725×10^{-9}	1.4587×10^{-8}	2.0869×10^{-8}	2.4976×10^{-8}	2.6400×10^{-8}
5	0.0000	7.3217×10^{-9}	1.4919×10^{-8}	2.1293×10^{-8}	2.5455×10^{-8}	2.6895×10^{-8}
6	0.0000	7.6477×10^{-9}	1.5419×10^{-8}	2.1881×10^{-8}	2.6073×10^{-8}	2.7520×10^{-8}
7	0.0000	8.0376×10^{-9}	1.6098×10^{-8}	2.2641×10^{-8}	2.6840×10^{-8}	2.8284×10^{-8}
8	0.0000	8.5608×10^{-9}	1.6973×10^{-8}	2.3563×10^{-8}	2.7738×10^{-8}	2.9168×10^{-8}
9	0.0000	9.3551×10^{-9}	1.8050×10^{-8}	2.4608×10^{-8}	2.8722×10^{-8}	3.0127×10^{-8}
10	0.0000	1.0610×10^{-8}	1.9283×10^{-8}	2.5701×10^{-8}	2.9717×10^{-8}	3.1088×10^{-8}
11	0.0000	1.2210×10^{-8}	2.0531×10^{-8}	2.6733×10^{-8}	3.0622×10^{-8}	3.1951×10^{-7}
Average	0.0000	8.4026×10^{-9}	1.6275×10^{-8}	2.2572×10^{-8}	2.6622×10^{-8}	2.8018×10^{-8}

Table 5. Finite element deflection values at $b/2$ of deep beam of $h/l = 0.7$.

ξ/l Layer	0.0	0.1	0.2	0.3	0.4	0.5
1	0.0000	3.1017×10^{-9}	5.8212×10^{-9}	8.0132×10^{-9}	9.4325×10^{-9}	9.9234×10^{-9}
2	0.0000	2.8962×10^{-9}	5.8270×10^{-9}	8.1642×10^{-9}	9.6678×10^{-9}	1.0186×10^{-8}
3	0.0000	2.8994×10^{-9}	5.8899×10^{-9}	8.3143×10^{-9}	9.8784×10^{-9}	1.0418×10^{-8}
4	0.0000	3.0591×10^{-9}	6.0742×10^{-9}	8.5549×10^{-9}	1.0165×10^{-8}	1.0721×10^{-8}
5	0.0000	3.2750×10^{-9}	6.3816×10^{-9}	8.9325×10^{-9}	1.0590×10^{-8}	1.1162×10^{-8}
6	0.0000	3.5249×10^{-9}	6.8060×10^{-9}	9.4694×10^{-9}	1.1184×10^{-8}	1.1774×10^{-8}
7	0.0000	3.8106×10^{-9}	7.3596×10^{-9}	1.0178×10^{-8}	1.1958×10^{-8}	1.2565×10^{-8}
8	0.0000	4.1698×10^{-9}	8.0792×10^{-9}	1.1063×10^{-8}	1.2898×10^{-8}	1.3516×10^{-8}
9	0.0000	4.7055×10^{-9}	9.0128×10^{-9}	1.2107×10^{-8}	1.3962×10^{-8}	1.4580×10^{-8}
10	0.0000	5.6591×10^{-9}	1.0161×10^{-8}	1.3248×10^{-8}	1.5074×10^{-8}	1.5682×10^{-8}
11	0.0000	7.0435×10^{-9}	1.1375×10^{-8}	1.4354×10^{-8}	1.6124×10^{-8}	1.6713×10^{-8}
Average	0.0000	4.0132×10^{-9}	7.5262×10^{-9}	1.0218×10^{-8}	1.1903×10^{-8}	1.2476×10^{-8}

Table 6. Deflection values at $h/l = 0.1, 0.2, 0.3$.

ξ/l	0.1		0.2		0.3	
	text	ANSYS	text	ANSYS	text	ANSYS
0.0	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
0.1	7.9142×10^{-7}	7.5748×10^{-7}	8.7793×10^{-8}	8.3528×10^{-8}	2.9763×10^{-8}	2.8220×10^{-8}
0.2	2.2249×10^{-6}	2.1505×10^{-6}	2.0818×10^{-7}	1.9756×10^{-7}	6.3376×10^{-8}	5.9793×10^{-8}
0.3	3.6758×10^{-6}	3.5824×10^{-6}	3.2031×10^{-7}	3.0627×10^{-7}	9.2288×10^{-8}	8.7665×10^{-8}
0.4	4.7231×10^{-6}	4.6163×10^{-6}	3.9902×10^{-7}	3.8242×10^{-7}	1.1199×10^{-7}	1.0654×10^{-7}
0.5	5.1042×10^{-6}	4.9900×10^{-6}	4.2754×10^{-7}	4.0962×10^{-7}	1.1909×10^{-7}	1.1319×10^{-7}

Table 7. Deflection values at $h/l = 0.4, 0.5, 0.6$.

ξ/l	0.4		0.5		0.6	
	text	ANSYS	text	ANSYS	text	ANSYS
0.0	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
0.1	1.4920×10^{-8}	1.4033×10^{-8}	9.0358×10^{-9}	8.4026×10^{-9}	6.0671×10^{-9}	5.6055×10^{-9}
0.2	2.9919×10^{-8}	2.8063×10^{-8}	1.7503×10^{-8}	1.6275×10^{-8}	1.1506×10^{-8}	1.0649×10^{-8}
0.3	4.2072×10^{-8}	3.9739×10^{-8}	2.4084×10^{-8}	2.2572×10^{-8}	1.5615×10^{-8}	1.4580×10^{-8}
0.4	5.0153×10^{-8}	4.7421×10^{-8}	2.8381×10^{-8}	2.6622×10^{-8}	1.8261×10^{-8}	1.7068×10^{-8}
0.5	5.3055×10^{-8}	5.0094×10^{-8}	2.9919×10^{-8}	2.8018×10^{-8}	1.9206×10^{-8}	1.7919×10^{-8}

Table 8. Deflection values at $h/l = 0.7, 0.8, 0.9$.

ξ/l	$h/l = 0.7$		$h/l = 0.8$		$h/l = 0.9$	
	text	ANSYS	text	ANSYS	text	ANSYS
0.0	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
0.1	4.3651×10^{-9}	4.0132×10^{-9}	3.3021×10^{-9}	3.0201×10^{-9}	2.5834×10^{-9}	2.3587×10^{-9}
0.2	8.1661×10^{-9}	7.5262×10^{-9}	6.1204×10^{-9}	5.6113×10^{-9}	4.7570×10^{-9}	4.3512×10^{-9}
0.3	1.0980×10^{-8}	1.0218×10^{-8}	8.1767×10^{-9}	7.5741×10^{-9}	6.3262×10^{-9}	5.8483×10^{-9}
0.4	1.2775×10^{-8}	1.1903×10^{-8}	9.4791×10^{-9}	8.7934×10^{-9}	7.3147×10^{-9}	6.7732×10^{-9}
0.5	1.3415×10^{-8}	1.2476×10^{-8}	9.9429×10^{-9}	9.2066×10^{-9}	7.6664×10^{-9}	7.0858×10^{-9}

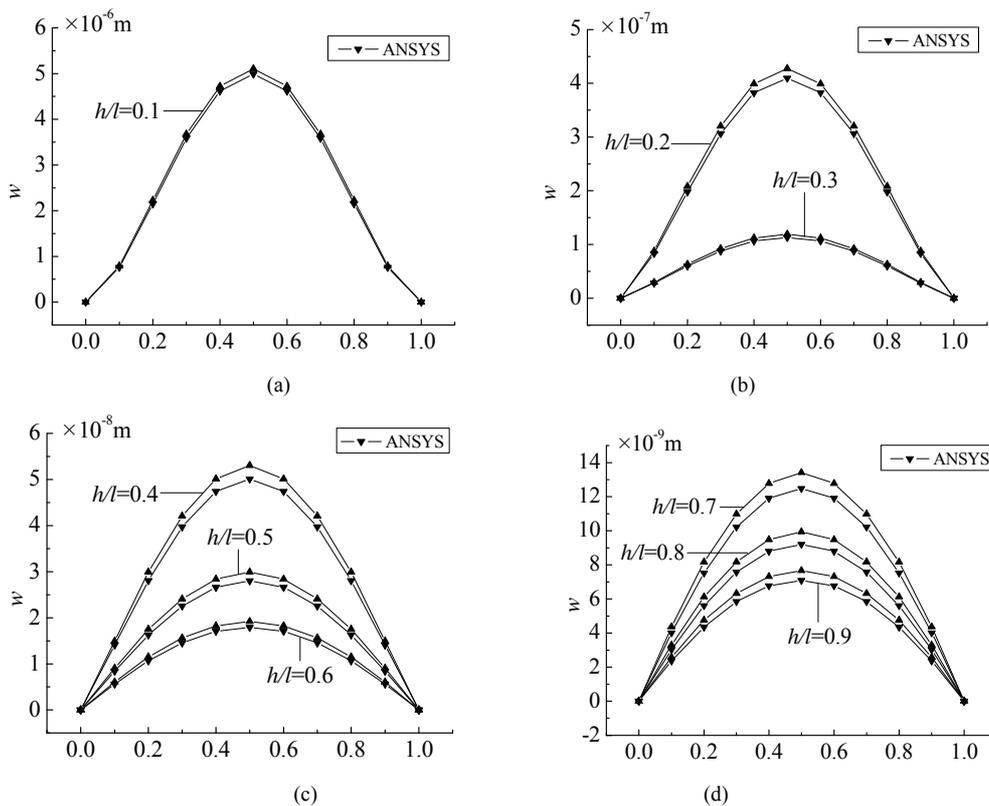


Figure 5. Deflection distribution curve at $b/2$ with different depth-span ratios.

numerical solution respectively are 4.29%, 5.10%, 5.65%, 6.20%, 7.02%, 7.61%, 8.06%, 8.54%, 8.70%, all of which are in the range of allowable error. We can know the result is correct, and considering the reciprocal method to solve the problem is right.

The deflection distribution curve of beams with different depth-span ratio at various points along the y direction and the distribution curve of the finite element solution of the deep beam ($h/l = 0.1, 0.2, 0.3, 0.4, \dots, 0.8, 0.9$) are respectively list in the **Figure 5**. Directly comparing

with numerical results, and the two results can well fitting.

3. Conclusions

Considering the effects of the beam section rotation, shear deformation of the adjacent section and transverse pressure, derived the new equation of rectangular section deep beams, and gives the basic solution of deep beams. And we solve the example of the bending problems of

deep rectangular beams with both ends simply supported, fixed at both ends under uniform load, based on the equations given in this paper, application of reciprocal law, doing numerical calculation in Matlab platform, compare with the results of ANSYS finite element analysis.

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